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STUDY ON THE PENNY PLATFORM OF VARIABLE STATOR VANE IN COMPRESSOR

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2015-10-07

Outline

◆ Background

◆ Experimental and numerical study on 2D cascade

◆ Numerical study on 3D cascade

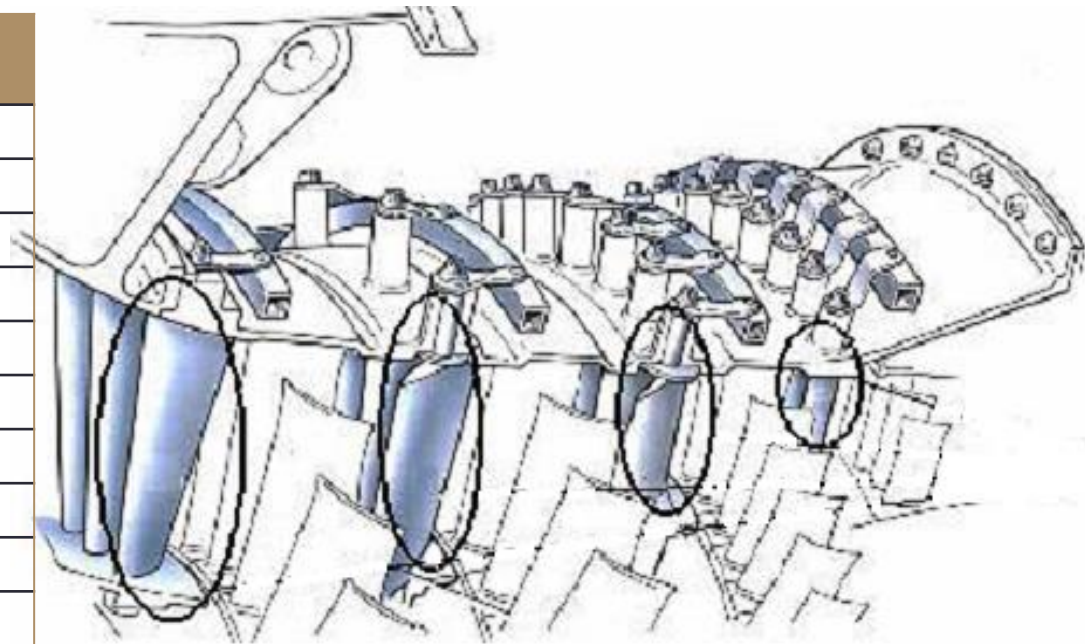
◆ Conclusion



Background

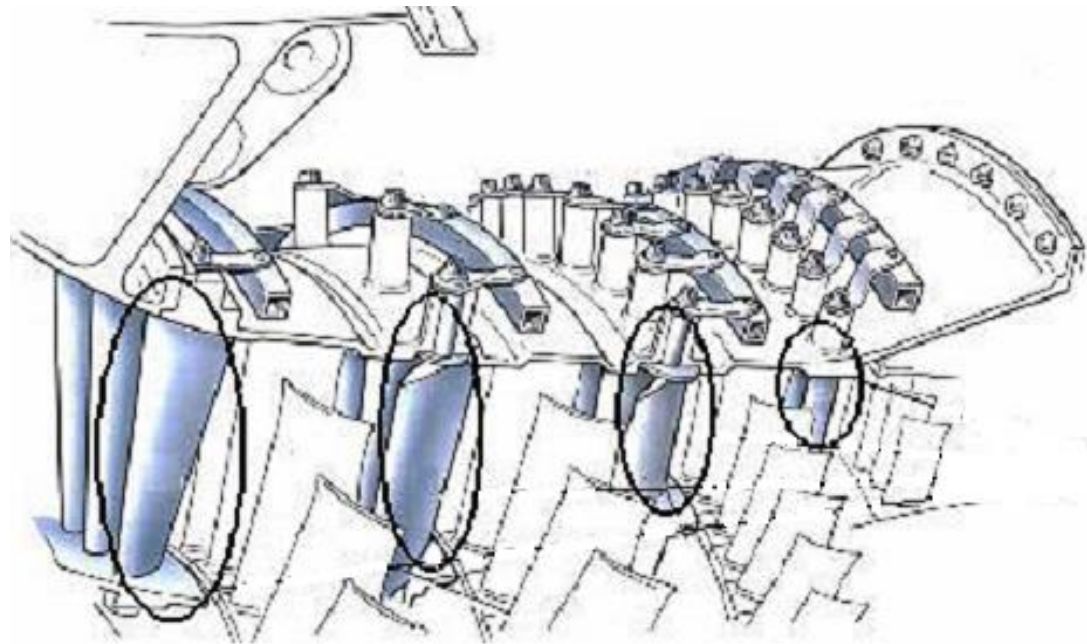
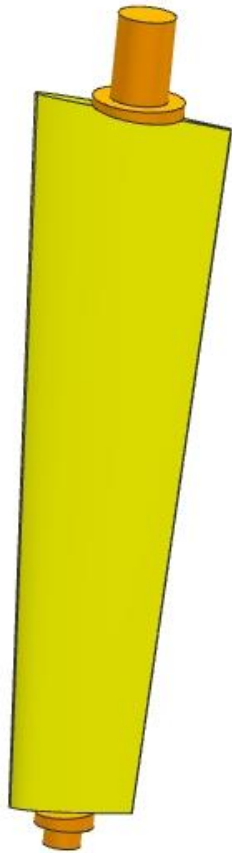
Variable Stator Vane (VSV) is widely used to improve multistage compressor performance at off-design condition.

