

Eddy-Current Signal Processing Applied to Aircraft Fuel Tubes Maintenance

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Air Transat Flight TSC 236

- August 24th, 2001, evening.
- North Atlantic Ocean, Azores Islands.
- FL 350, Airbus A330-243.
- FR Toronto (YYZ) TO Lisbon (ZYL).



Air Transat Flight TSC 236

→ Actually FR Toronto TO Azores...



Airbus A330 - Engine Number 2

→ Intriguing oil parameters:

- Low oil temperature.
- High oil pressure.
- Low volume in the oil reservoir.



	ENGINE Nr 1	ENGINE Nr 2
OIL TEMPERATURE	110°C	65°C
OIL PRESSURE	80 psi	150 psi
OIL QUANTITY	17 liters	14 liters

Rolls Royce RB 211 Trent 772B



Airbus A330 Engines

- Normal engines parameters, but...
- Inexplicable low FOB (Fuel on Board) quantity readings.



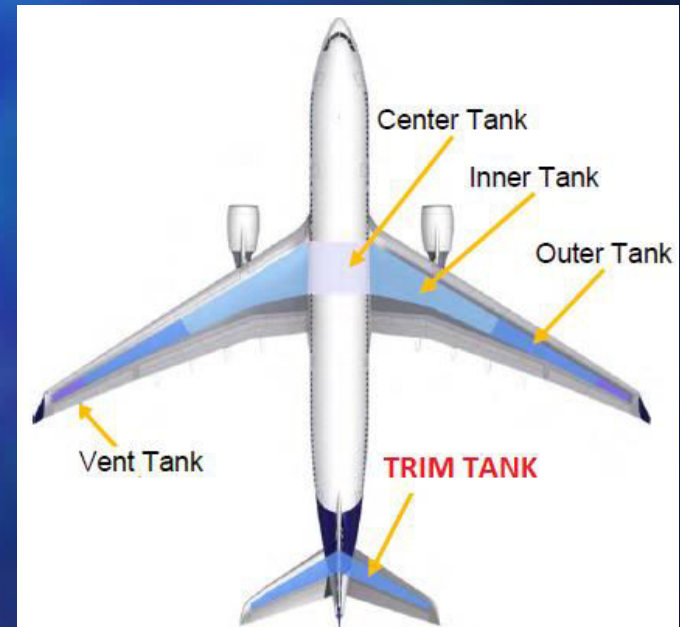
E / WD Screen

→ In addition, the following message illuminate in the E / WD (Engine Warning Display):



T TK XFRD

→ which means:
“Trim Tank Empty”



The Azores Glider

- At 06:13 pm of that day the engine number 2 flamed out.
- 13 minutes later, the left engine flamed out too.
- With no fuel, 65 nautical miles away from Lajes Airport, Terceira Island, Azores and at 34,500 ft altitude, the ~ 160 tons aircraft became an immense glider.



Lajes Airport Approach



Azores Glider Landing

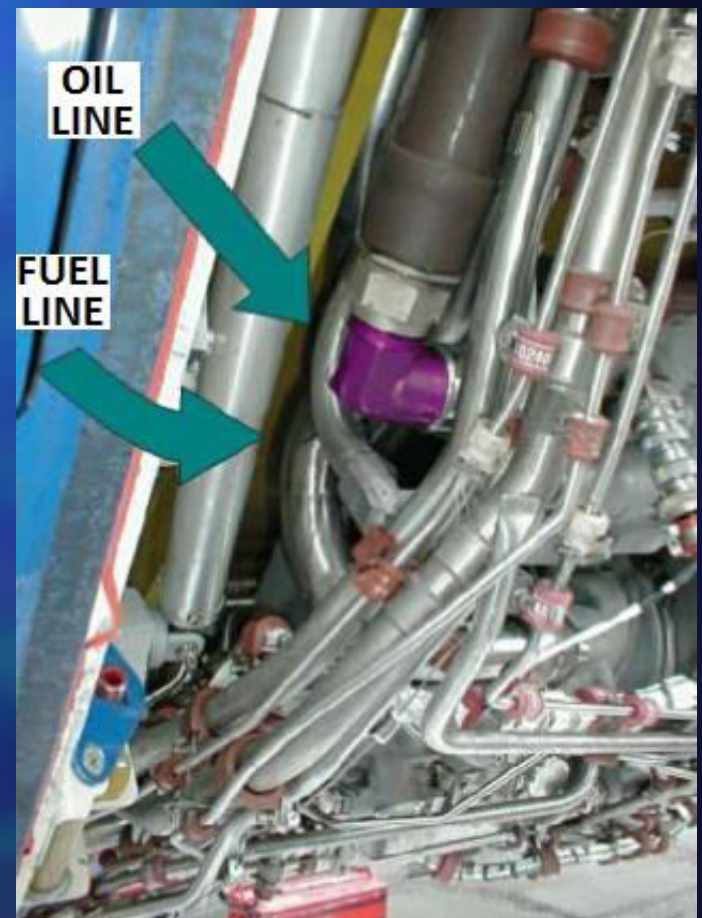
- Despite small fires in the landing gear and minor structural damage, the emergency landing was ok at RWY 33.
- However, eighteen people suffered injuries during the emergency evacuation.



Accident Investigation

→ Investigators found that the strange parameters of engine Nr 2 were caused by a fuel leak through a crack in the fuel line downstream of the fuel-oil heat exchanger.

OIL	ENGINE Nr 1	ENGINE Nr 2
TEMP.	110°C	65°C
PRESS.	80 psi	150 psi
VOLUME	17 liters	14 liters



Accident Investigation

- The fuel leak rate reached 13 ton/h.
- That's the reason why the aircraft got no fuel.



Accident Investigation

→ The crack was a result of a hard contact between fuel and oil lines due to a non appropriate routine maintenance dozens of flight-hours before.



Why an aircraft get no fuel ?

→ In the aviation history, a lot of accidents have been occurred due to lack of fuel.

→ The main reasons commonly are:

- wrong flight range calculations
- fuel indicators malfunction.
- bad weather conditions.
- hard air traffic.
- navigation errors, etc...

Non Destructive Testing

→ The rupture of a fuel tube is a quite rare reason for lack fuel because the tubes are properly inspected.

→ The aircraft tubing is inspected through several Non Destructive Testing methods:

- Liquid Penetrant;
- Ultra Sound;
- X-Ray;
- Eddy-Current.

Eddy-Current Testing

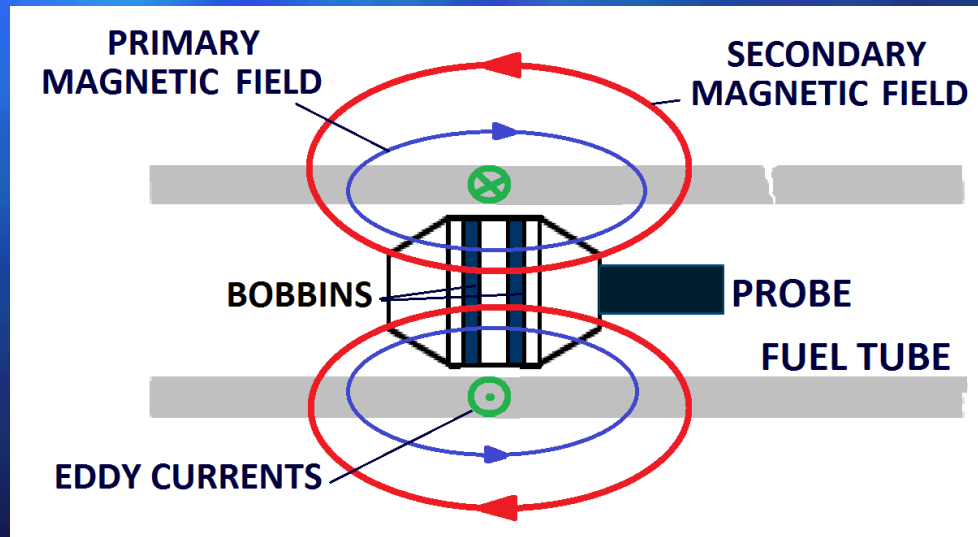
- The Eddy-Current Testing (ECT) is an electromagnetic method widely used in aircraft inspection.
- The defects identified through ECT allows important inspection decisions and guarantees the aircrafts safe operations.



