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OMICS Group has organized 500 conferences, workshops and national symposiums across the major cities including San Francisco, Las Vegas, San Antonio, Omaha, Orlando, Raleigh, Santa Clara, Chicago, Philadelphia, Baltimore, United Kingdom, Valencia, Dubai, Beijing, Hyderabad, Bengaluru and Mumbai.

Design and Performance Prediction of Centrifugal Compressor

**Mech – Aero 2015
- Aerodynamics -**

Oct. 05, 2015

Won-Seok Heo

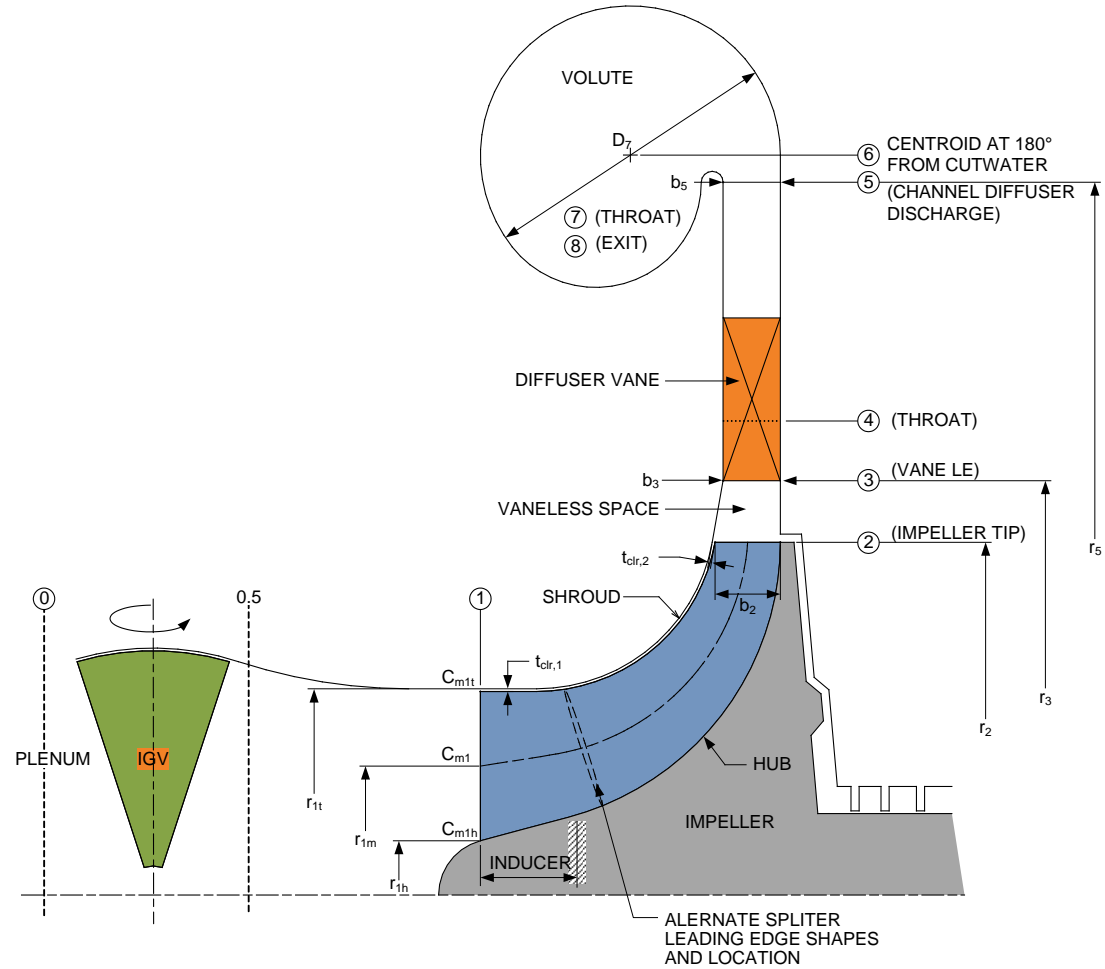


**Turbo System & Control Laboratory
Institute of Advanced Machines and Design(IAMD)
School of Mechanical and Aerospace Engineering
Seoul National University**

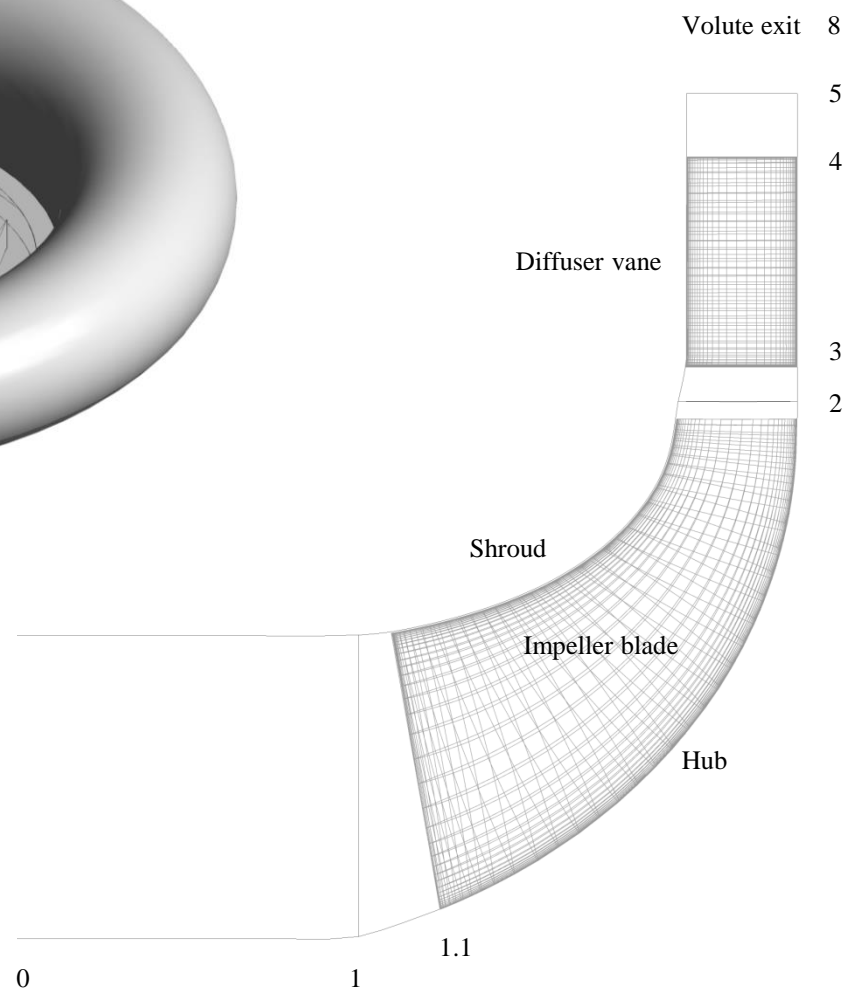
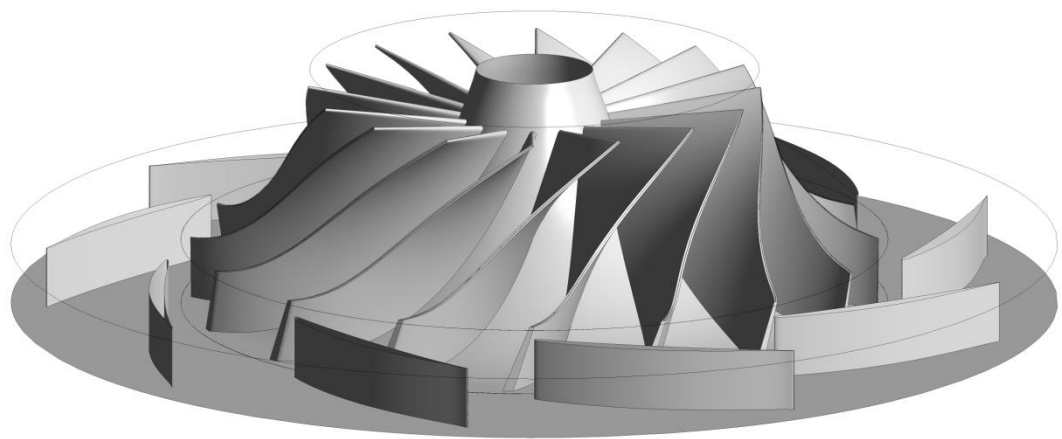
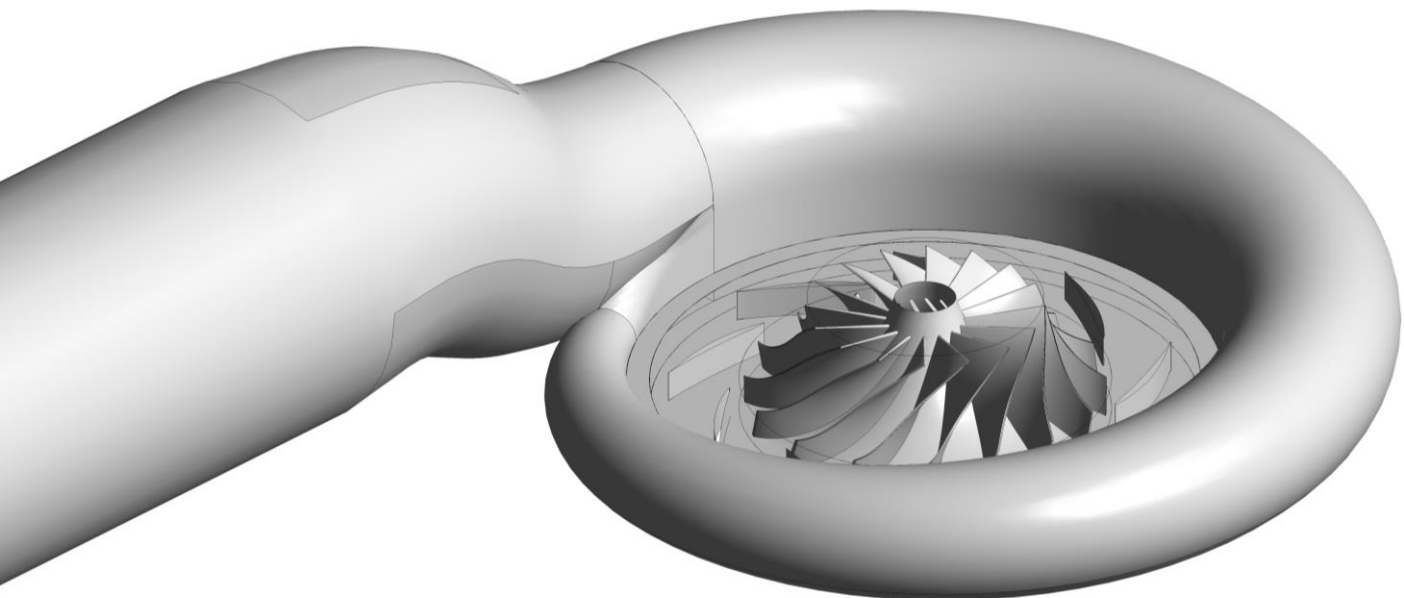