Engine	VSV/total Stages
CFM56-3	4/9
PW4000	4/11
V2500	4/10
PW2037	5/12
JT9D-7R4E	4/11
CF6-80A	6/14
АЛ-31Ф	3/9
F119	3/6
M88	3/9
EJ200	2/5
TECH56	3/6
GE90	1/510



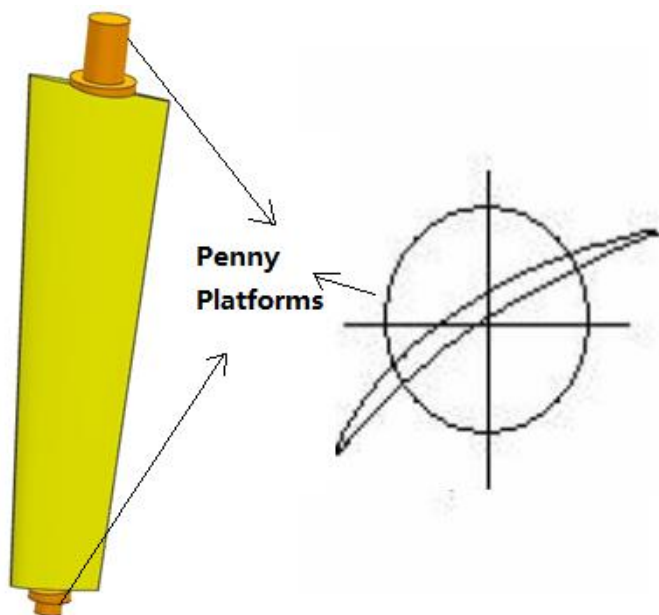
Background

VSV consist of blade and penny platforms.

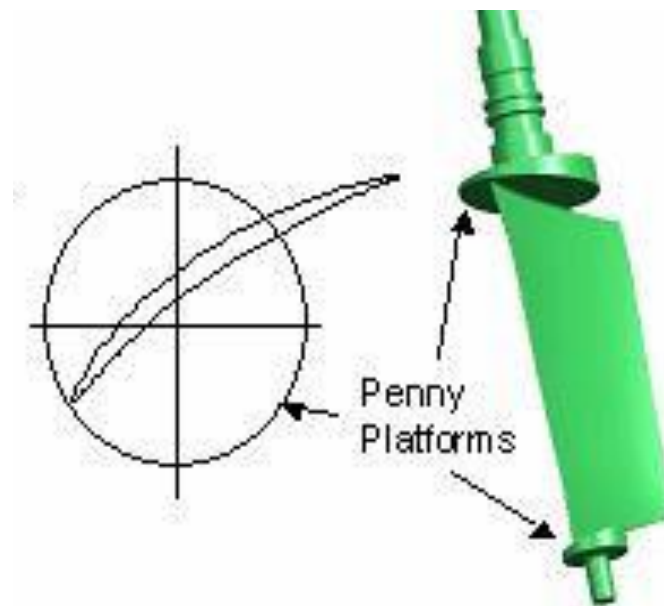


Background

Previous Design

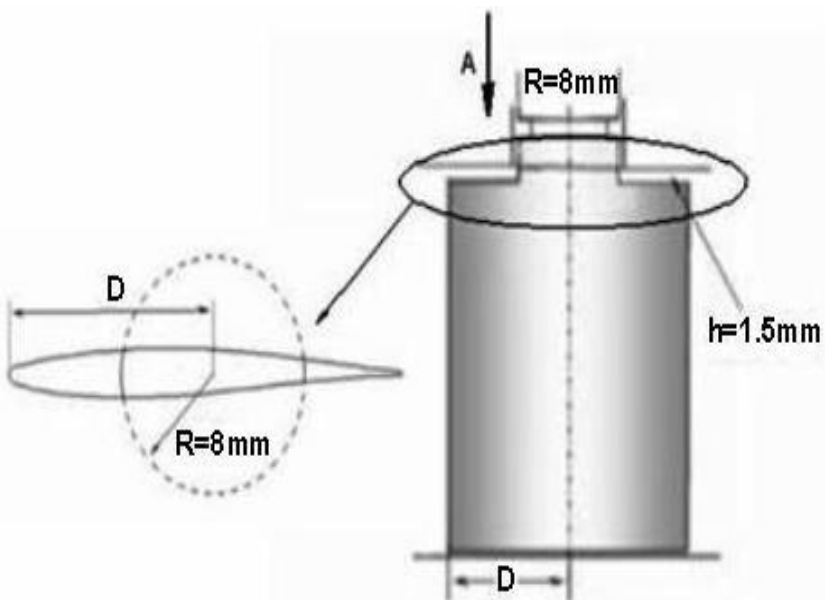


Advanced Design



Does the location and size of penny platforms impact on leakage loss only in VSV?