Eddy-Current Testing (ECT)

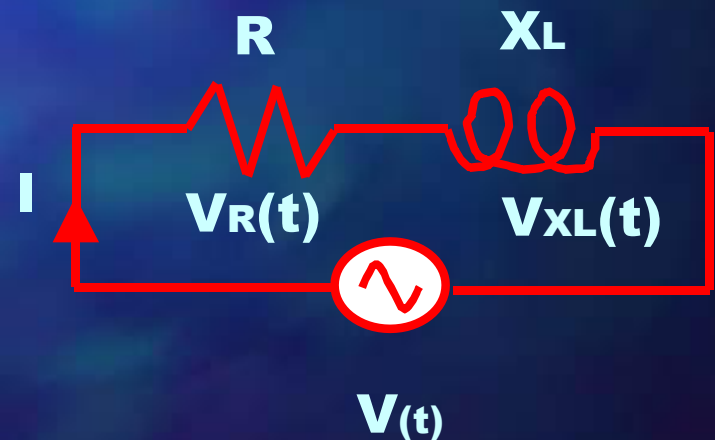
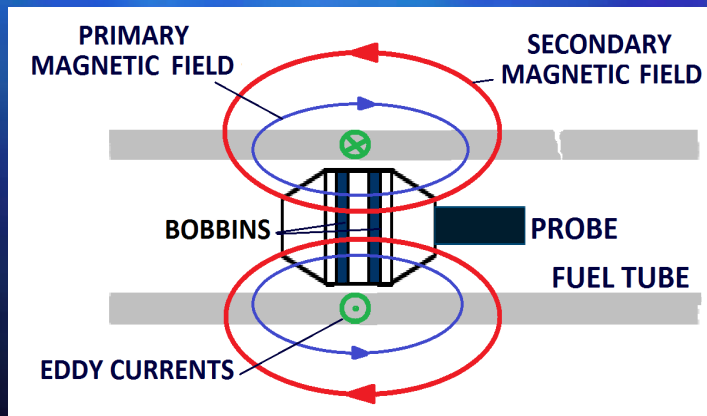
→ The ECT is based on the introduction of an alternate current bobbin coil probe in the fuel tube.

→ A primary magnetic field is generated.



Eddy-Current Testing (ECT)

- The primary magnetic field induces circular Eddy Currents in the fuel tube which generates a secondary magnetic field opposite to the primary field.
- The electric circuit have resistive (R) and inductive (X_L) components.

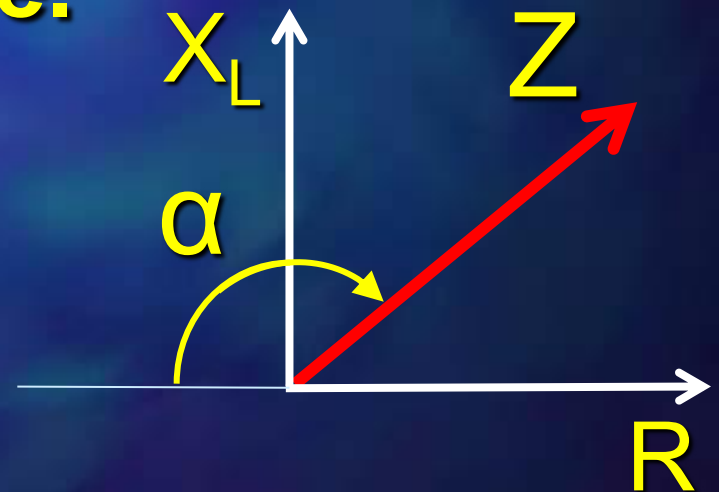


Eddy-Current Testing (ECT)

→ By using the complex plan to represent the resistance R and the inductive reactance X_L , the impedance of the circuit is represented by a vector which length is Z .

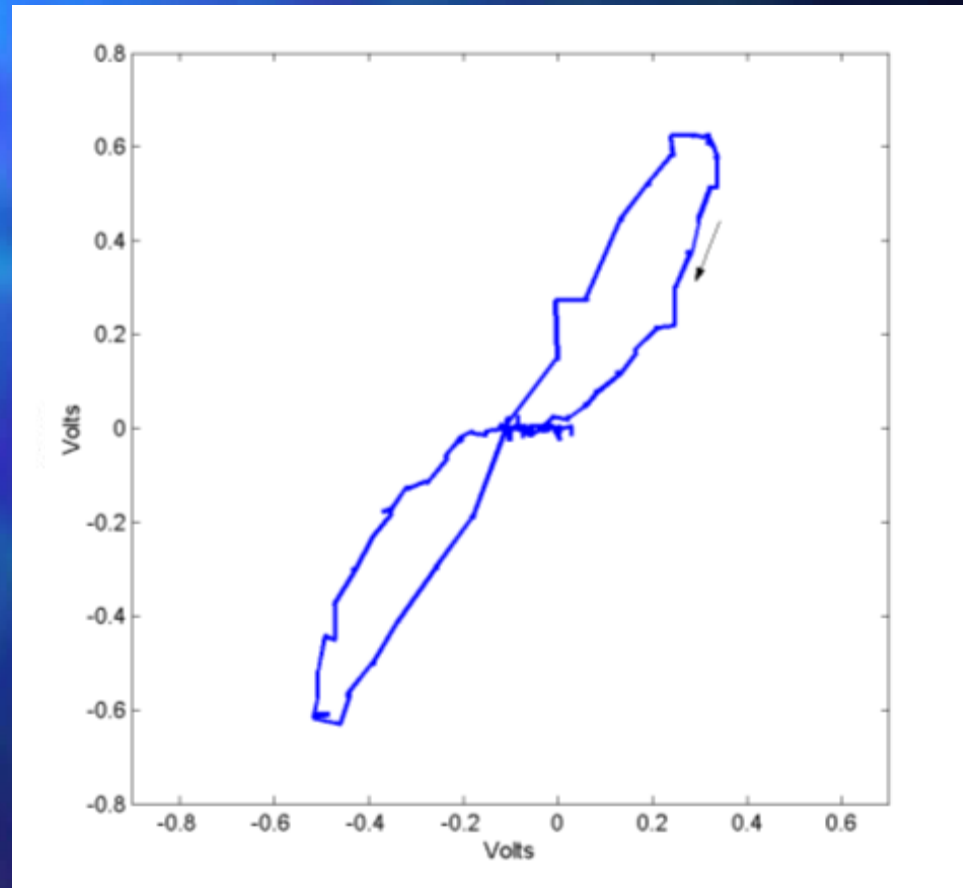
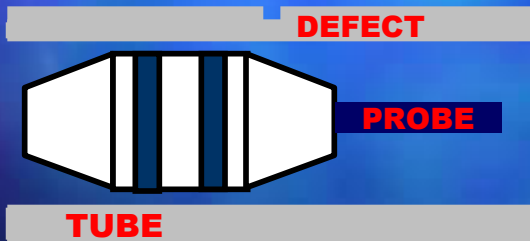
→ α is the Phase Angle.

$$Z = \sqrt{R^2 + X_L^2}$$



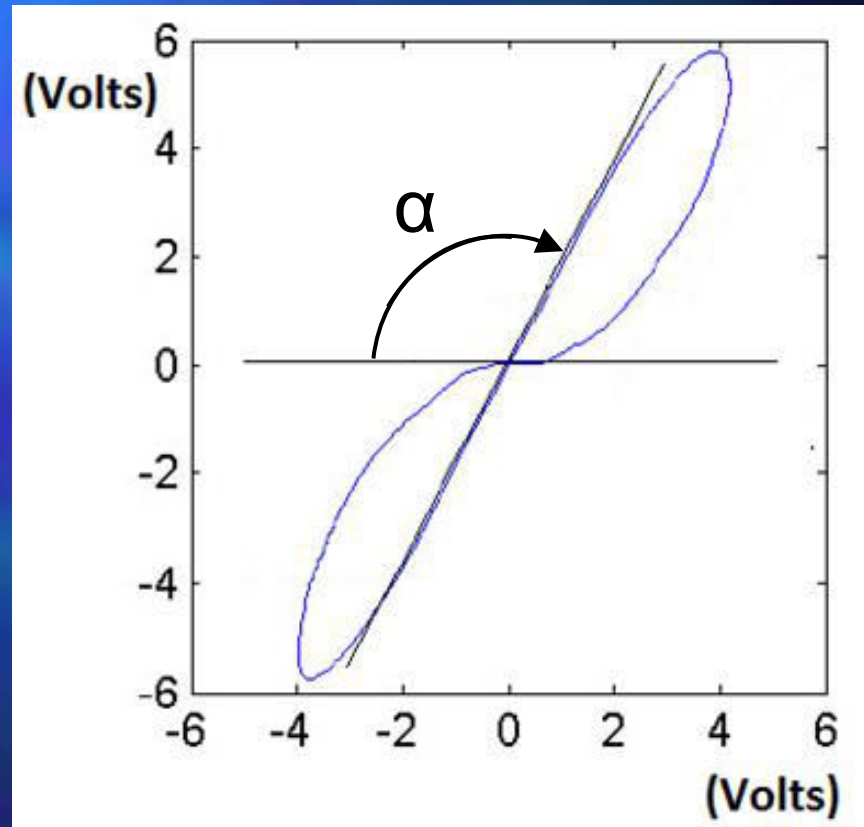
Eddy-Current Testing (ECT)

→ When the probe passes through a defect, the impedance forms an “8” shaped Lissajous figure.