- ❖ **Introduction**
- ❖ **Research Objective and Numerical Method**
- ❖ **Validation of CFD calculation results**
- ❖ **Analysis & Prediction model**
- ❖ **Conclusions**

Centrifugal compressor



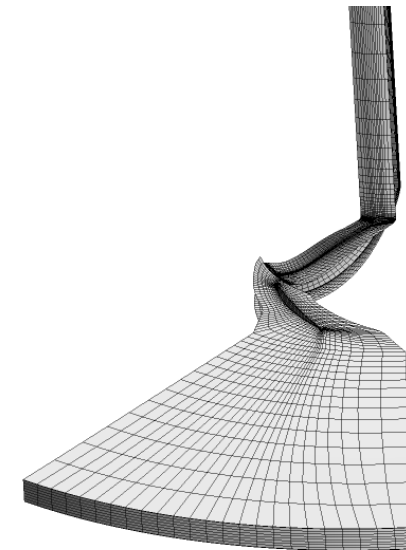
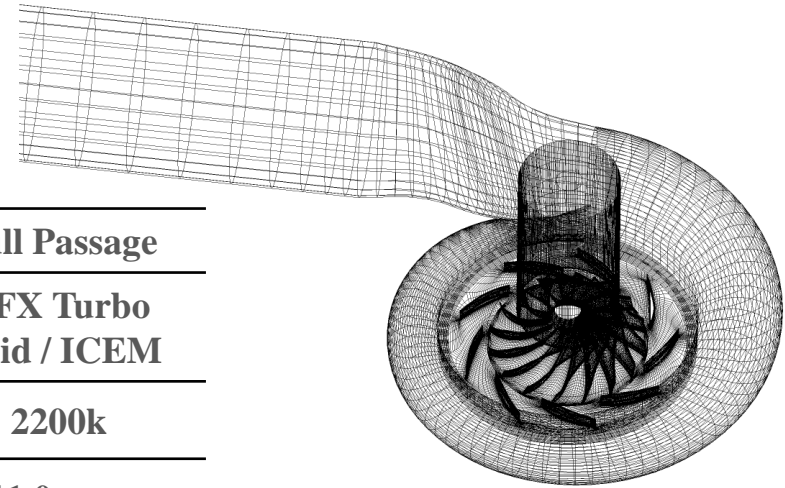
Research subject



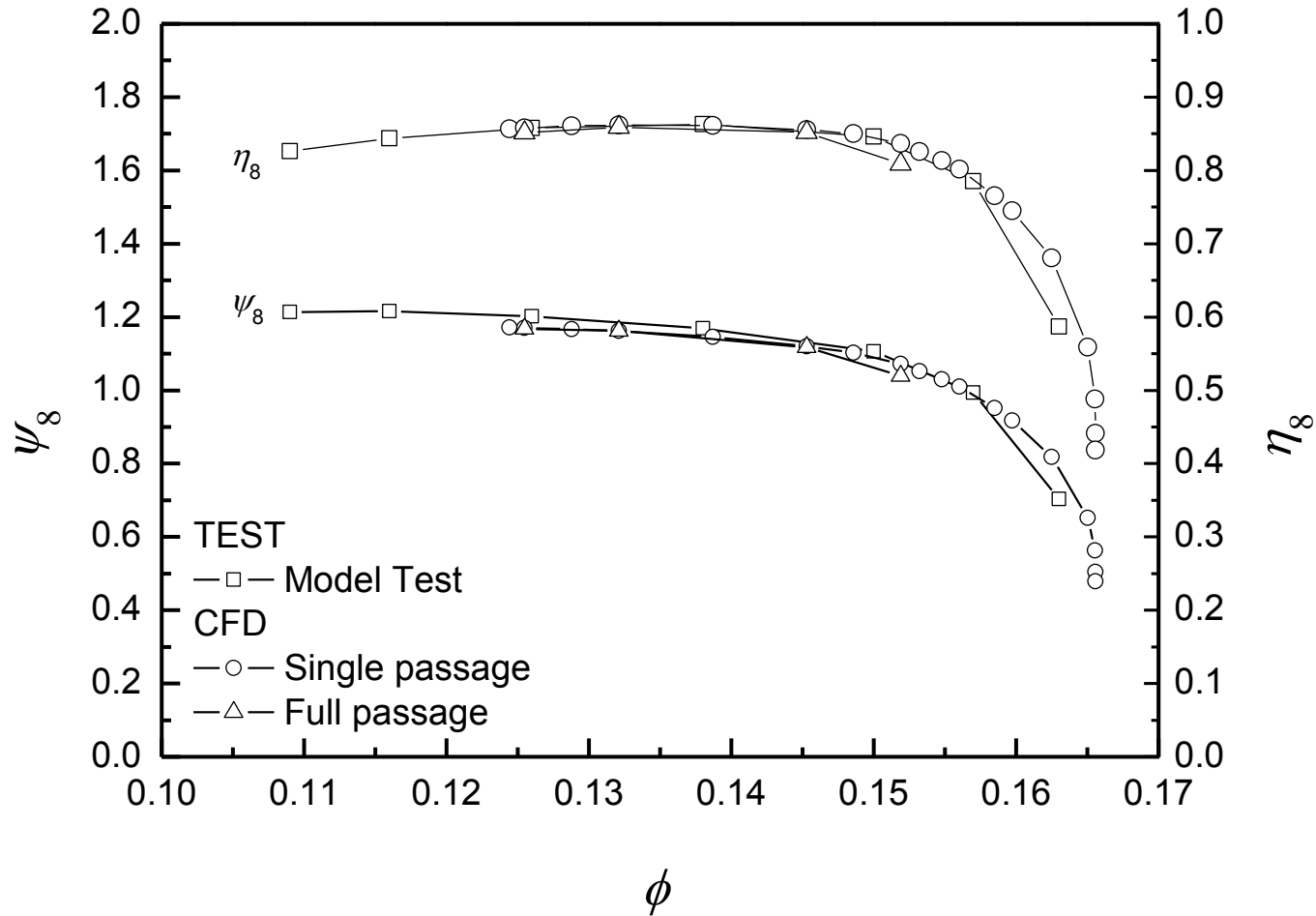
Simulation Condition



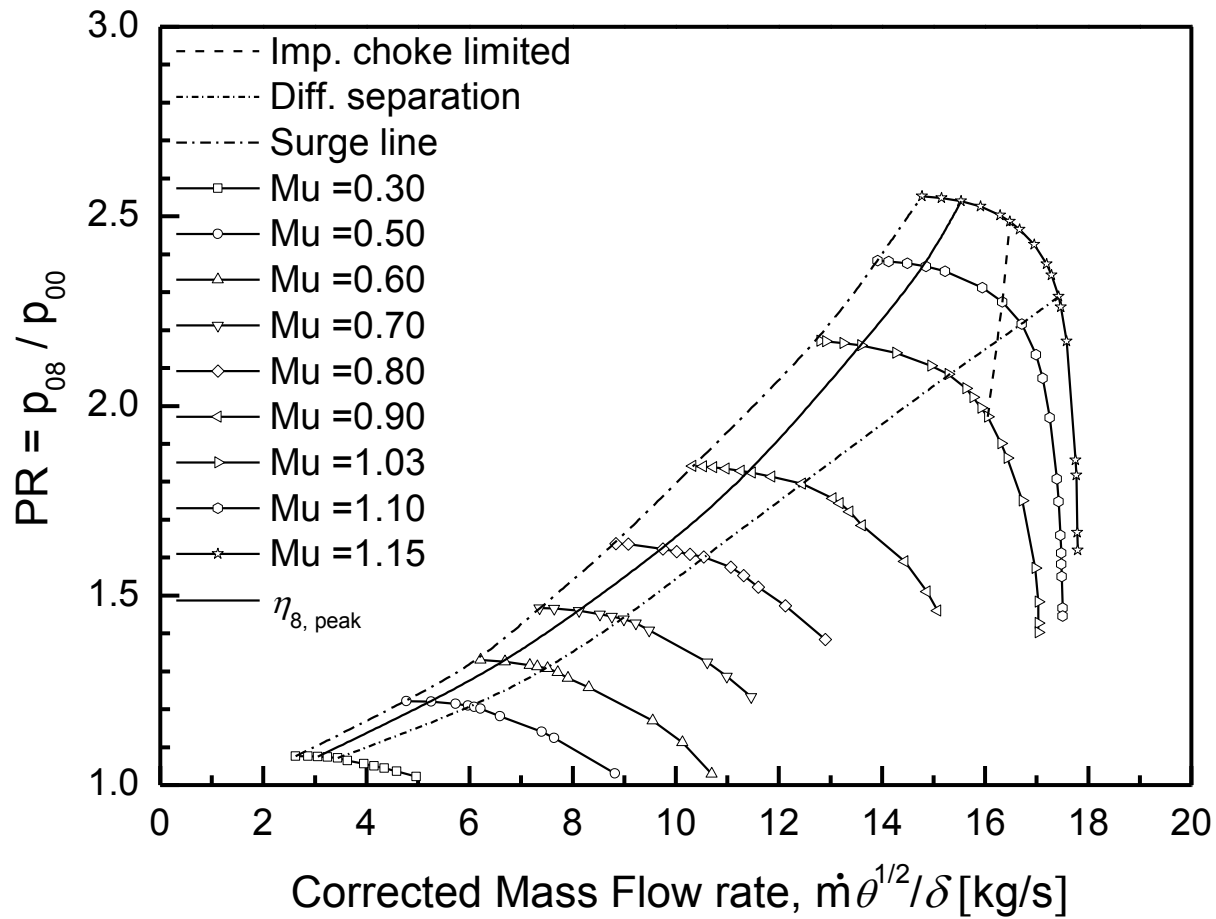
Type		Single Passage	Full Passage
Grid	Tool	CFX Turbo Grid	CFX Turbo Grid / ICEM
	Total nodes	120k	2200k
Solver	Tool	ANSYS CFX 11.0	
	Turbulence model	k-ε RANS	
	Simulation type	Steady	
Boundary condition	Inlet	Total pressure Total temperature Flow angle	
	Outlet	Mass flow rate	
	Interface	Mixing plane	