Experimental and numerical study on 2D cascade



Blade height	100mm
gap	1.5mm
pitch	30mm
chord	34.3mm
Blade number	8

Blade A



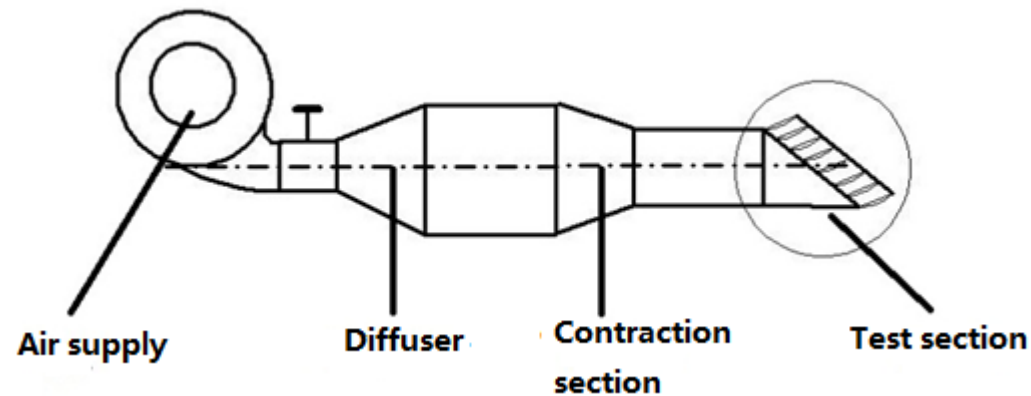
Blade B



Blade C



□ Plane cascade test rig at NUAA



Inflow condition:

Mach : 0.5

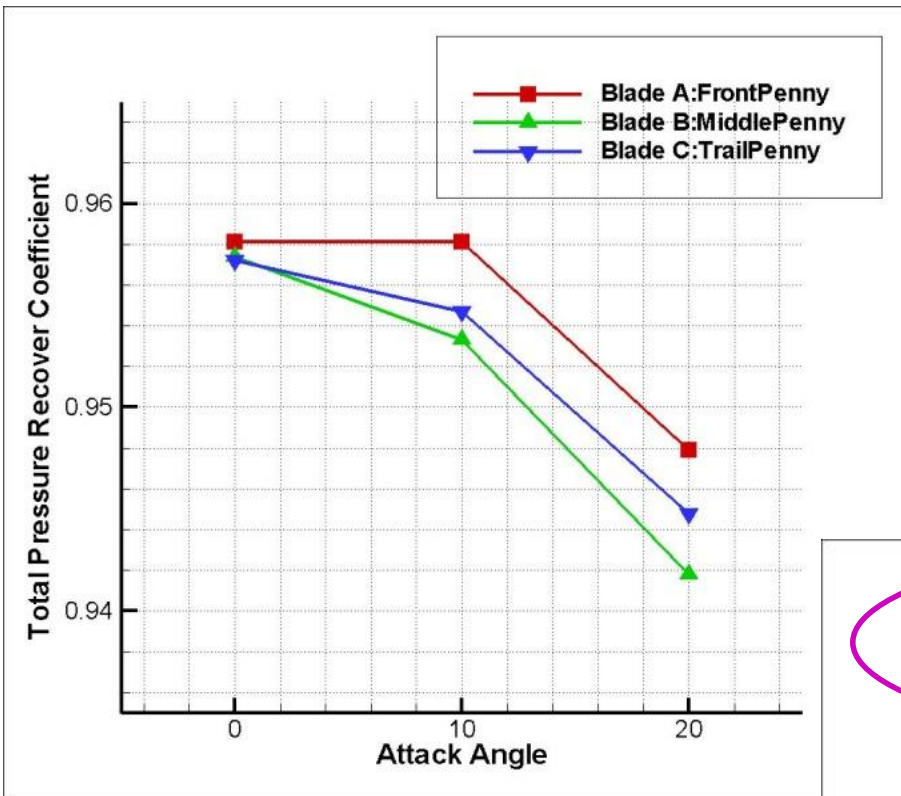
Attack : 0° , 10° , 20°



Outflow measure:

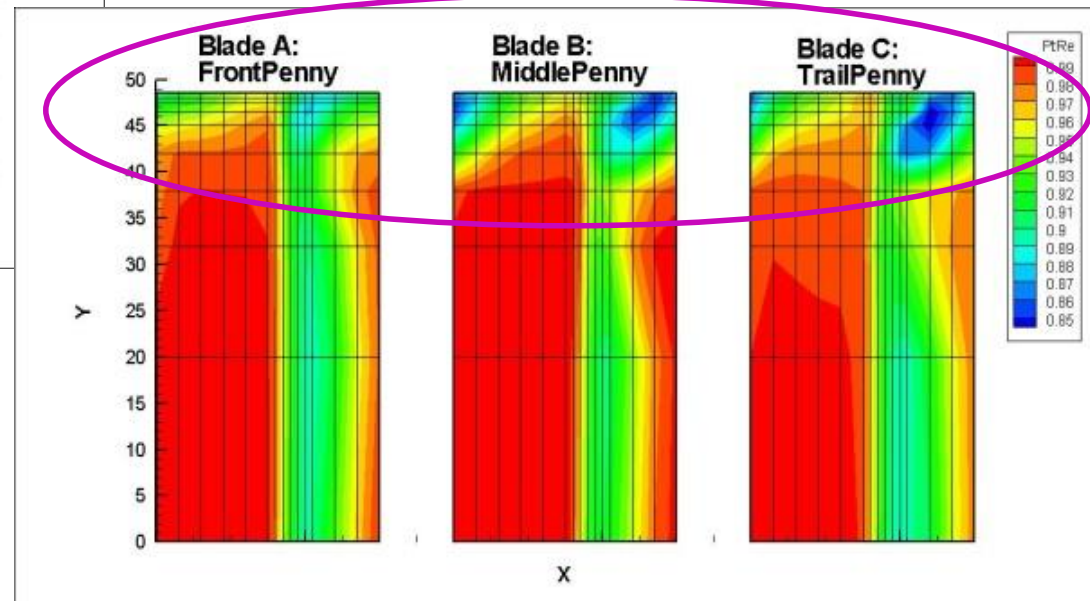
Total pressure is acquired at a blade passage section downstream of $0.5 \times$ blade chord