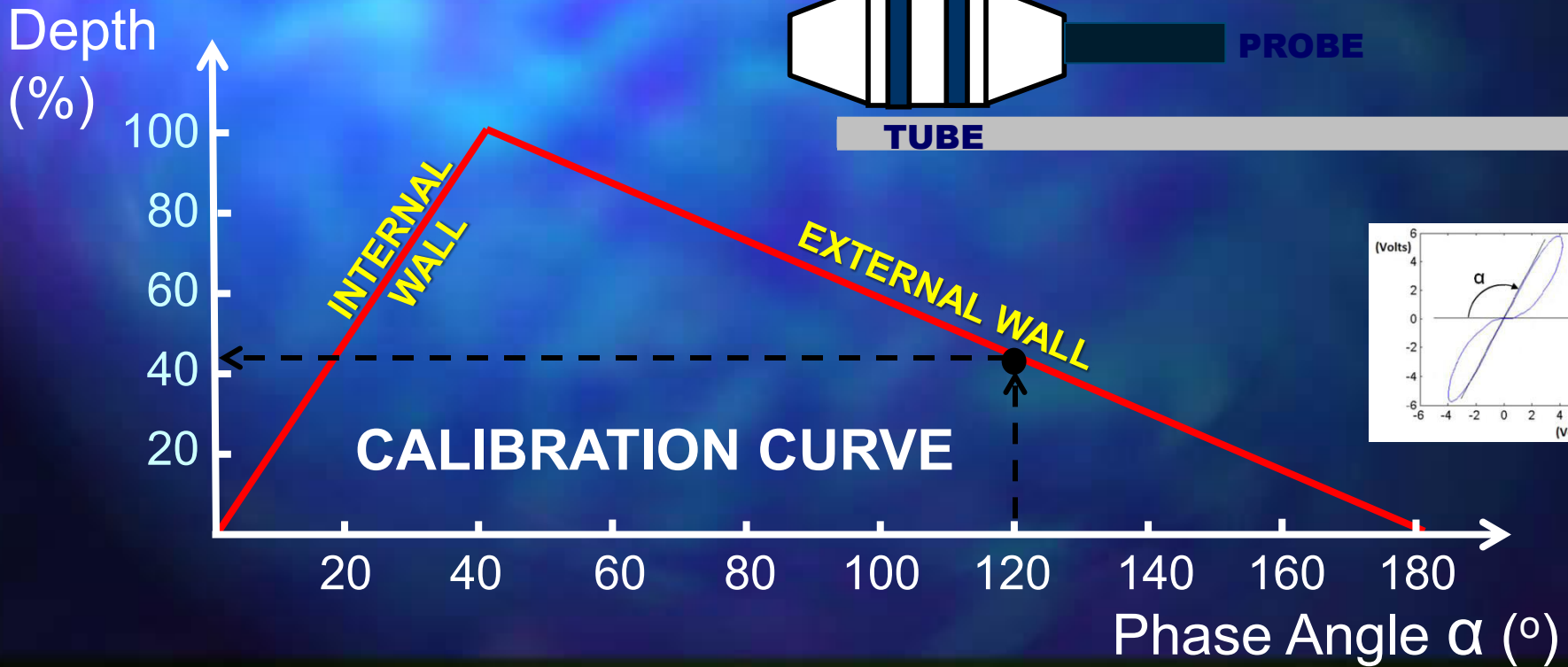
Eddy-Current Testing (ECT)

→ One of the most important signal characteristics is the Phase Angle α formed by a horizontal line and the straight line common to the two petals of the Lissajous figure, clockwise measured.



Defect Depth and Localization

→ The Phase Angle determines the depth of the defect and if it is internal or external to the fuel tube by using the Calibration Curve.



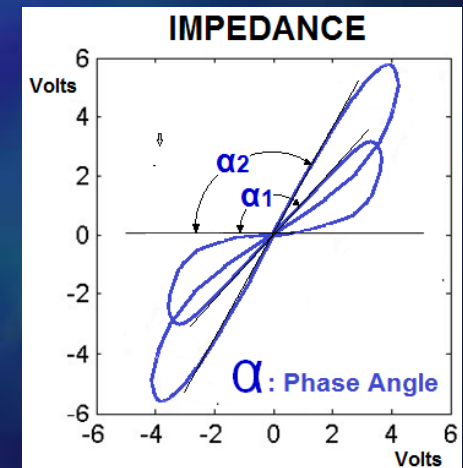
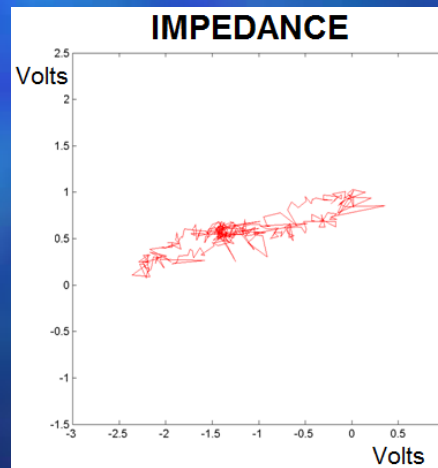
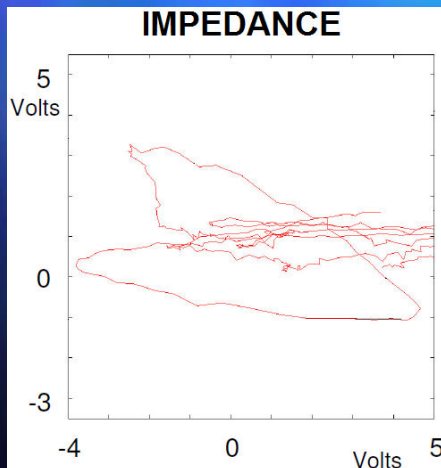
Eddy Current Signal Noises

- ECT signals contain noises which can strongly interfere in their analysis.
- These noises have many origins:

Material
Itself

Data
Acquisition

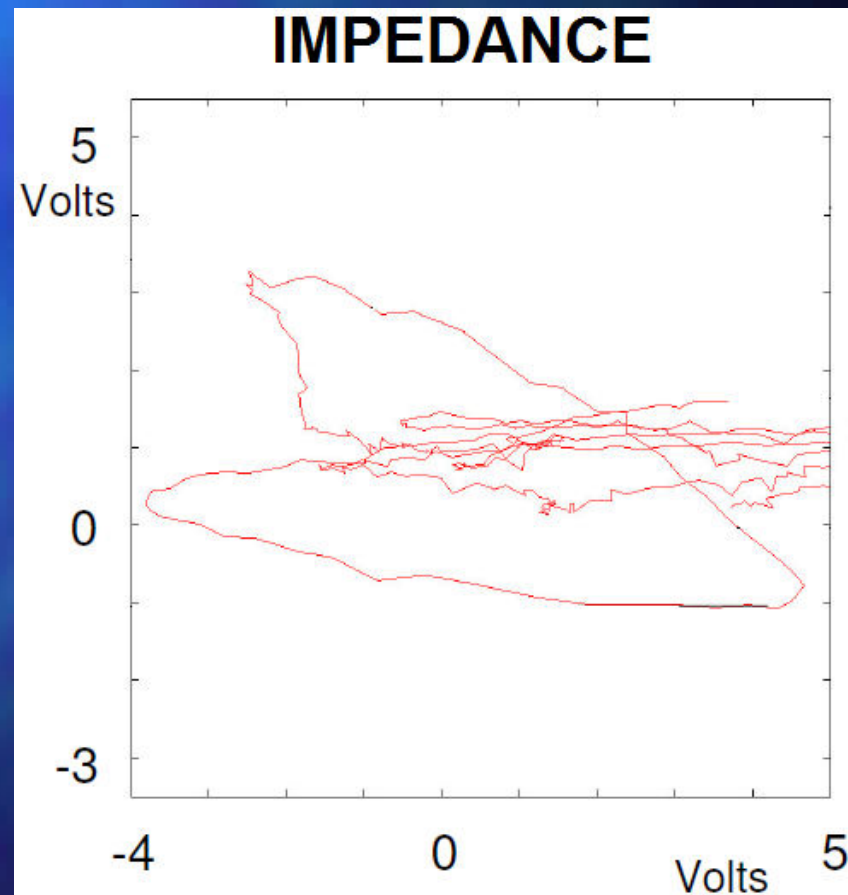
Probe
Wobble



Eddy Current Material Noise

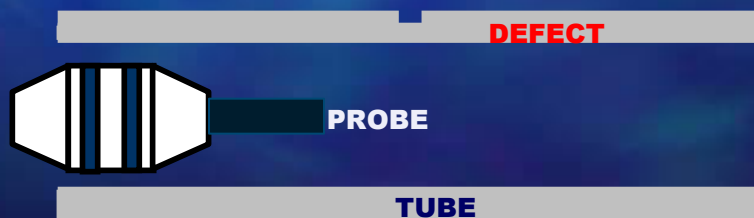
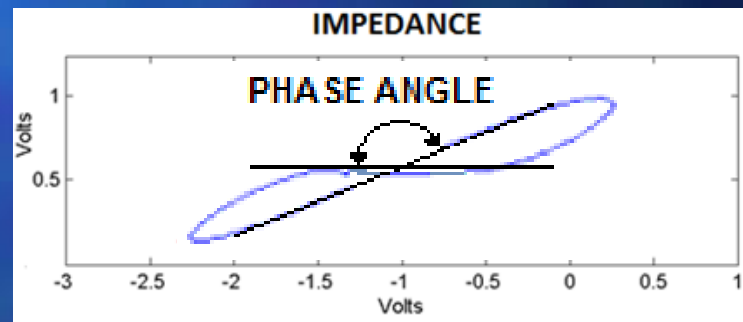
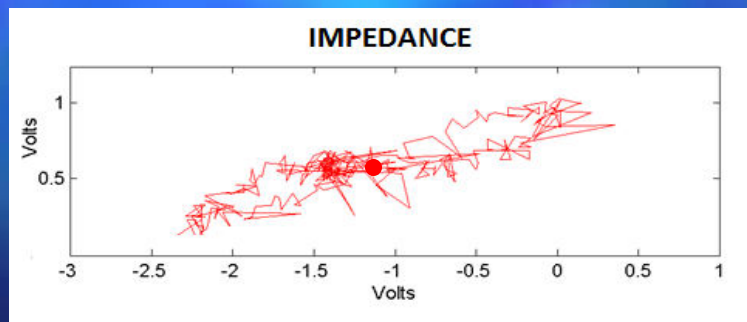
→ This noise is originated in the variations of magnetic properties of fuel tube material.