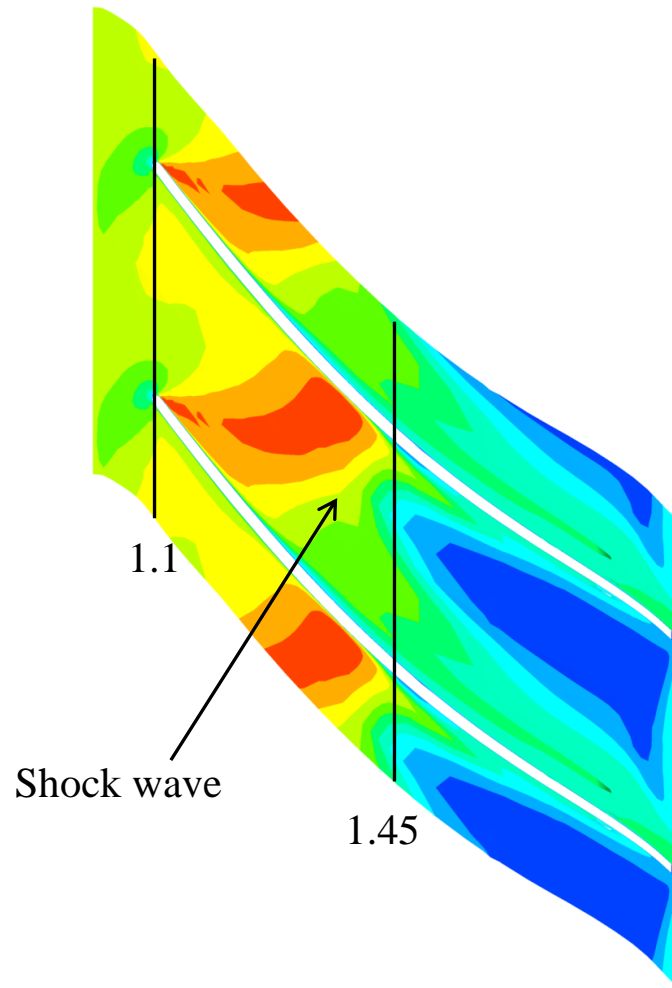
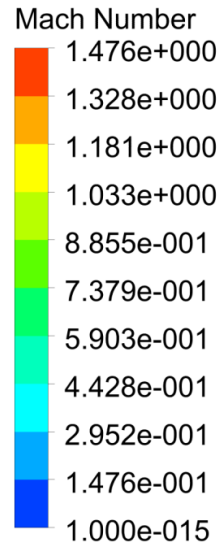
Validation of CFD result



Performance map



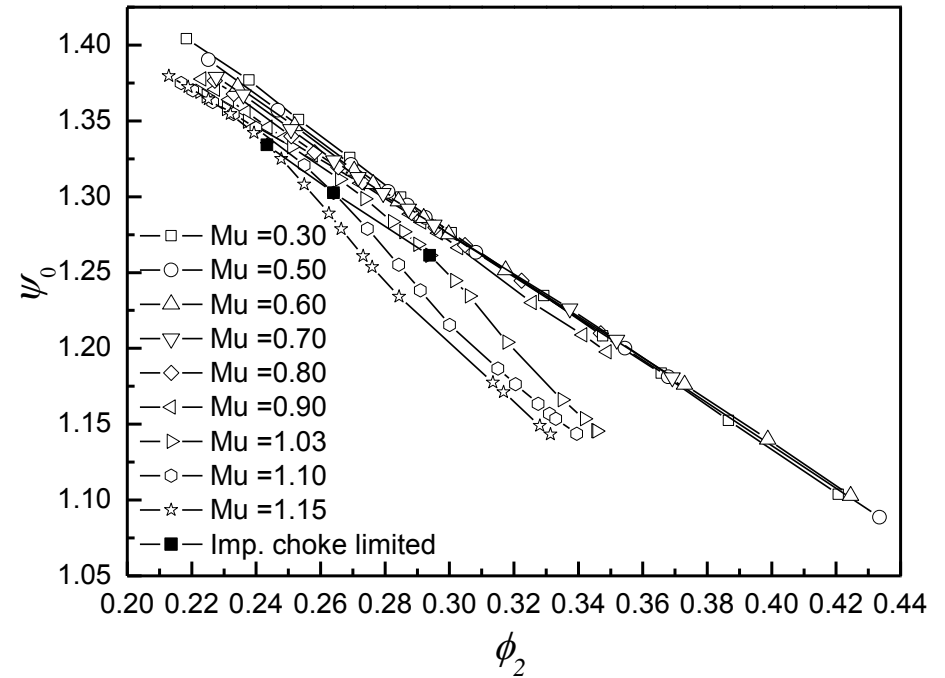
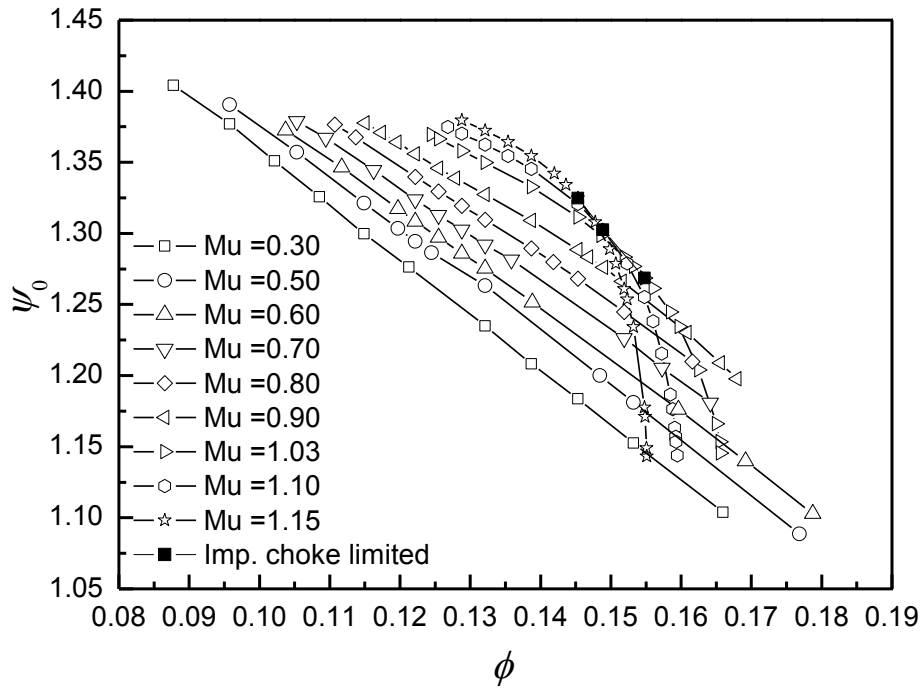
Impeller choking near shroud(0.9span) ; $Mu=1.10$, $\phi=0.158$



Impeller Performance ; Work coefficient



$$\psi_0 = \frac{\psi_2}{\eta_2} = 2 \frac{\Delta h_{02}}{U_2^2} = 2 \frac{U_2 c_{\theta 2} - \overline{U_1 c_{\theta 1}}^M}{U_2^2} = 2 \frac{c_{\theta 2}}{U_2} = 2 \frac{C_{p0}(T_{02} - T_{01})}{U_2^2}$$



