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### Simulation Research of Driving Schemes for a Dynamic Calibration System of Fuel Turbine Flowmeters

**Bin Wang** Nanjing University of Aeronautics and Astronautics, China October 7, 2015



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#### **ENERGY MEASUREMENT**



# AEROSPACE & AVIATION

#### **INDUSTRIAL PROCESSES**



#### BIOTECHNOLOGY





#### **SEMI-PHYSICAL EXPERIMENTATION**







long-term usage — performance degradation

fluctuation of meter coefficient —> calibration needed



#### **Theoretical basis**

$$K_{s}Q^{2}-Q\omega=K_{d}\dot{\omega}$$

#### **Review of research**







#### Principle and method presented by our research





#### **Alternative actuating solutions:**



Torque motor driving NFV.

#### Piezoelectric actuator driving NFV.









Structure of nozzle-flapper stimulating system



Piezoelectric stack actuator 
$$x_0 = nx_n = n\left(\frac{t_n S_{33}^E F_s}{A_s} + d_{33}U\right)$$
  
 $x_o = x_0 \frac{K_p}{K_p + K_f} F_o = \frac{K_p K_f}{K_p + K_f} x_0$   
Control cavity  $q_L = q_g - q_k = C_{dg} A_g \sqrt{\frac{2}{\rho} (p_s - p_c)} - C_{dk} A_N \sqrt{\frac{2}{\rho} p_c}$   
 $A_g = \frac{\pi}{4} D_g^2 \quad A_N = \pi D_N (x_{f0} + x_f)$   
Flowmeter and pressure difference transducer

$$K_s Q^2 - Q\omega = K_d \dot{\omega}$$

$$G(s) = \frac{A_2 s^2 + A_1 s + A_0}{s^2 + B_1 s + B_0}$$



#### **Equations of nozzle-flapper stimulating system**

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#### **Parameters**

Elements	Parameters	Value	Units	Origin
Piezoelectric	excitation voltage $U_e$	0~150	V	datasheet
stack actuator	electrostatic capacity Ce	1.6e-6	F	datasheet
	stiffness Kp	1.2e7	N/m	datasheet
	number of stack n	420	-	datasheet
	piezoelectric constant d <sub>33</sub>	4.4e-10	pC/N	datasheet
Flapper	amplification factor n <sub>a</sub>	10	_	designed
	length of flapper L	0.12	m	designed
	moment of inertia J	1.63e-4	<mark>kg∙m</mark> ²	calculated
	elasticity modulus E	2e11	Pa	datasheet
	stiffness K <sub>f</sub>	5e6	N/m	calculated
Nozzle	inner diameter d <sub>i</sub>	2.5	mm	designed
	external diameter de	5	mm	designed
Flowmeter	time constant τ	0.02	S	estimate
	fuel density p	800	kg/m <sup>3</sup>	datasheet
Control cavity	reference orifice D <sub>0</sub>	1.1	mm	designed 🦨
/24	draining orifice D <sub>d</sub>	2	mm	designed



#### Load analysis of the flapper



Disp. of piezoelectric stack system

Disp. of torque motor system

#### **Comparison between these two solutions**





#### Performance of torque motor stimulating system







#### a. Pressure difference

**b. Flowrate** 

#### Performance of piezoelectric stack stimulating system





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Response of the adopted system to 80V excitation voltage





Calibrated flow rate with different excitation voltage



Calibrated flow rate vs excitation voltage Driving force vs excitation voltage





Output force and disp. vs flapper stiffness Acceleration time vs spring preload



1. As the relatively small output torque the torque motor can provide, diameter of the nozzle can't be set to a large enough value. Oscillation of the flapper can't be neglected.

2. The piezoelectric stack actuator (PSA) can produce much greater force to balance the flow force acting on the flapper. Larger size nozzles are feasible. Piezoelectric structure effectively controls the high-frequency oscillation of the flapper.

3. Dynamic calibration system driven by piezoelectric-stack provides a faster and larger excitation flow than by the torque motor. In the mean time, the PSA can control the flapper flutter more effectively.

4. Structural parameters of the flapper needs to be carefully designed on the basis of a thorough understanding of the characteristics of PSA. As the preload spring is essential for stabilization of output displacement, its magnitude should be considered carefully.







#### Virtual prototype of experimental setup



**Nozzle-flapper valve under in machining** 





**Piezoelectric-stack and its power supply** 

FT Series flowmeters to be calibrated

**Primary equipments prepared** 



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