→ It produces distortions in the Lissajous figures to such an extension that Phase Angle reading is simply impossible at all.



Data Acquisition Noise

- The instrumentation noise produces a blurry Lissajous figure which can jeopardize Phase Angle determination.
- The Wavelet Transform methodology is used for the removal of this noise.



The Wavelet Transform (WT)

- WT is a time-scale signal transformation which combines the use of variable time and scale windows.
- When lower frequency information is needed, large time intervals are used and vice versa.



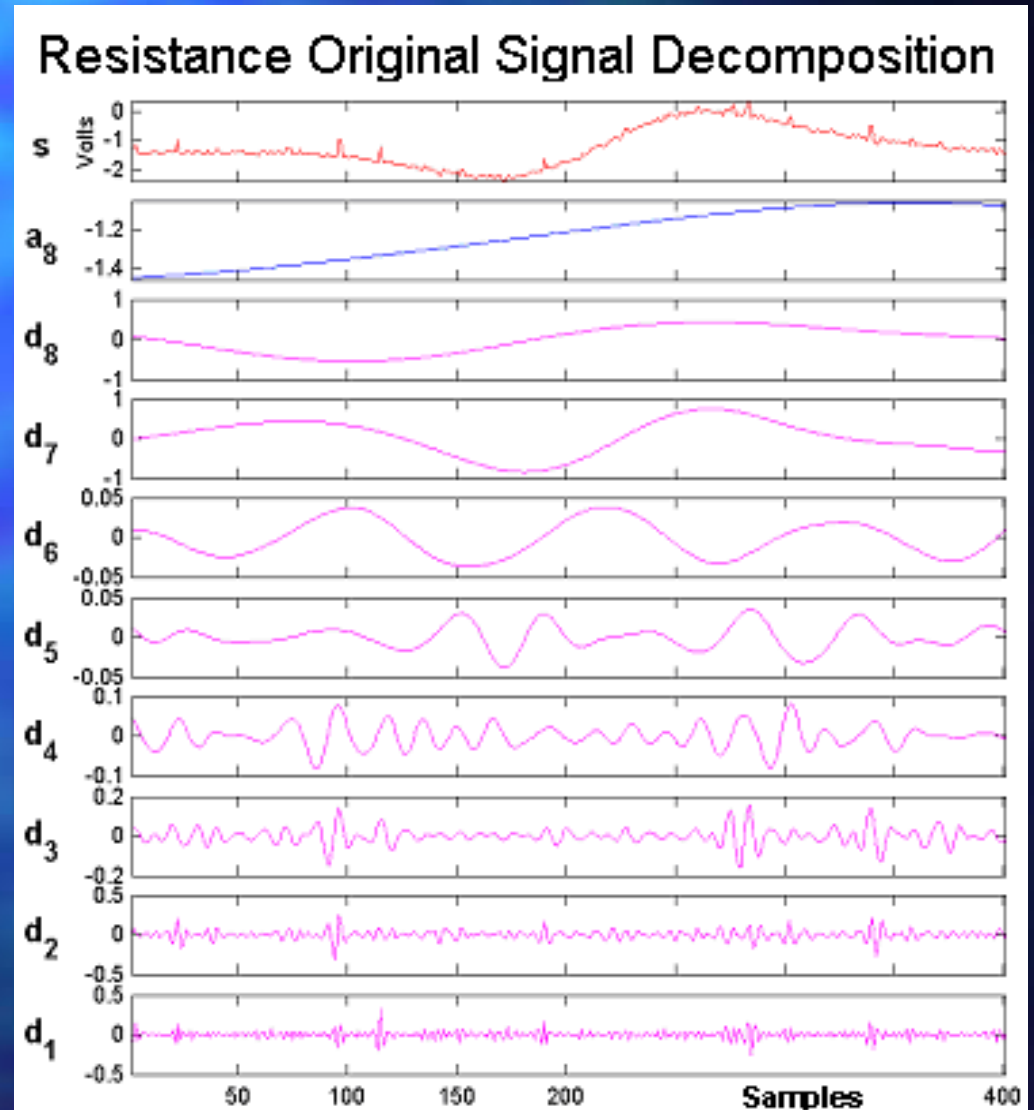
The Wavelet Transform (WT)

→ The Wavelet Transform is very sensitive to discontinuities in time domain, which is typical of Eddy Current signals.

→ It is a method where the transformed time-frequency domain can be varied, allowing for a more detailed description of the time-frequency behavior of the signal being analyzed.

The Wavelet Transform (WT)

→ Through the Matlab tool, the signal can be decomposed and analysed in how many separate frequency bands are needed.



The Wavelet Transform (WT)

→ For a given Eddy-Current signal $s(x)$, the WT is the convolution of $s(x)$ with a set of Wavelet functions $\psi_{a,b}(x)$ resulting in a set of coefficients $C_{a,b}$:

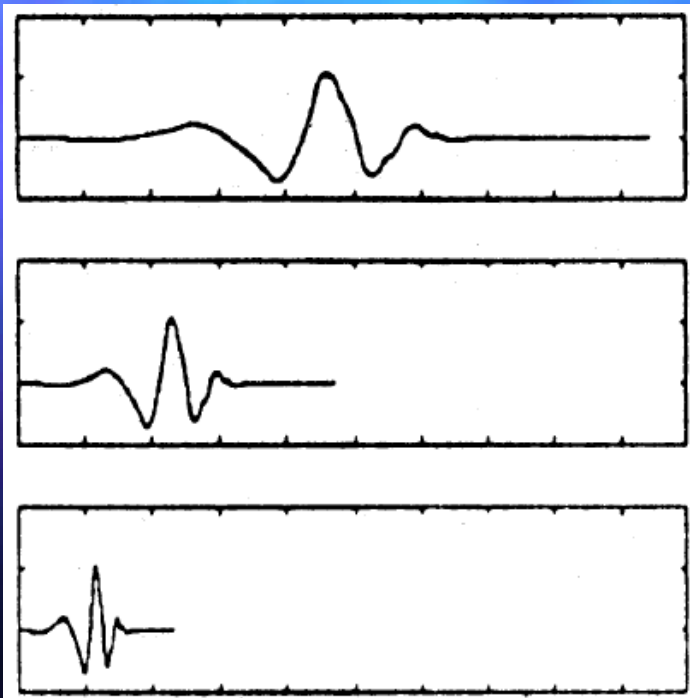
$$C_{(a,b)} = \int_{-\infty}^{+\infty} s(x) \psi_{a,b}(x) dx$$

→ A Wavelet function $\psi_{a,b}(x)$ has two characteristic parameters: the *scale* “ a ” and the *position* “ b ” which change continually.

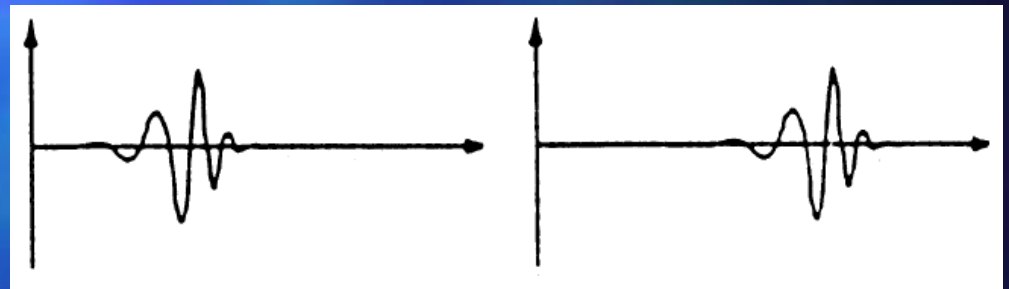
The Wavelet Function $\psi_{a,b}(x)$

→ The characteristic parameters:

Scale "a"



Position "b"



$$C_{(a,b)} = \int_{-\infty}^{+\infty} s(x) \psi_{a,b}(x) dx$$

The Wavelet Function $\psi_{a,b}(x)$

→ A complete set of basic Wavelet functions $\psi_{a,b}(x)$ can be obtained by the dilation and shifting of $\psi(x)$ according to:

$$\psi_{a,b}(x) = \frac{1}{\sqrt{|a|}} \psi\left(\frac{x-b}{a}\right)$$

where $(a,b) \in R$ and $a \neq 0$.

$$C_{(a,b)} = \int_{-\infty}^{+\infty} s(x) \psi_{a,b}(x) dx$$

The Wavelet Function $\psi_{a,b}(x)$

→ If one multiplies each coefficient by the properly dilated and shifted Wavelet function, one should regenerate the original function $s(x)$.

The Wavelet Function $\psi_{a,b}(x)$

- There exists a relationship between the Wavelet scale and the frequency.
- A small scale will produce compressed Wavelet where the details change rapidly in higher frequencies.
- With a large scale, the Wavelet is expanded and details change slowly with lower frequencies.