Flow chart ; Work coefficient Prediction model

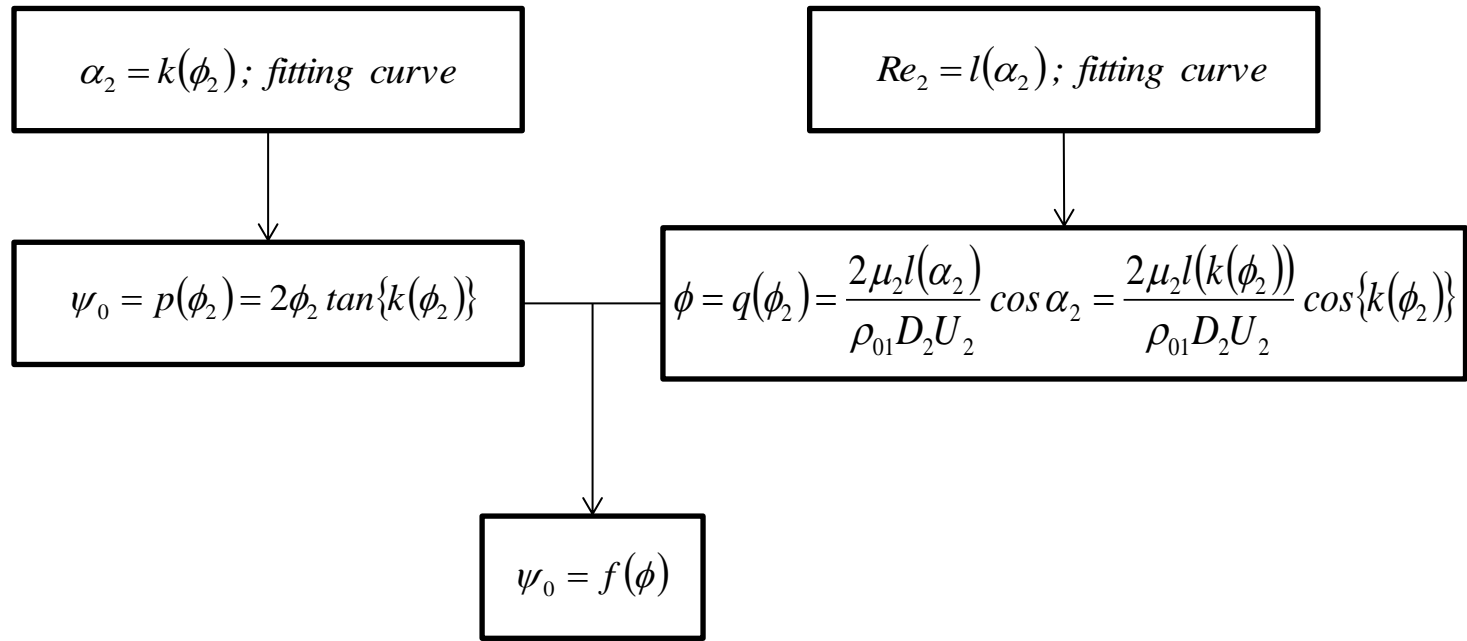
$$\frac{\psi_0}{2} = \phi_2 \tan \alpha_2$$

$$\psi_0 = 2\phi_2 \tan [k(M_u, \phi_2)]$$

$$\psi_0 = p(M_u, \phi_2)$$

$$\phi = \frac{\dot{m}}{\rho_{01} \frac{\pi}{4} D_2^2 U_2} = \frac{\rho_2 A_2 c_2 \cos \alpha_2}{\rho_{01} \frac{\pi}{4} D_2^2 U_2} = \frac{\rho_2 \pi D_2 b_2 c_2 \cos \alpha_2}{\rho_{01} \frac{\pi}{4} D_2^2 U_2} = \frac{\rho_2}{\rho_{01}} \frac{4b_2}{D_2} \frac{c_2}{U_2} \cos \alpha_2$$

$$= \frac{2\mu_2 Re_2}{\rho_{01} D_2 U_2} \cos \alpha_2 \left[Re_2 = \frac{\rho_2 c_2 2b_2}{\mu_2} \right]$$



Flow angle function of flow coefficient at impeller exit

$$0.3 < M_u < 0.8$$

$$\text{Linear fit } (x, y) = (\phi_2, \alpha_2)$$

$$\alpha_2 = A + B\phi_2$$

parameters A, B

A : intercept

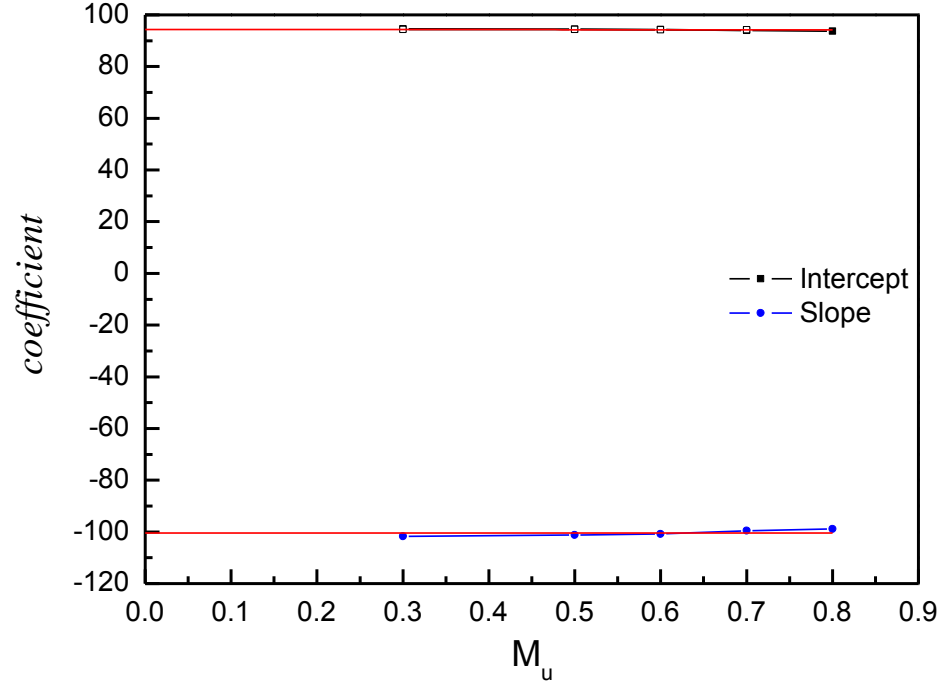
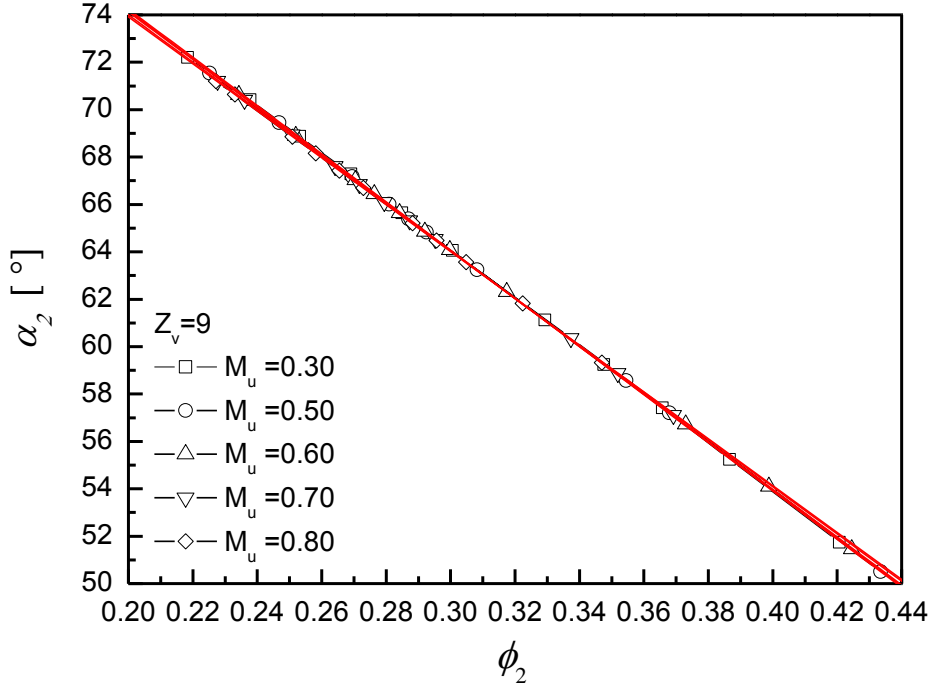
B : slope

$$Z_v = 9$$

$$\alpha_2 = A(\text{result data}) + \bar{B}\phi_2$$

$$\left[\begin{array}{l} \text{Linear fit } (x, y) = (M_u, A) \\ A = a + bM_u = 94.37 - 0.197M_u \\ \bar{B} = -100.418 \end{array} \right]$$

$$\begin{aligned} \therefore \alpha_2 &= a + bM_u + \bar{B}\phi_2 \\ &= 94.37 - 0.197M_u - 100.418\phi_2 \end{aligned}$$