□ Experimental result



Total pressure recovery coefficient:
Blade A > Blade C > Blade B

Attack of 10°

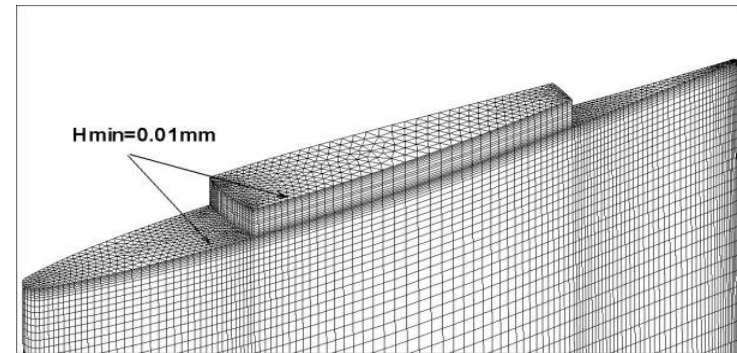
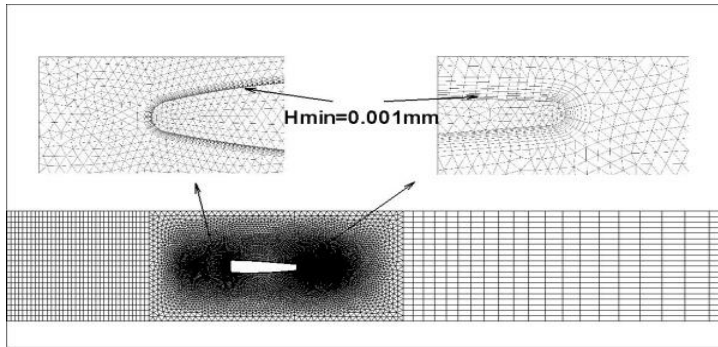


Trends of end-wall loss is consistent with blade loss

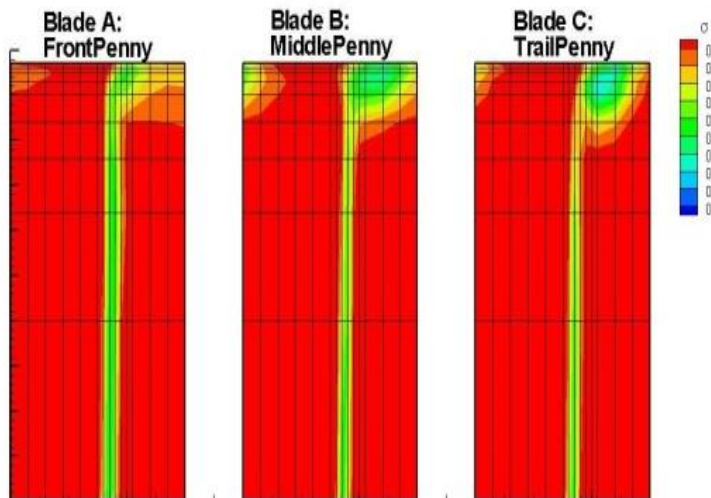
CFD method

Mesh: hybrid grid, $y^+ < 5$

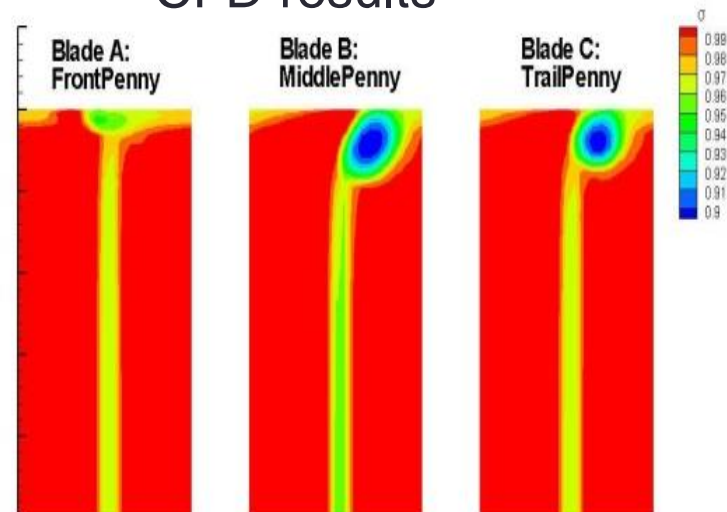
Method: commercial CFD software CFX; $k-\epsilon$ turbulent model;



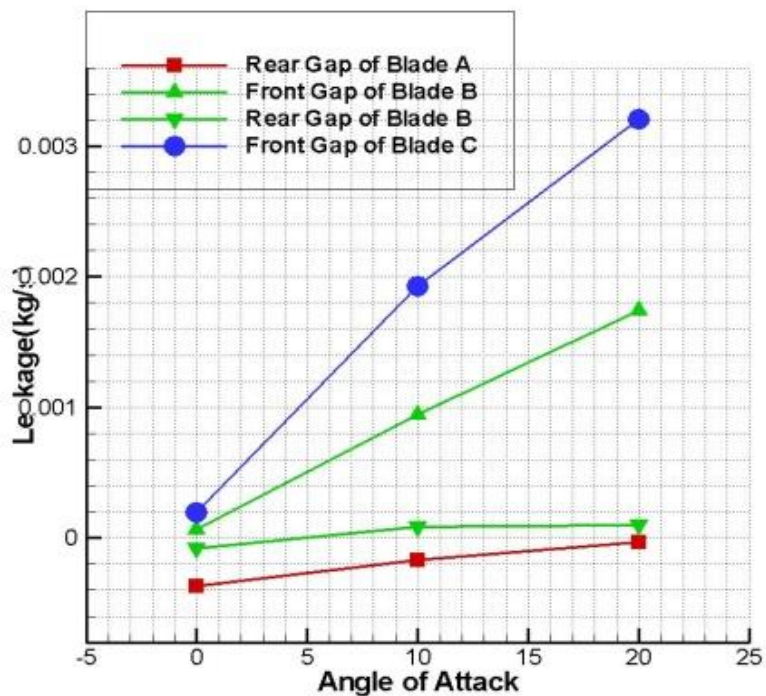
Experimental results



CFD results



CFD result



Leakage flow:

Blade A < Blade B < Blade C

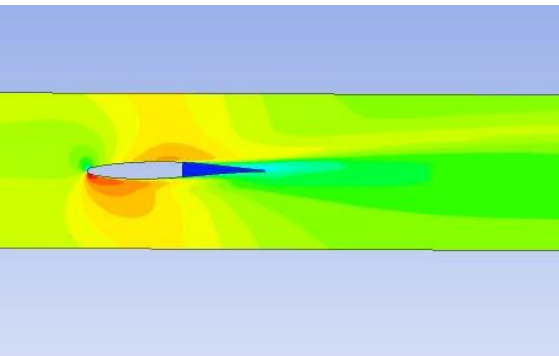
Total pressure recovery coefficient:

Blade A > Blade C > Blade B

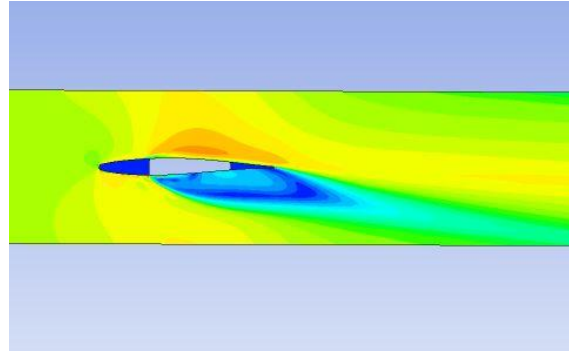
Endwall loss \neq leakage loss

CFD result

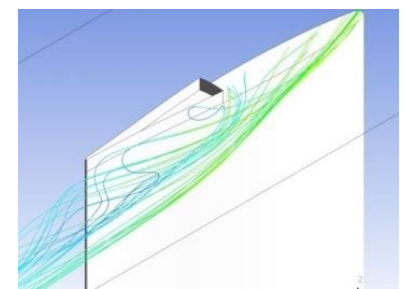
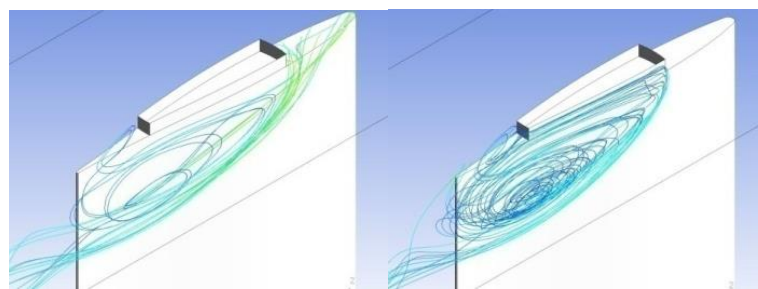
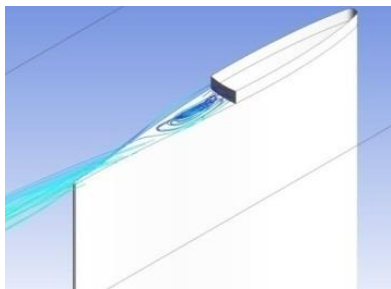
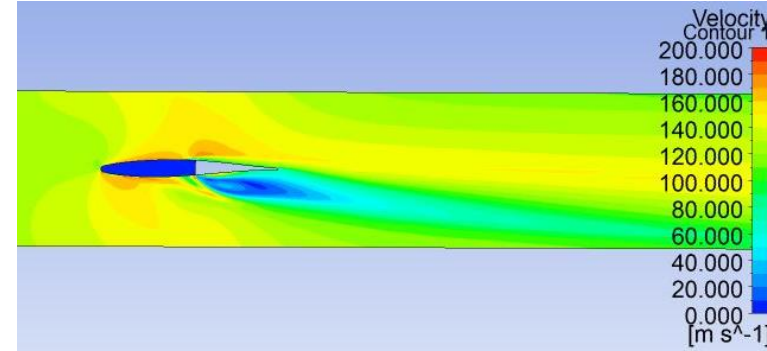
Blade A



Blade B



Blade C



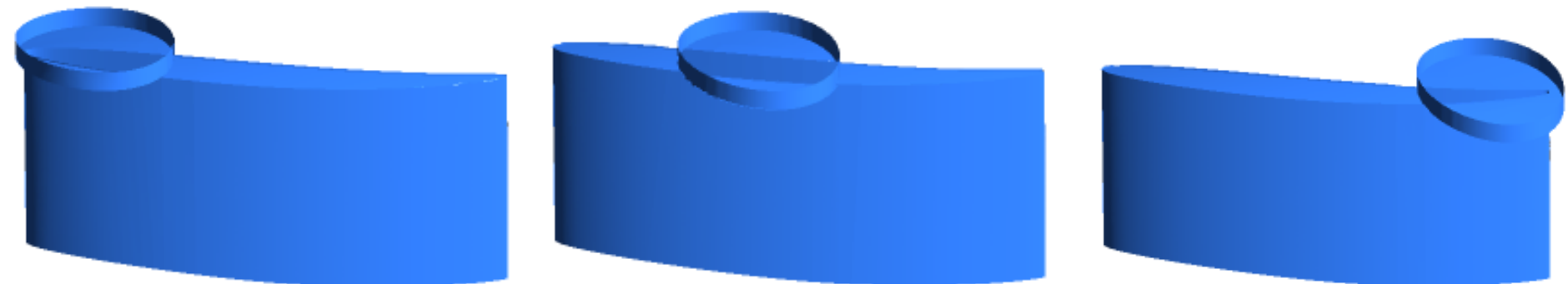
Endwall loss = leakage loss + leakage flow induced loss

Front penny platforms eliminate the induced loss.