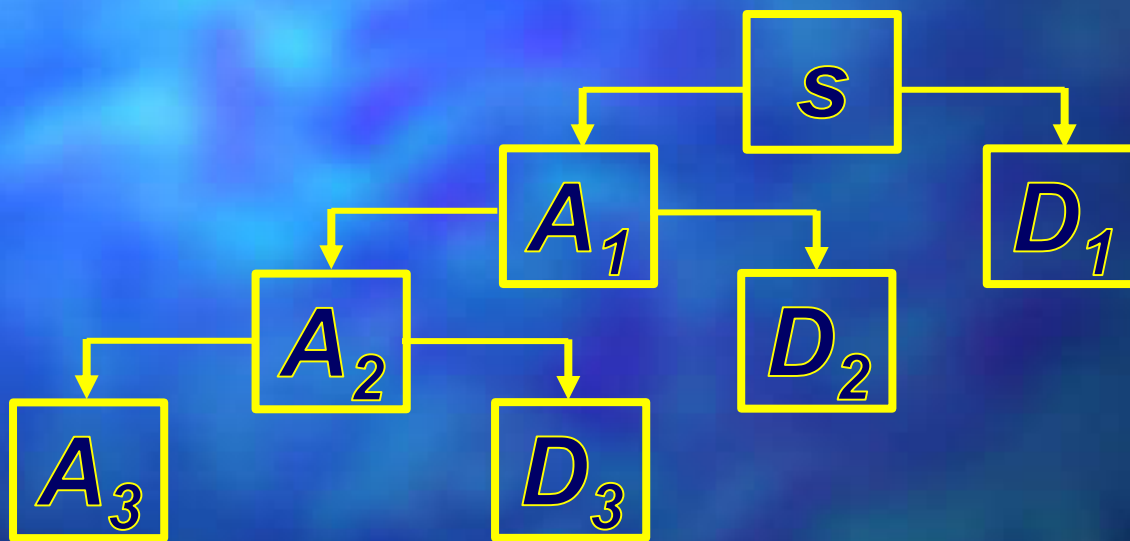
De-noising Methodology

→ Eddy-Current signals de-noising is performed by transforming the data to the time frequency domain using multiple scale levels by a Wavelet transform.

→ A given signal $s(x)$ is decomposed in each scale level into an “*approximation*” component A_i and a “*detail*” component D_i .

De-noising Methodology

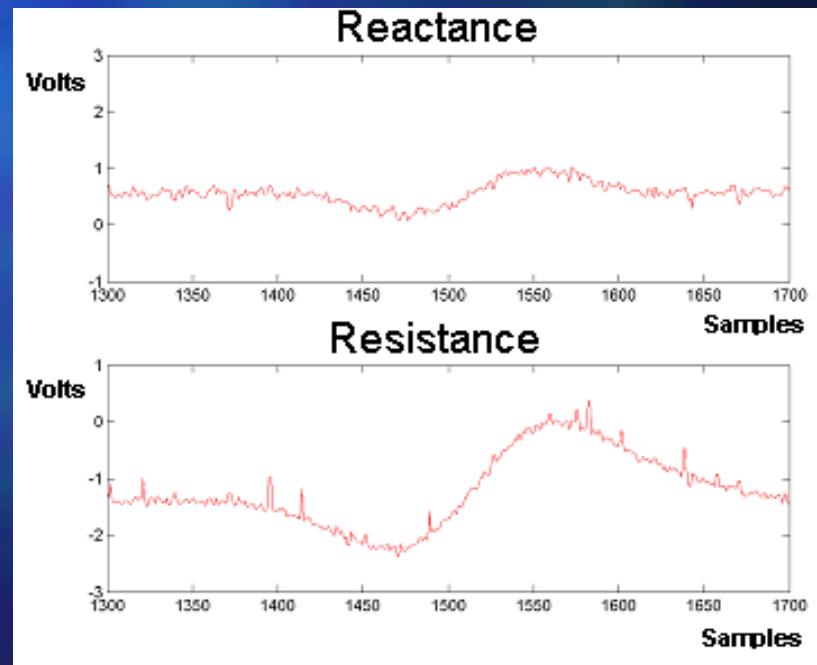
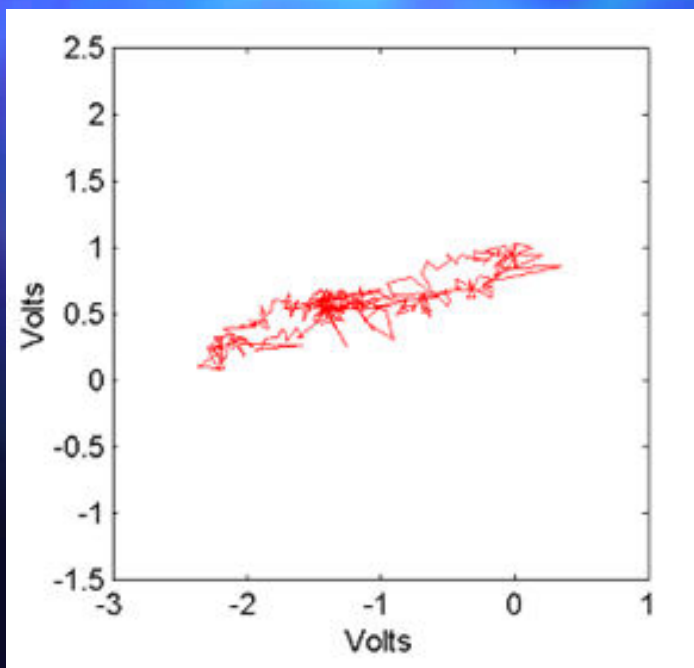
→ Example: 3 level decomposition tree of a signal s into approximations and details.



$$s = A_3 + D_3 + D_2 + D_1 = A_2 + D_2 + D_1 = A_1 + D_1$$

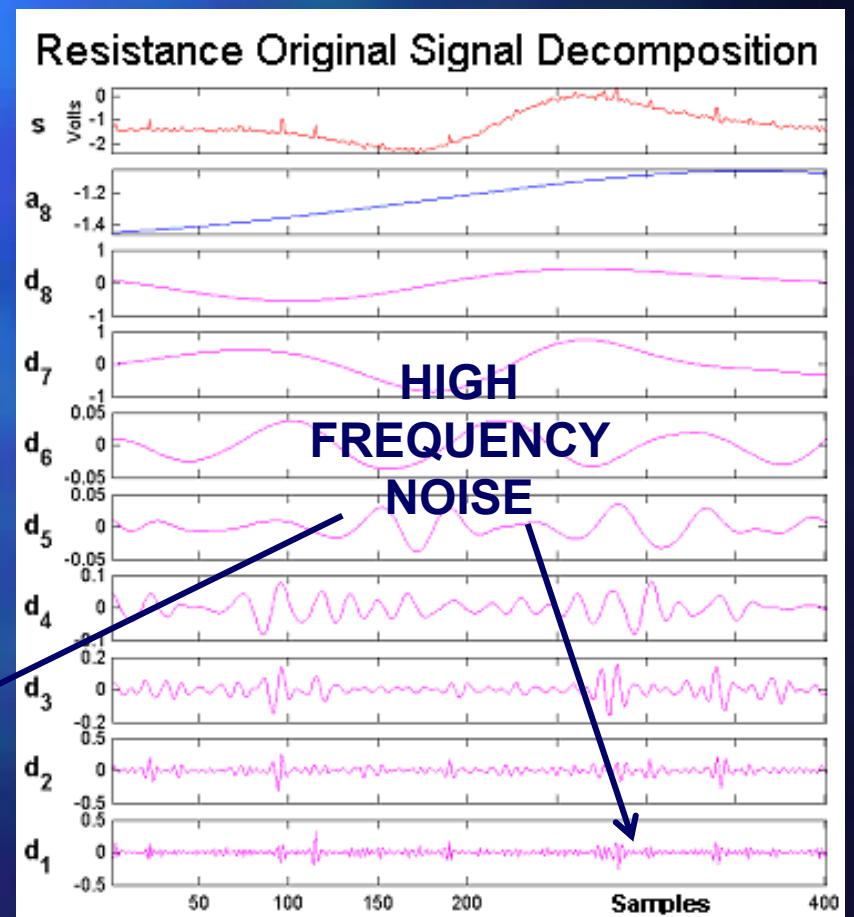
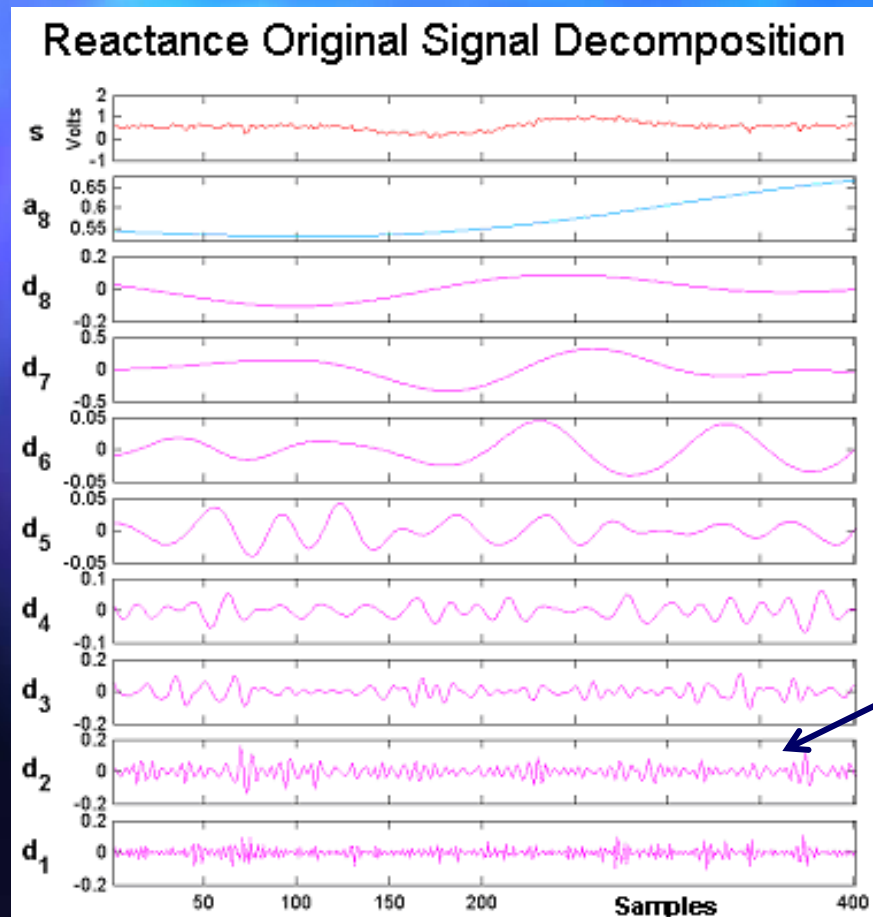
Applying the Methodology