Flow angle function of flow coefficient at impeller exit

$$0.8 < M_u < 1.2$$

$$\text{Parabola fit } (x, y) = (\phi_2, \alpha_2)$$

$$\alpha_2 = A + B\phi_2 + C\phi_2^2$$

parameters A, B, C

$$Z_v = 9$$

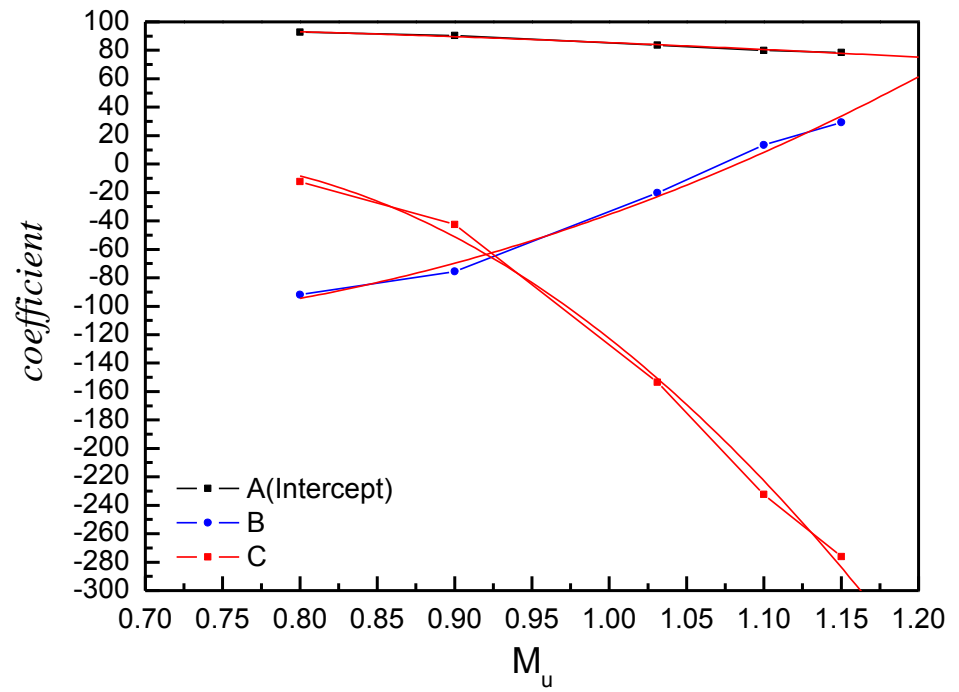
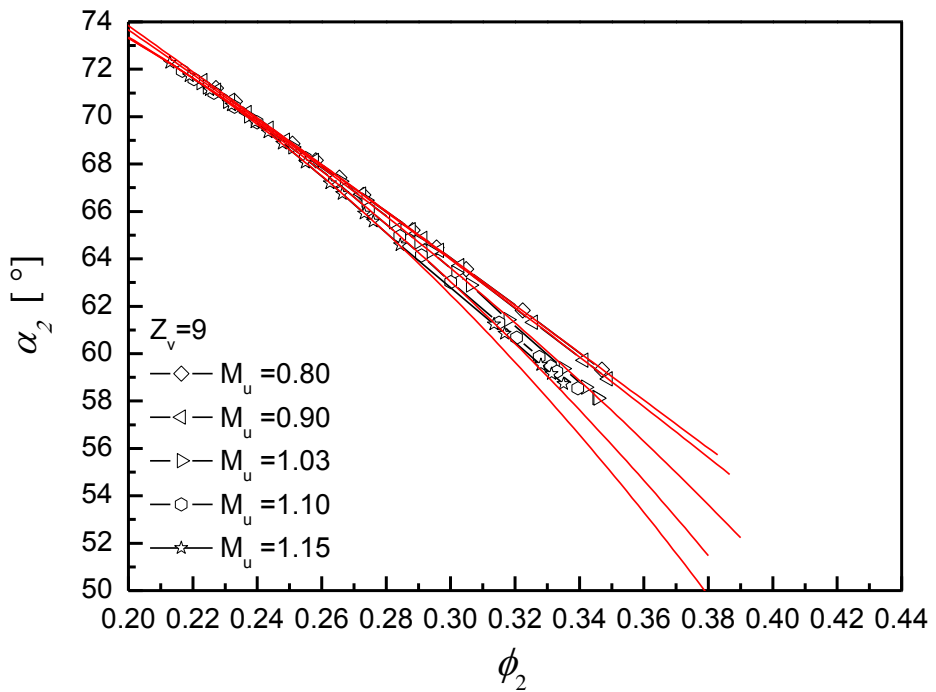
$$\text{Parabola fit } (x, y)$$

$$A = a + bM_u + cM_u^2 ; a = 97.72, b = 20.325, C = -32.631$$

$$B = a + bM_u + cM_u^2 ; a = 48.98, b = -558.8, C = 474.345$$

$$C = a + bM_u + cM_u^2 ; a = -696, b = 2004.5, C = -1431.3$$

$$\therefore \alpha_2 = A(M_u) + B(M_u)\phi_2 + C(M_u)\phi_2^2$$



1D equation for predicting work coefficient

$Z_v = 9$

$$\alpha_2 = k(M_u, \phi_2) \begin{cases} \alpha_2 = a + bM_u + \bar{B}(M_u)\phi_2 \\ = 94.37 - 0.197M_u - 100.418\phi_2 & ; 0.3 < M_u < 0.8 \\ \alpha_2 = A(M_u) + B(M_u)\phi_2 + C(M_u)\phi_2^2 & ; 0.8 < M_u < 1.2 \end{cases}$$

$$\frac{\psi_0}{2} = \phi_2 \tan \alpha_2$$

$$\psi_0 = 2\phi_2 \tan [k(M_u, \phi_2)]$$

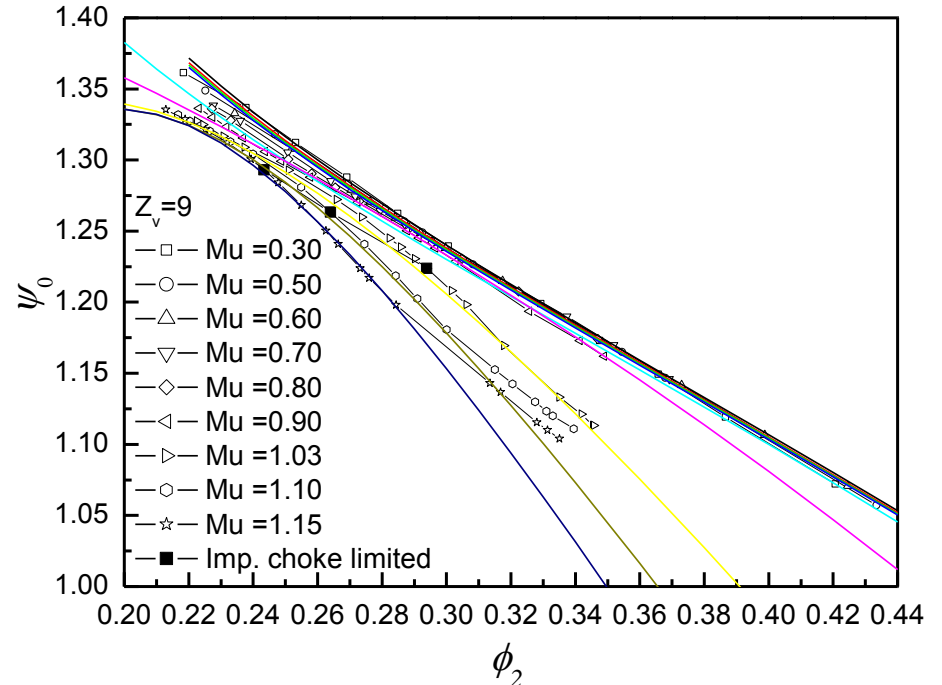
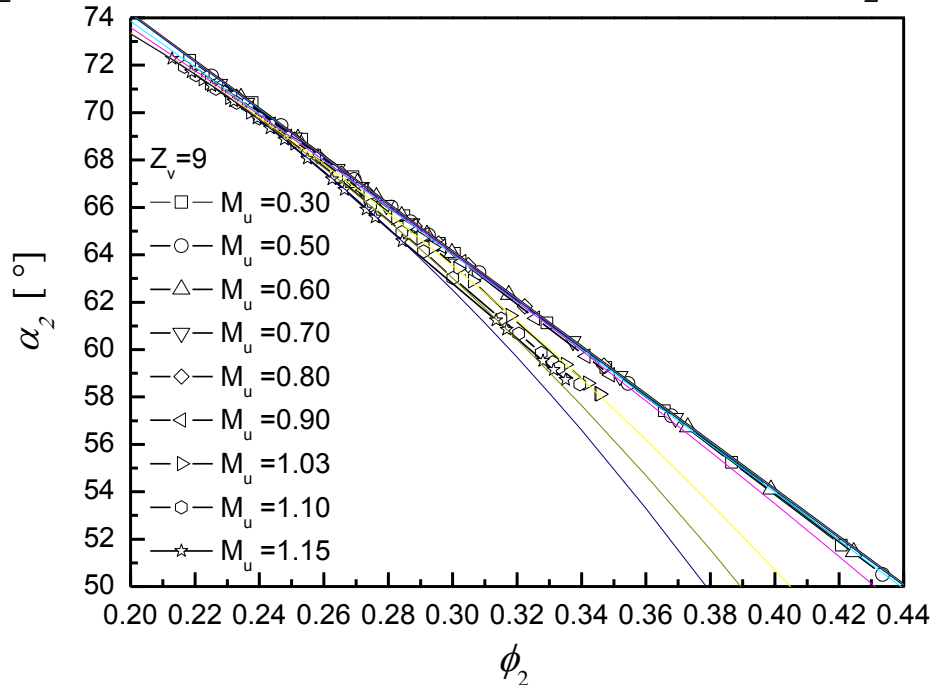
$$\psi_0 = p(M_u, \phi_2)$$

Parabola fit (x, y)

$$A = a + bM_u + cM_u^2 ; a = 97.72, b = 20.325, C = -32.631$$

$$B = a + bM_u + cM_u^2 ; a = 48.98, b = -558.8, C = 474.345$$

$$C = a + bM_u + cM_u^2 ; a = -696, b = 2004.5, C = -1431.3$$



1D equation for predicting work coefficient

$$Z_v = 12$$

$$\alpha_2 = k(M_u, \phi_2) \begin{cases} \alpha_2 = a + bM_u + \bar{B}(M_u)\phi_2 \\ = 94.45 - 0.546M_u - 100.242\phi_2 & ; 0.3 < M_u < 0.8 \\ \alpha_2 = A(M_u) + B(M_u)\phi_2 + C(M_u)\phi_2^2 & ; 0.8 < M_u < 1.2 \end{cases}$$