Numerical study on 3D cascade

Blade height	56.25mm
Tip gap	4.9mm
solidity	2.24
chord	125mm
Blade number	19
Stagger angle	51.12°

- Tip gap is obtained by blade rotation angle 10°
- Penny radius is 20mm.

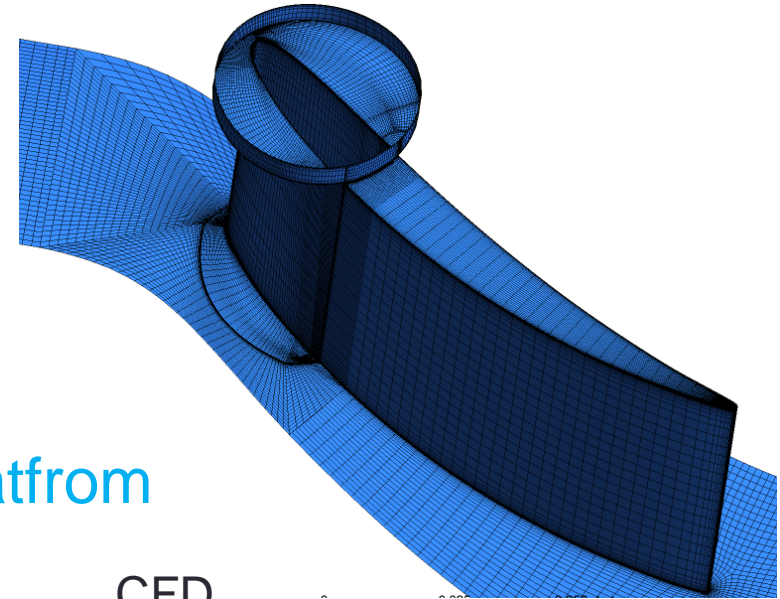


CFD method

Mesh: structured grid, $y^+ < 5$

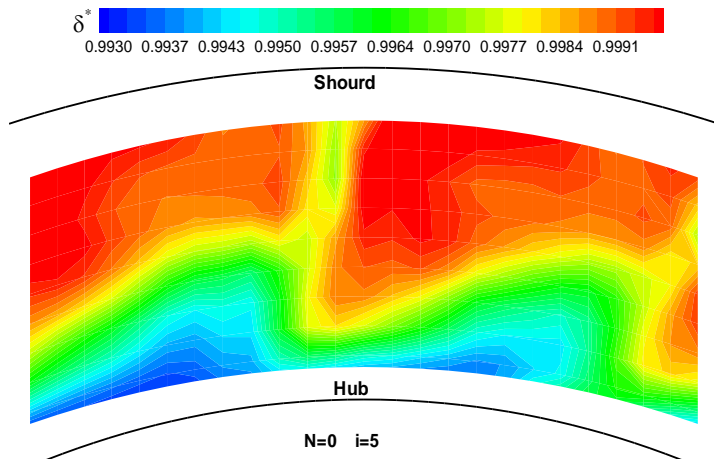
Method: CFX; k-e turbulent model;