→ The first step is to separate the inductive component (X_L) and the resistive component (R) of the noisy Eddy Current signal:



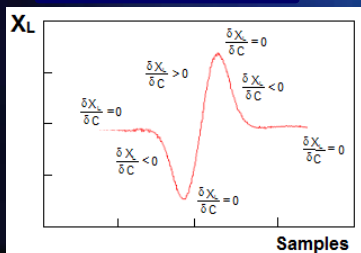
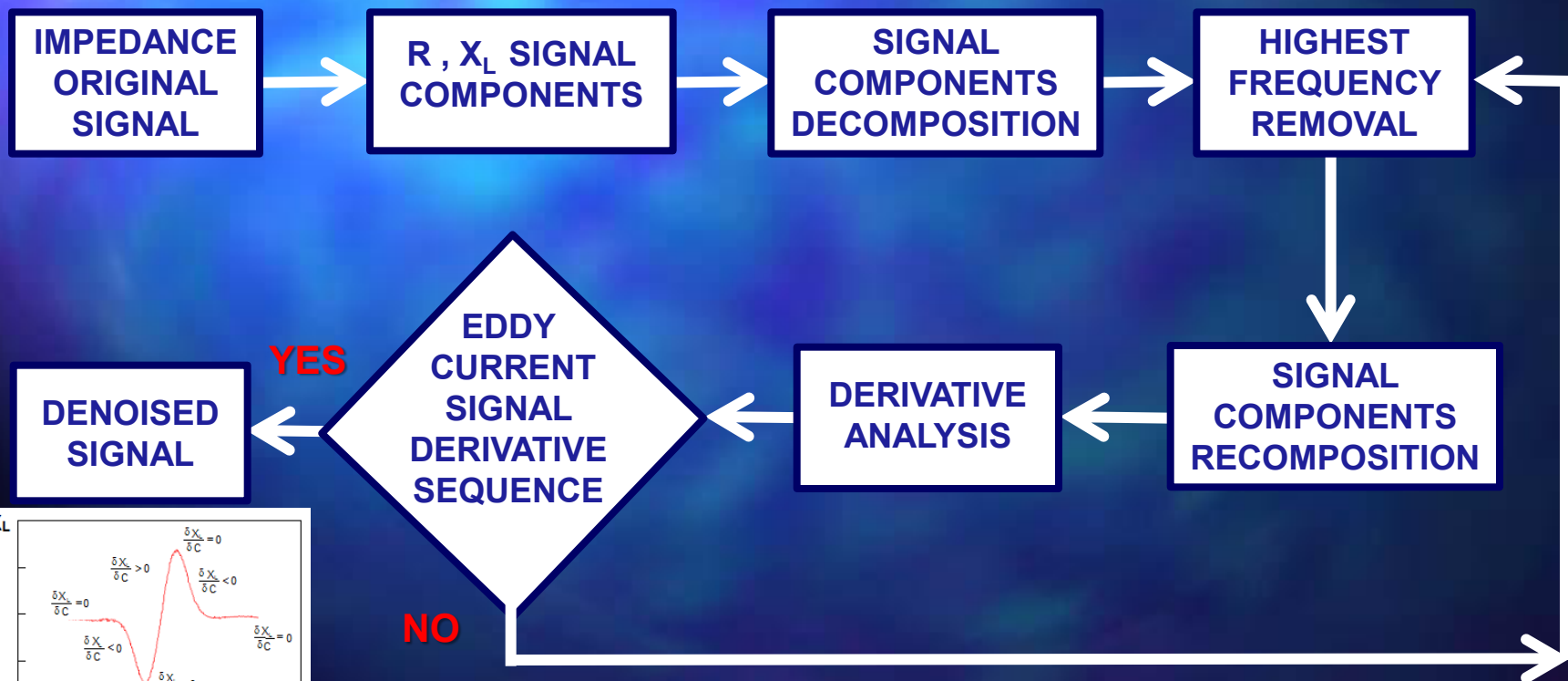
Frequency Decomposition

→ Then, X_L and R are multi level frequency decomposed through WT:



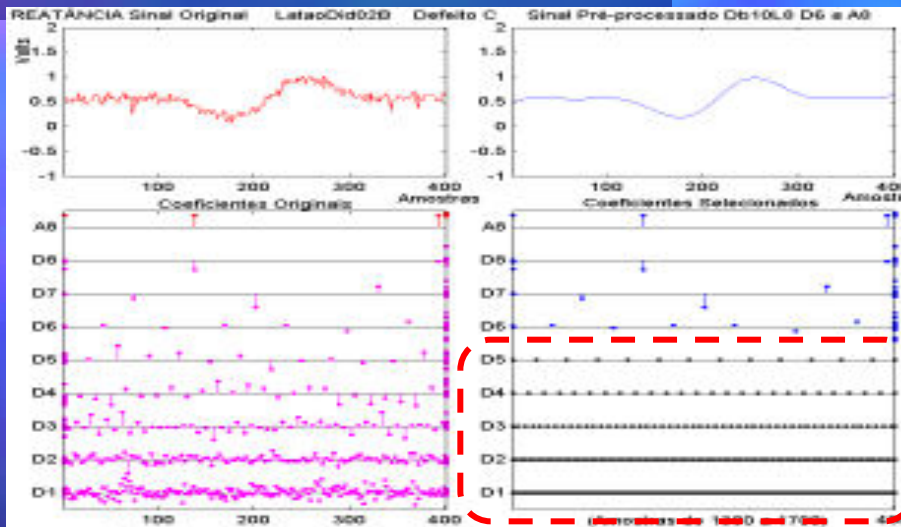
Derivative Analysis Algorithm

→ The noise presented is a typical high frequency noise which is removed through a derivative analysis algorithm.



Wavelet $C(a,b)$ Coef. Selection

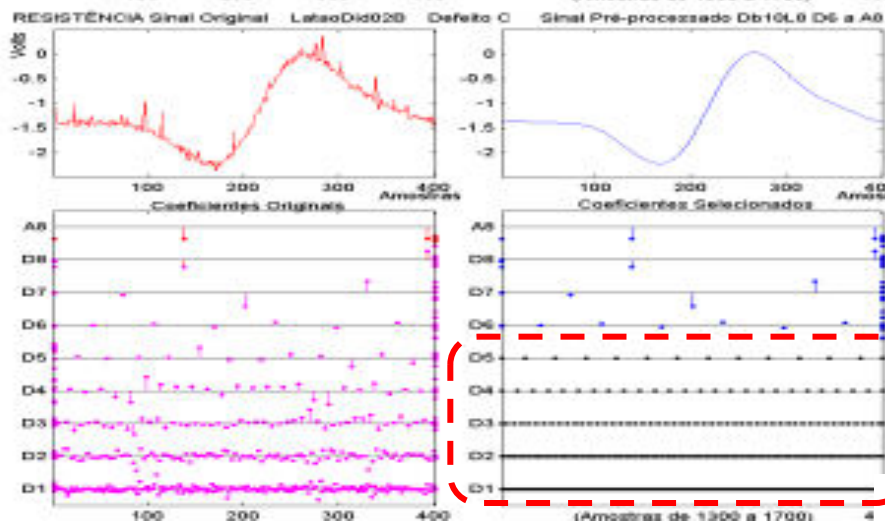
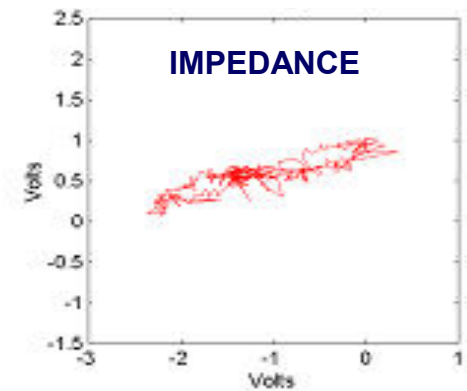
→ Frequencies removal:



REACTANCE
Coefficients $C(a,b)$
Selection

A0	20	+	+	20
D8	20	+	+	20
D7	21	+	+	21
D6	24	+	+	24
D5	30	+	+	0
D4	42	+	+	0
D3	66	+	+	0
D2	114	+	+	0
D1	210	+	+	0
	517	+	+	85

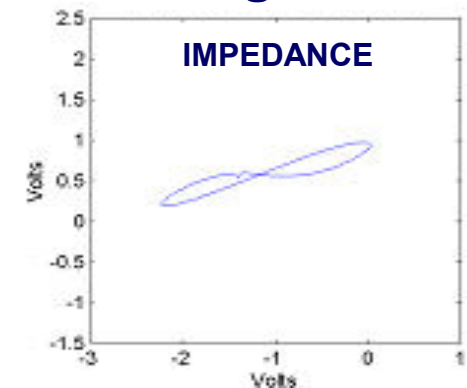
Original Signal



RESISTANCE
Coefficients $C(a,b)$
Selection

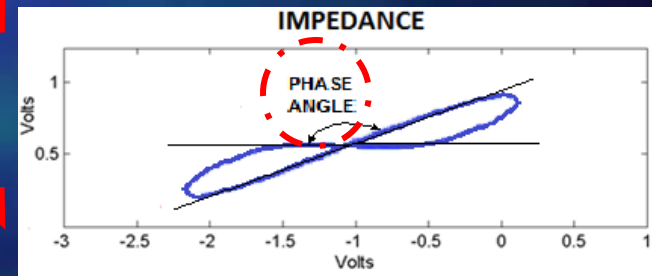
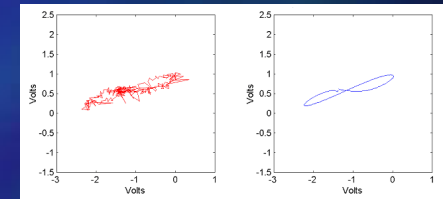
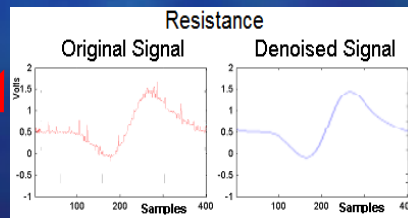
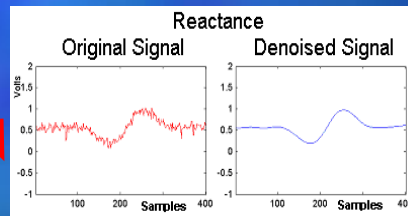
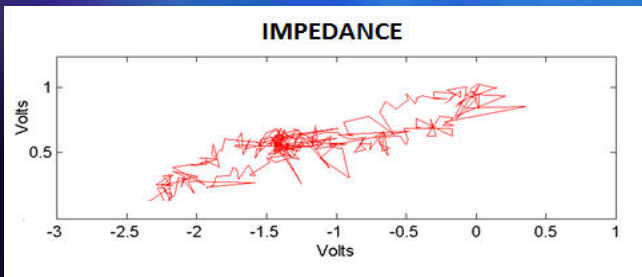
A0	20	+	+	20
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	517	+	+	85

De-noised
Signal



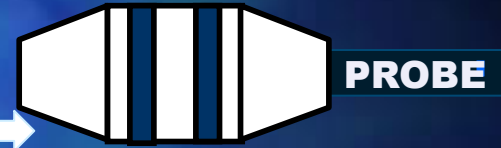
Signal Recomposition

➔ After noise removal, the de-noised reactance and resistance components are recomposed producing a clean Lissajous figure from which the Phase Angle can be extracted.

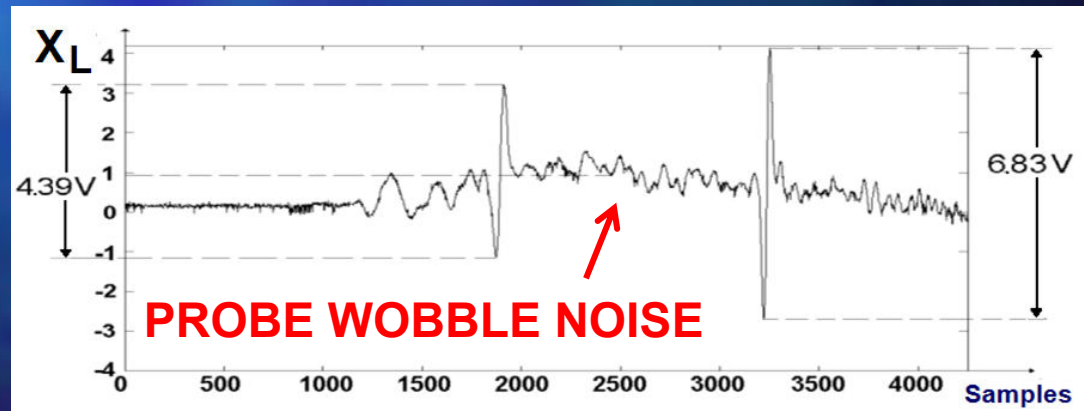


Probe Wobble Noise

→ This noise is generated by the necessary slack between the probe and the fuel tube.

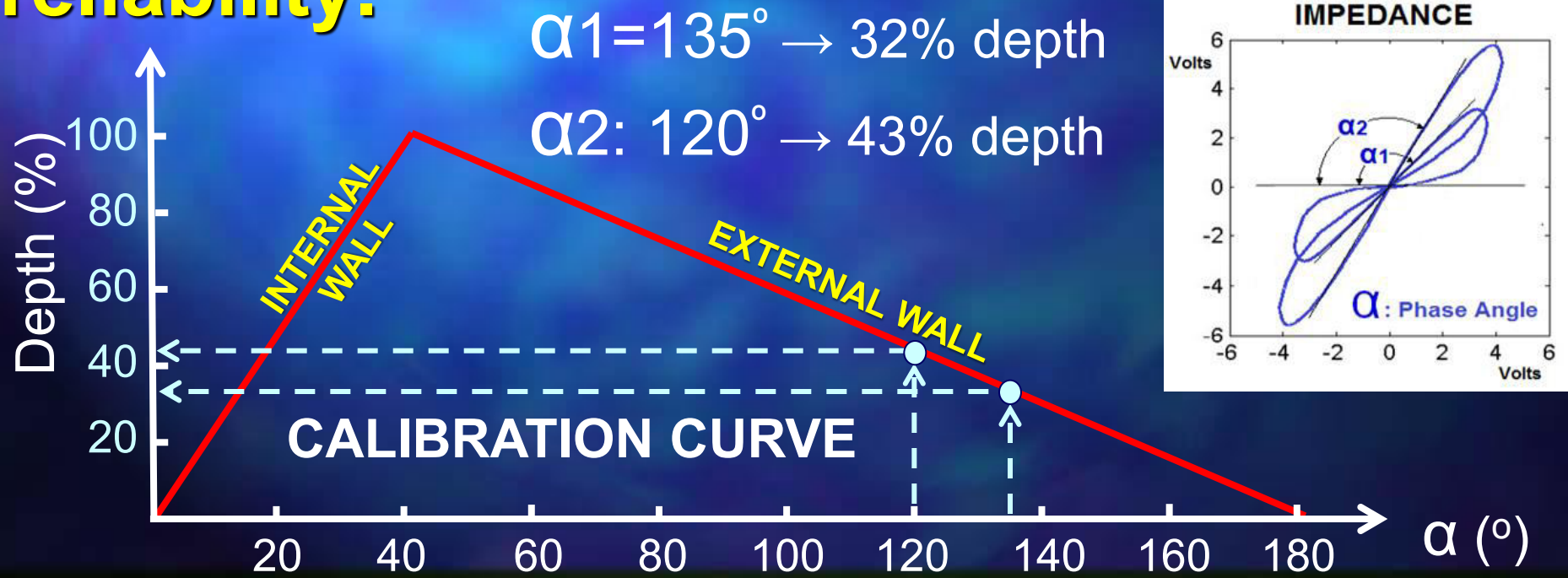


→ The greater the radial movement of the probe, the greater the inductive component variation of the Eddy Current signal.