$$\frac{\psi_0}{2} = \phi_2 \tan \alpha_2$$

$$\psi_0 = 2\phi_2 \tan [k(M_u, \phi_2)]$$

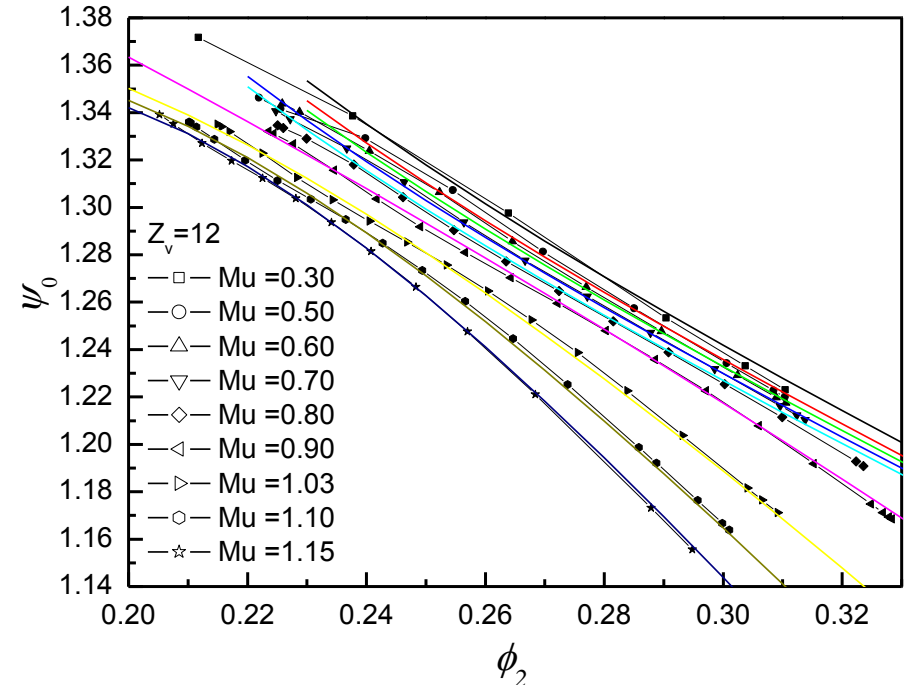
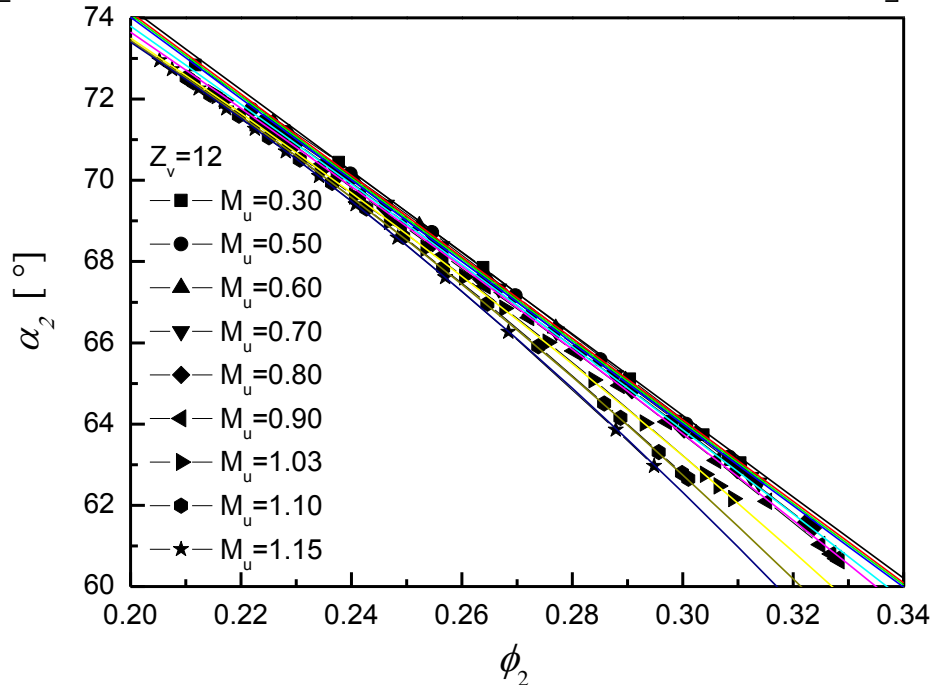
$$\psi_0 = p(M_u, \phi_2)$$

Parabola fit (x, y)

$$A = a + bM_u + cM_u^2 ; a = 85.668, b = 29.443, C = -27.849$$

$$B = a + bM_u + cM_u^2 ; a = 59.662, b = -449.97, C = 343.48$$

$$C = a + bM_u + cM_u^2 ; a = -541.7, b = 1419.9, C = -987.37$$





Reynolds number function of flow angle at impeller exit

$$0.3 < M_u < 0.8$$

Linear fit $(x, y) = (\alpha_2, Re_2)$

$$Re_2 = A + B\alpha_2$$

parameters A, B

A : intercept

B : slope

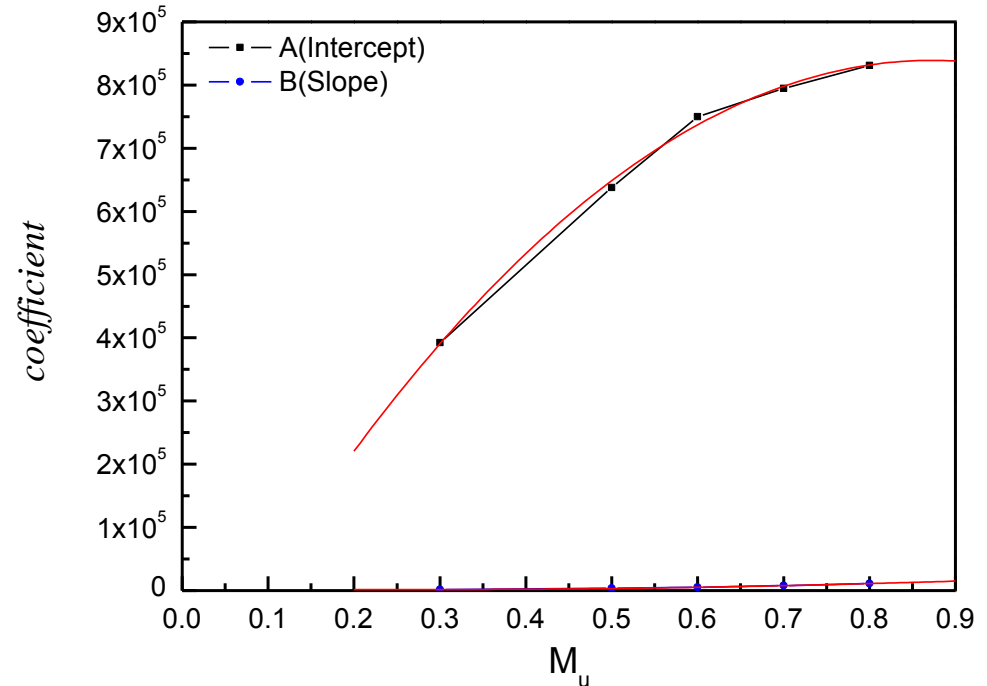
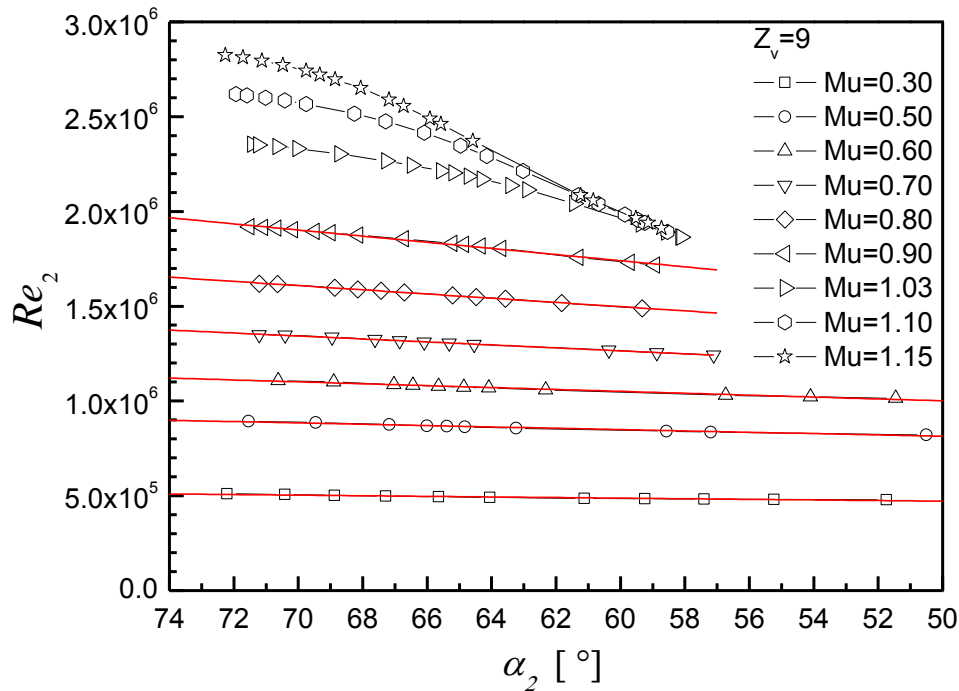
$$Z_v = 9$$