Boundary condition: inflow attack is 5° ;
mass flow is 1.9kg/s

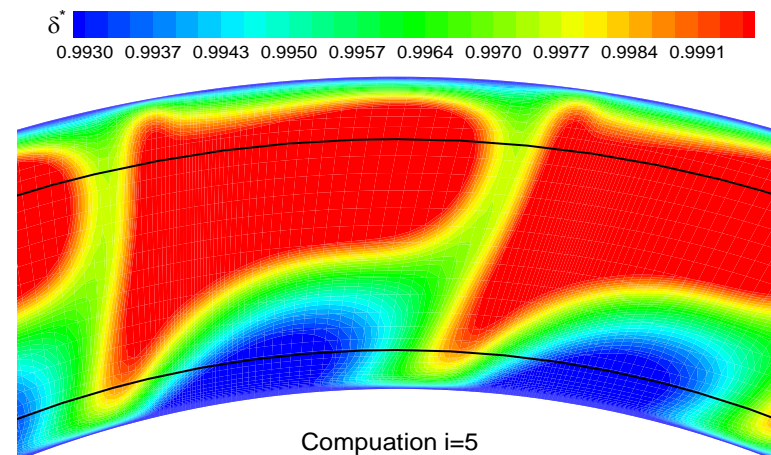


Results of cascade without penny platform

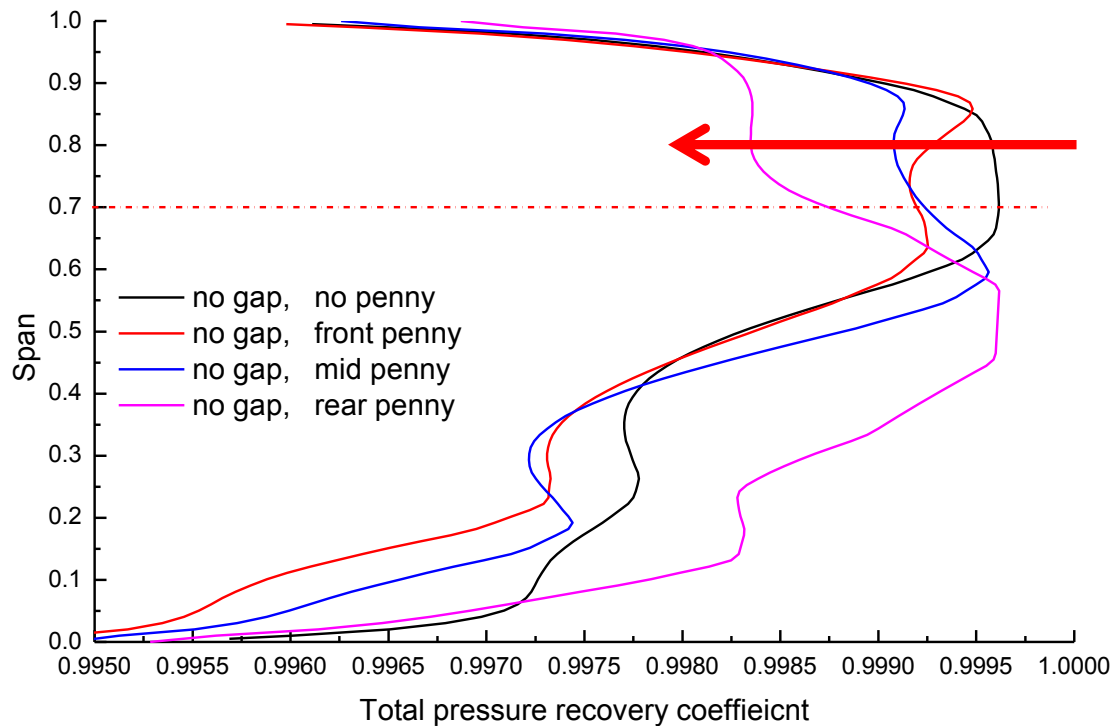
Experimental



CFD



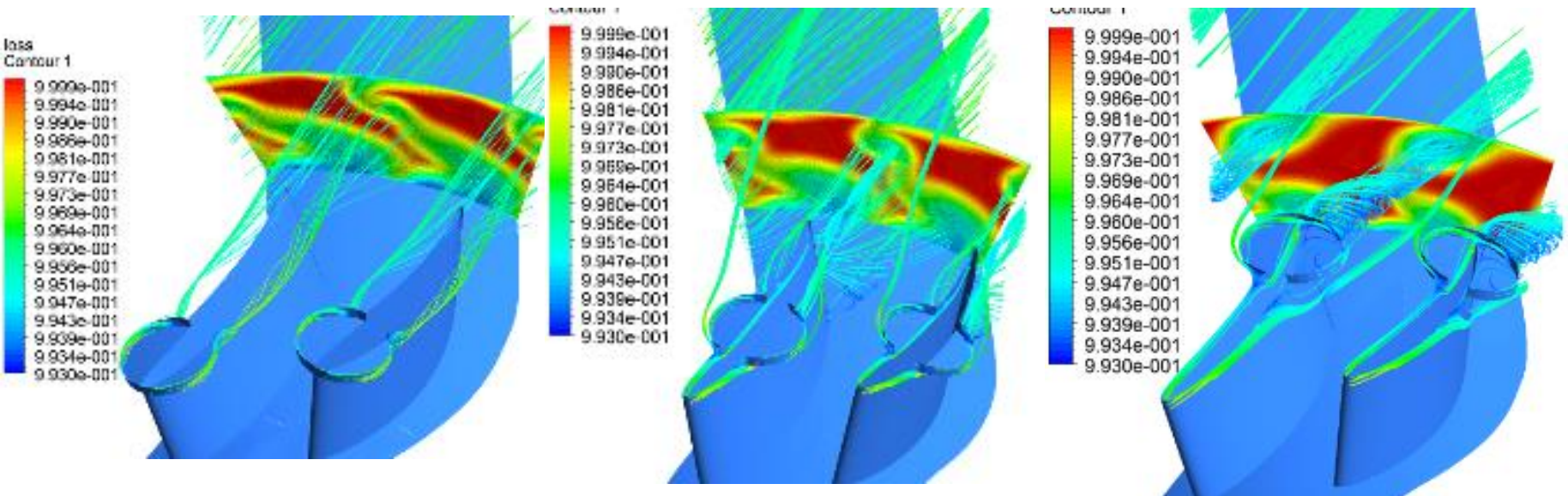
CFD results — cascade without tip gap



The end-wall loss become larger as it move downstream

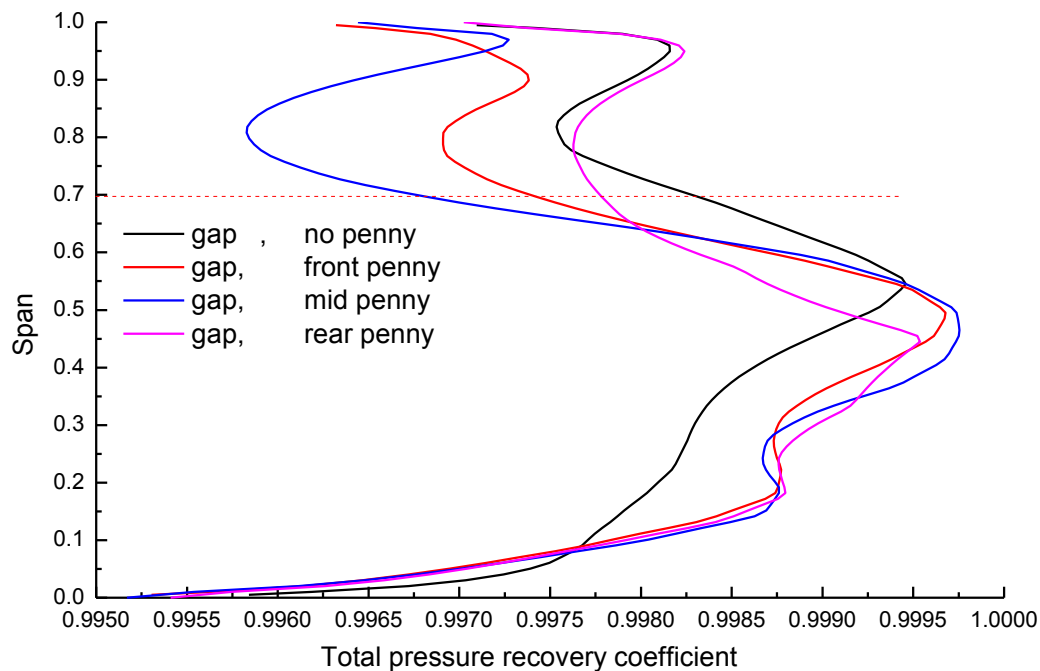


CFD results——cascade without tip gap



Penny platform wake loss is affected by the pressure difference through the blade passage

CFD results — cascade with tip gap



End-wall loss :

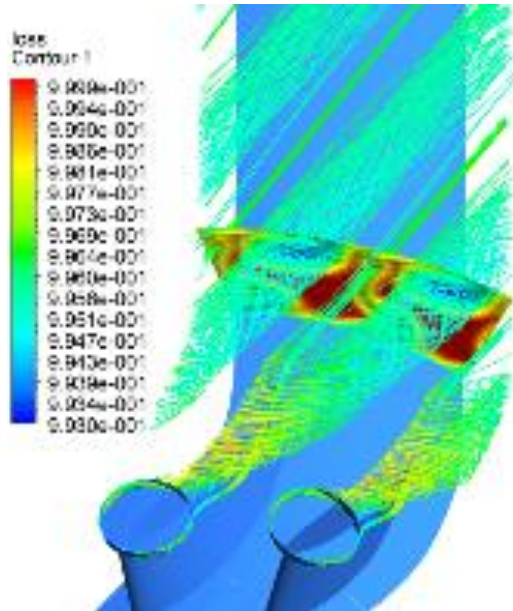
Mid penny > front penny > rear penny