Parabola fit (x, y)

$$A = a + bM_u + cM_u^2 ; a = -2.01E + 5, b = 2.38E + 6, C = -1.36E + 6$$

$$B = a + bM_u + cM_u^2 ; a = 4.254E + 3, b = -1.91E + 4, C = 3.46E + 4$$

$$\therefore Re_2 = A(M_u) + B(M_u)\alpha_2$$



Reynolds number function of flow angle at impeller exit

$$0.8 < M_u < 1.2$$

$$\text{Parabola fit } (x, y) = (\alpha_2, Re_2)$$

$$Re_2 = A + B\alpha_2 + C\alpha_2^2$$

parameters A, B, C

$$Z_v = 9$$

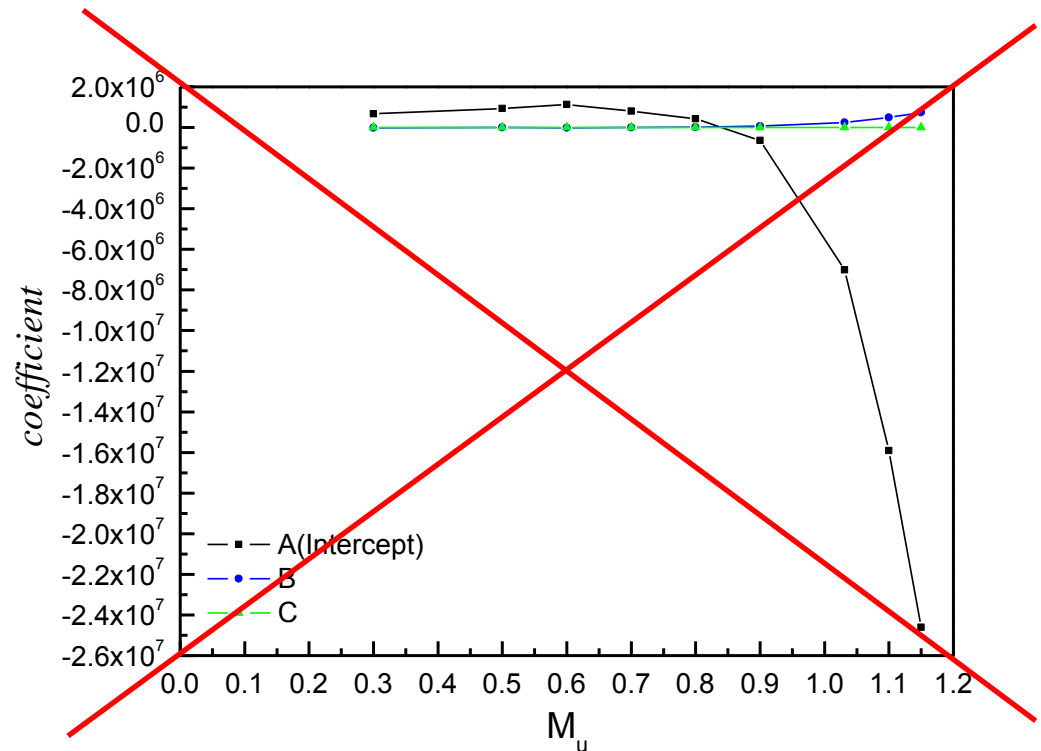
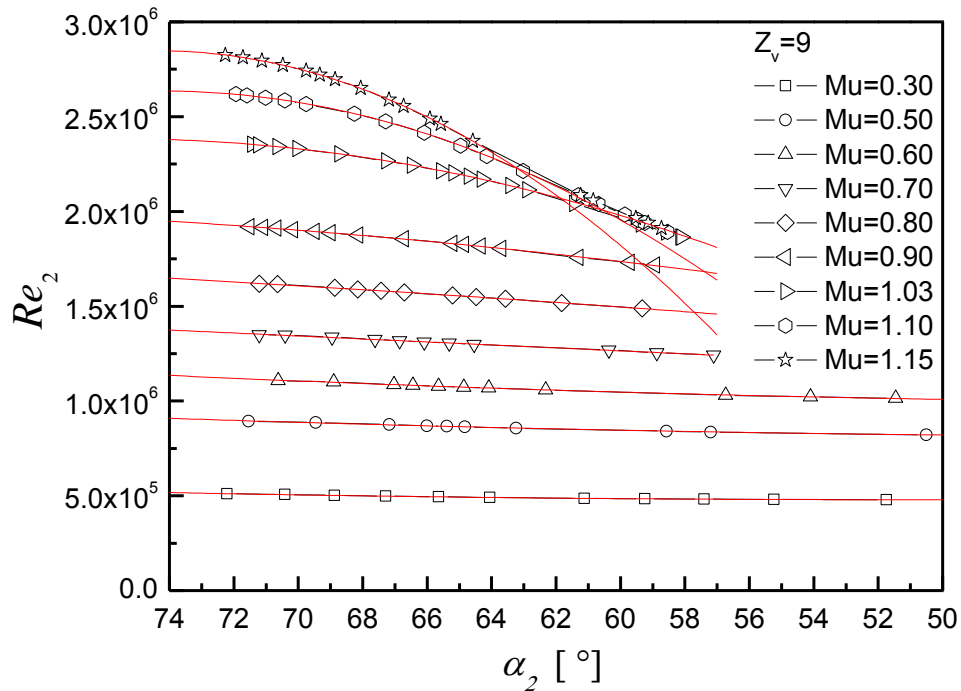
determine A, B, C from 3 point of test or CFD

\therefore all curves are shaped parabola fit

$$(\alpha_2, Re_2)_1, (\alpha_2, Re_2)_2, (\alpha_2, Re_2)_3$$

$$Re_2 = A + B\alpha_2 + C\alpha_2^2$$

Mu	A(Intercept)	B	C
0.8	428304.9	23436.26	-94.0204
0.9	-6.42E+05	59452.49	-330.27
1.03113249	-7.01E+06	247965.3	-1636.95
1.1	-1.59E+07	500285.9	-3371.06
1.15	-2.46E+07	737054.4	-4953.38



Reynolds number function of flow angle at impeller exit

$$Z_v = 9$$

$$Re_2 = f(M_u, \alpha_2) \begin{cases} Re_2 = A(M_u) + B(M_u)\alpha_2 & ; 0.3 < M_u < 0.8 \\ Re_2 = A + B\alpha_2 + C\alpha_2^2 & ; 0.8 < M_u < 1.2 \end{cases}$$

Parabola fit (x, y) ; $0.3 < M_u < 0.8$

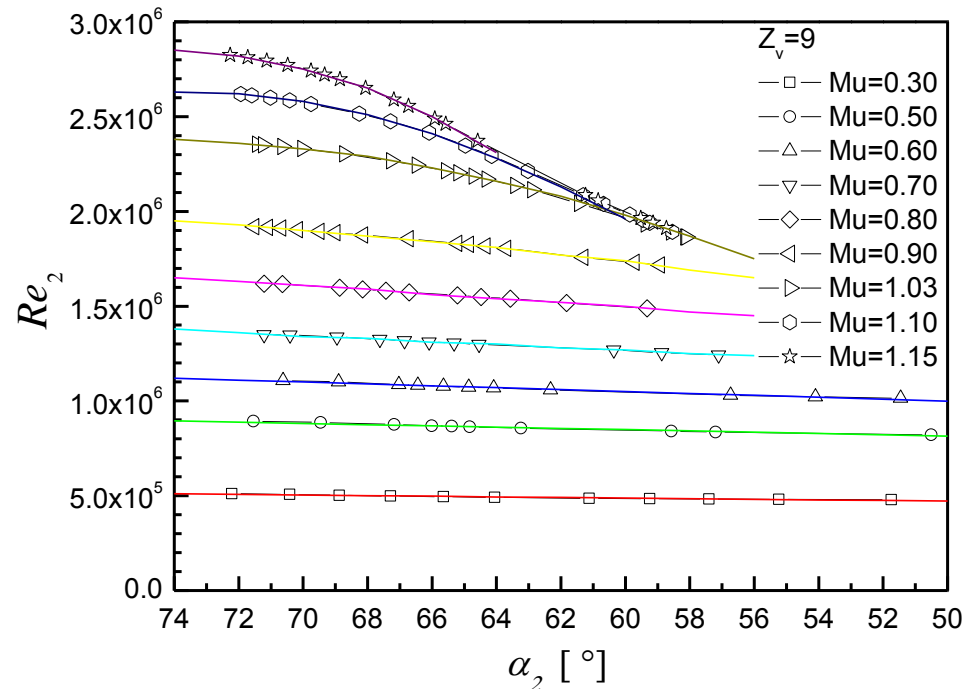
$$A = a + bM_u + cM_u^2 ; a = -2.01E+5, b = 2.38E+6, C = -1.36E+6$$

$$B = a + bM_u + cM_u^2 ; a = 4.254E+3, b = -1.91E+4, C = 3.46E+4$$

A, B, C are determined by 3 data of test or CFD ; $0.8 < M_u < 1.2$

$$(\alpha_2, Re_2)_1, (\alpha_2, Re_2)_2, (\alpha_2, Re_2)_3$$

$$Re_2 = A + B\alpha_2 + C\alpha_2^2$$



Flow coefficient correlation & Prediction of work coefficient

$$\phi = \frac{\dot{m}}{\rho_{01} \frac{\pi}{4} D_2^2 U_2} = \frac{\rho_2 A_2 c_2 \cos \alpha_2}{\rho_{01} \frac{\pi}{4} D_2^2 U_2} = \frac{\rho_2 \pi D_2 b_2 c_2 \cos \alpha_2}{\rho_{01} \frac{\pi}{4} D_2^2 U_2} = \frac{\rho_2 4b_2 c_2}{\rho_{01} D_2 U_2} \cos \alpha_2$$

$$= \frac{2\mu_2 Re_2}{\rho_{01} D_2 U_2} \cos \alpha_2 \left[Re_2 = \frac{\rho_2 c_2 2b_2}{\mu_2} = l(\alpha_2) = l(k(\phi_2)) \right]$$