End-wall loss of blade shaped penny platform:

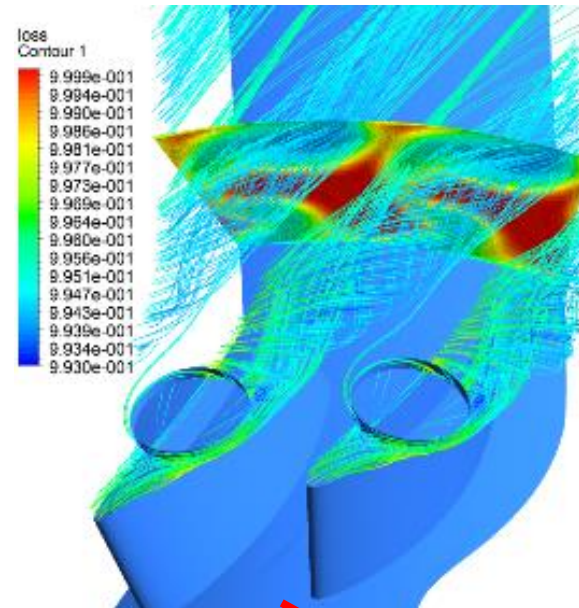
Mid penny > rear penny > front penny



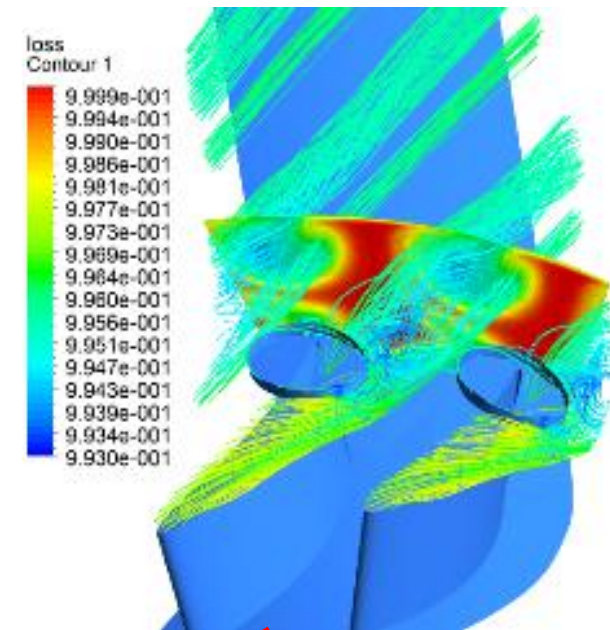
CFD results——cascade with tip gap



Leakage loss +
Penny platforms wake loss



Leakage loss +
Leakage flow induced loss



Conclusion

- ◆ VSV end-wall loss includes leakage loss and leakage flow induced loss/penny platform wake loss.
- ◆ Front penny platforms diminish the leakage flow induced loss and generate wake loss around it.
- ◆ The location and size of penny platforms is determined by end-wall loss. Blades with middle penny has greatest loss.



Thanks for your attention!