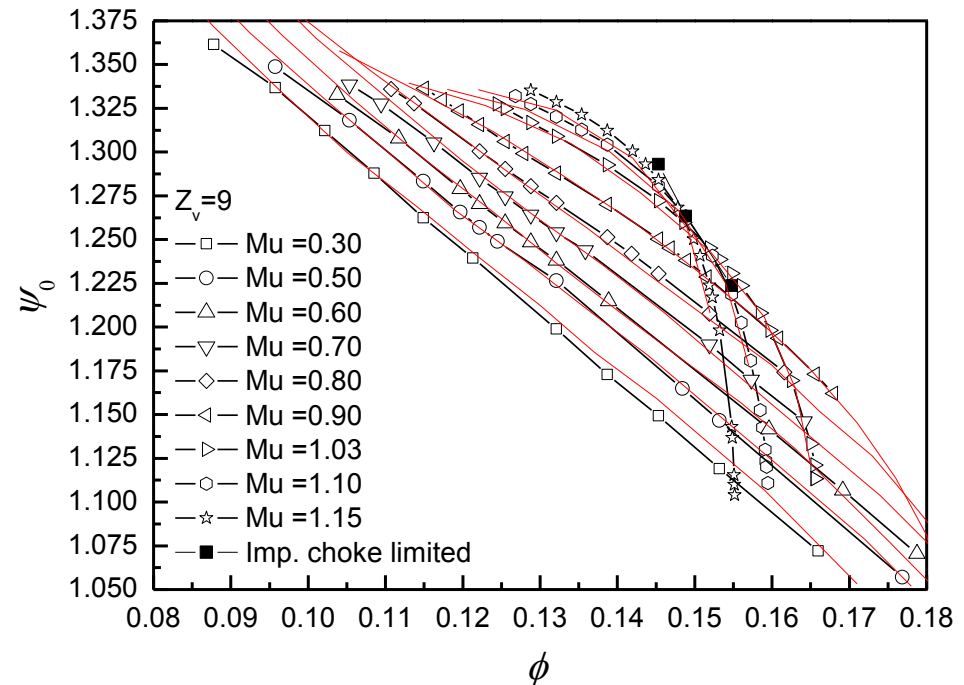
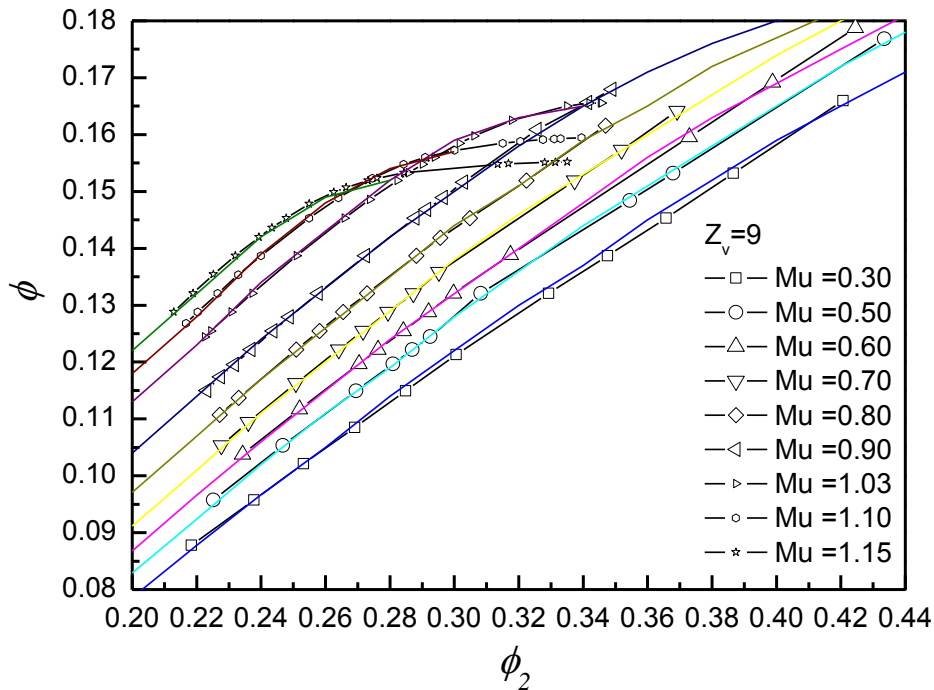
$$= \frac{2\mu_2 l(k(\phi_2))}{\rho_{01} D_2 U_2} \cos \{k(\phi_2)\}$$

$$= q(\phi_2)$$

$$\frac{\psi_0}{2} = \phi_2 \tan \alpha_2$$

$$\psi_0 = 2\phi_2 \tan \{k(\phi_2)\}$$

$$\psi_0 = p(\phi_2) = p(q^{-1}(\phi)) = f(\phi)$$



Flow coefficient correlation & Prediction of work coefficient

$$\phi = \frac{\dot{m}}{\rho_{01} \frac{\pi}{4} D_2^2 U_2} = \frac{\rho_2 A_2 c_2 \cos \alpha_2}{\rho_{01} \frac{\pi}{4} D_2^2 U_2} = \frac{\rho_2 \pi D_2 b_2 c_2 \cos \alpha_2}{\rho_{01} \frac{\pi}{4} D_2^2 U_2} = \frac{\rho_2 4b_2 c_2}{\rho_{01} D_2 U_2} \cos \alpha_2$$

$$= \frac{2\mu_2 Re_2}{\rho_{01} D_2 U_2} \cos \alpha_2 \left[Re_2 = \frac{\rho_2 c_2 2b_2}{\mu_2} = l(\alpha_2) = l(k(\phi_2)) \right]$$

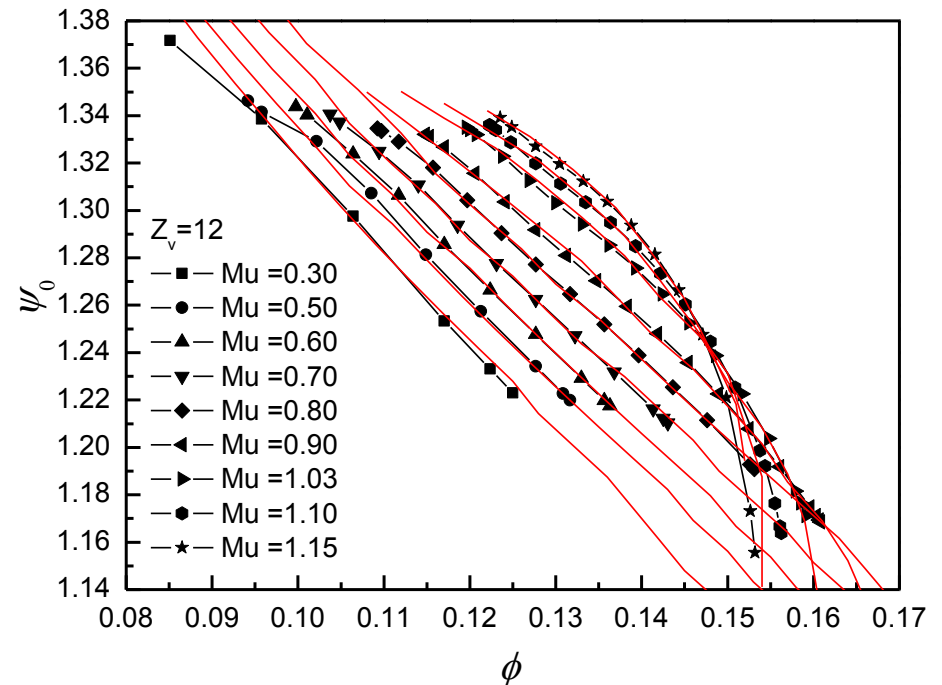
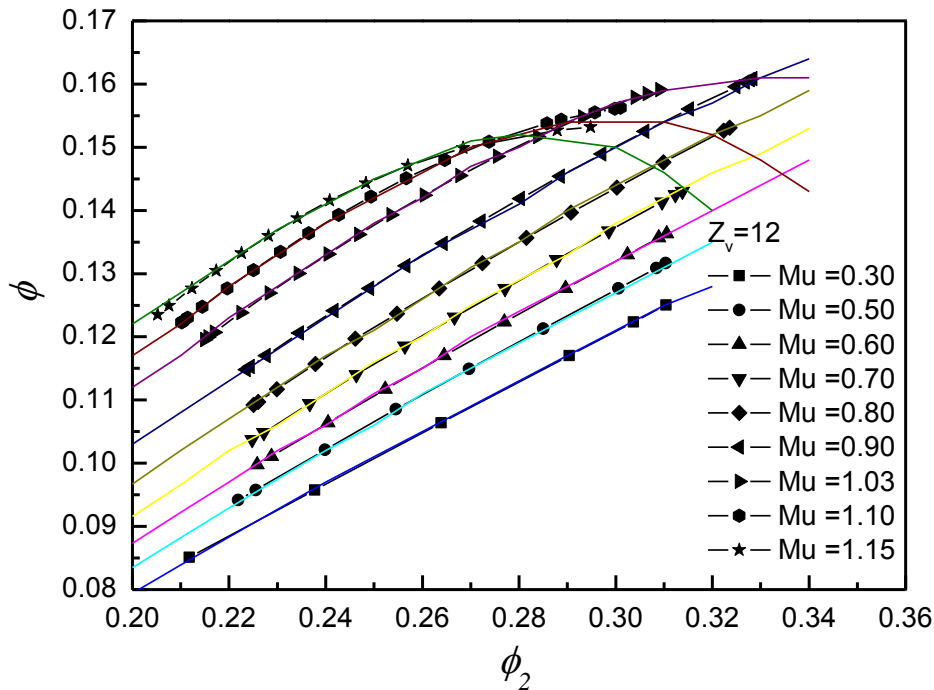
$$= \frac{2\mu_2 l(k(\phi_2))}{\rho_{01} D_2 U_2} \cos \{k(\phi_2)\}$$

$$= q(\phi_2)$$

$$\frac{\psi_0}{2} = \phi_2 \tan \alpha_2$$

$$\psi_0 = 2\phi_2 \tan \{k(\phi_2)\}$$

$$\psi_0 = p(\phi_2) = p(q^{-1}(\phi)) = f(\phi)$$



- ❖ **Centrifugal compressor**
 - **Centrifugal compressor($Z_v=12$) data analysis**
 - **Imp.-diff. matching correlation of the number of vanes**

- ❖ **Vaned diffuser**
 - **Calculation of diverse vaned diffusers($\xi = 70^\circ$) by CFD**
 - **Investigation of design parameters**

- ❖ **Accumulation of data base : CFD(by SNU), test(by STW)**

- ❖ **Development of the in-house program(by MATLAB)**
 - **1D aerodynamic design model**
 - **Performance prediction model**



Thank you for your attention.