Probe Wobble Noise

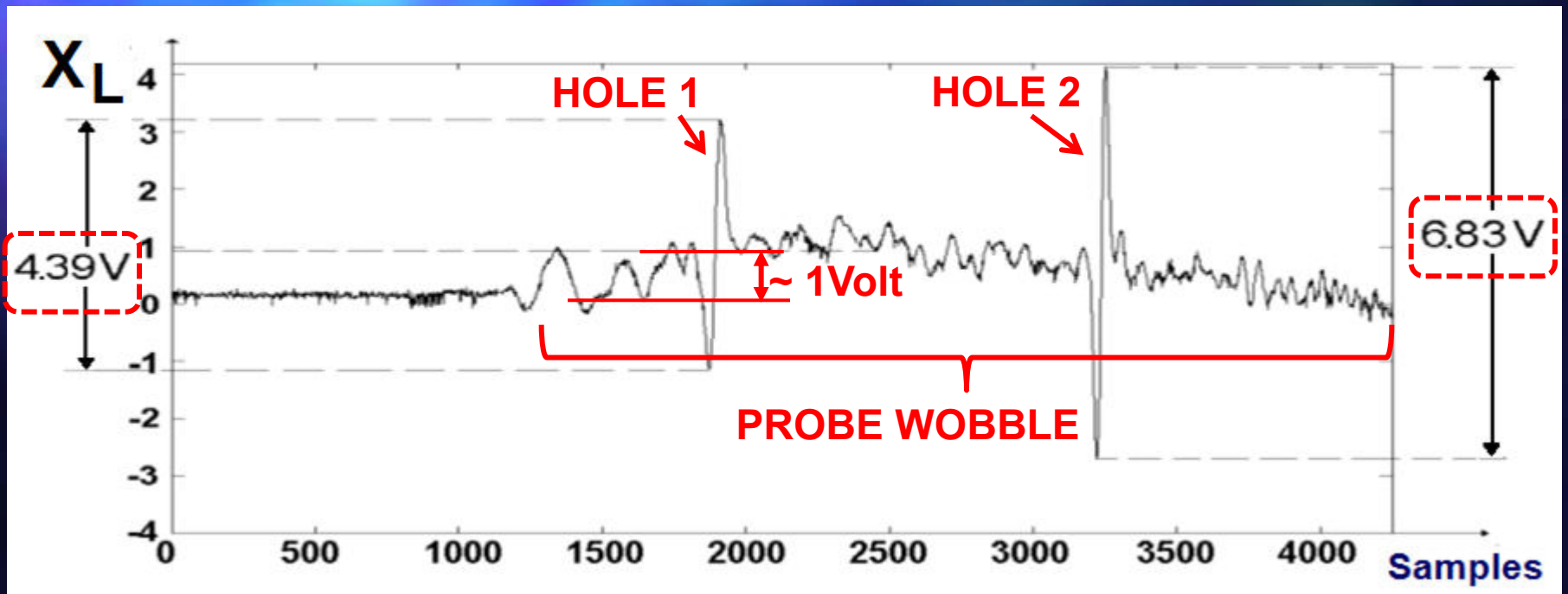
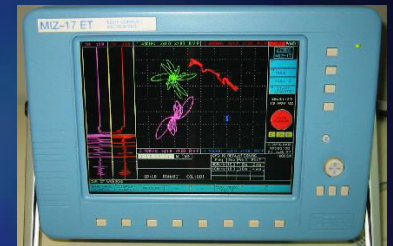
- Due to this noise, the Lissajous figure changes as well as its Phase Angle α .
- Consequently, this noise hinders the signal analysis and reduces inspection reliability.



Probe Wobble De-Noising

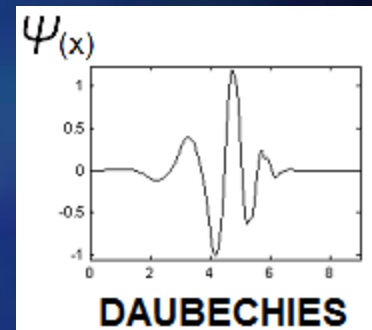
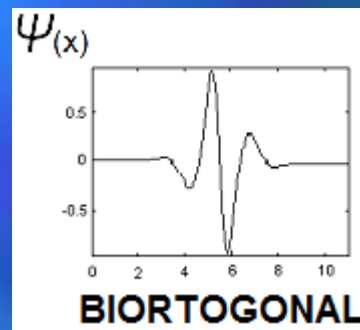
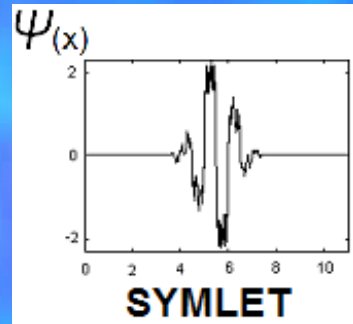
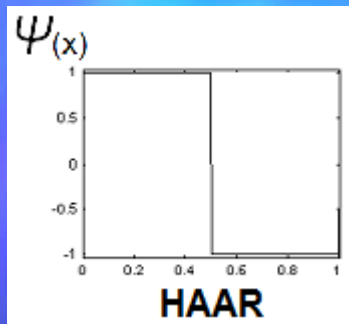
→ Example: X_L component, $\text{Ø}19.05\text{mm}$ fuel tube with 2 artificial holes.

→ The signal was generated by a MIZ-17ET equipment.



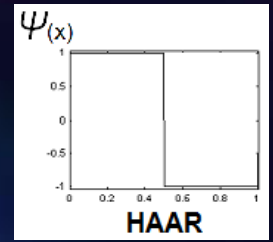
Probe Wobble De-Noising

→ Four wavelet functions were used for de-noising:

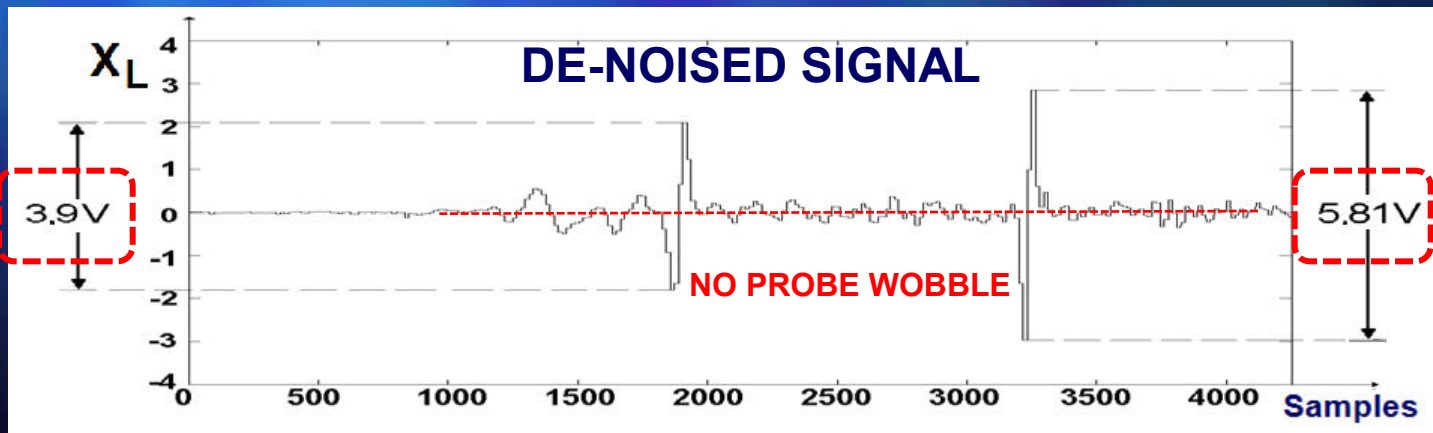
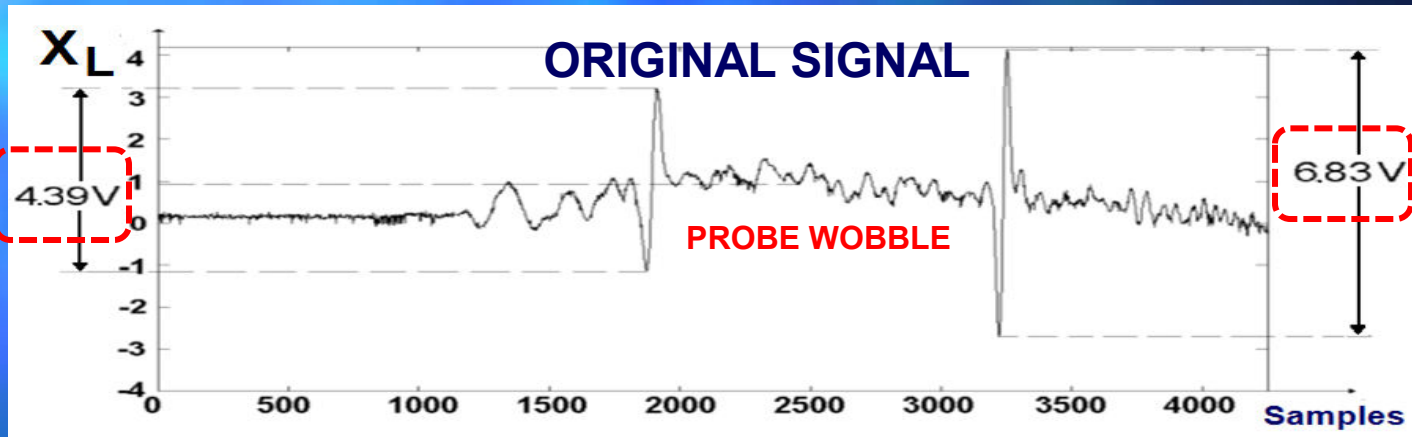


→ An 8 level frequency decomposition allowed a convenient selection of the approximation (A_i) and the detail (D_i) components.

De-Noising – Haar wavelet



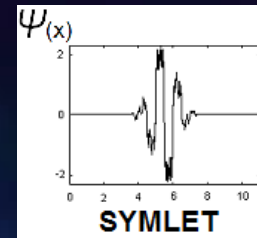
→ No probe wobble; different X_L ...



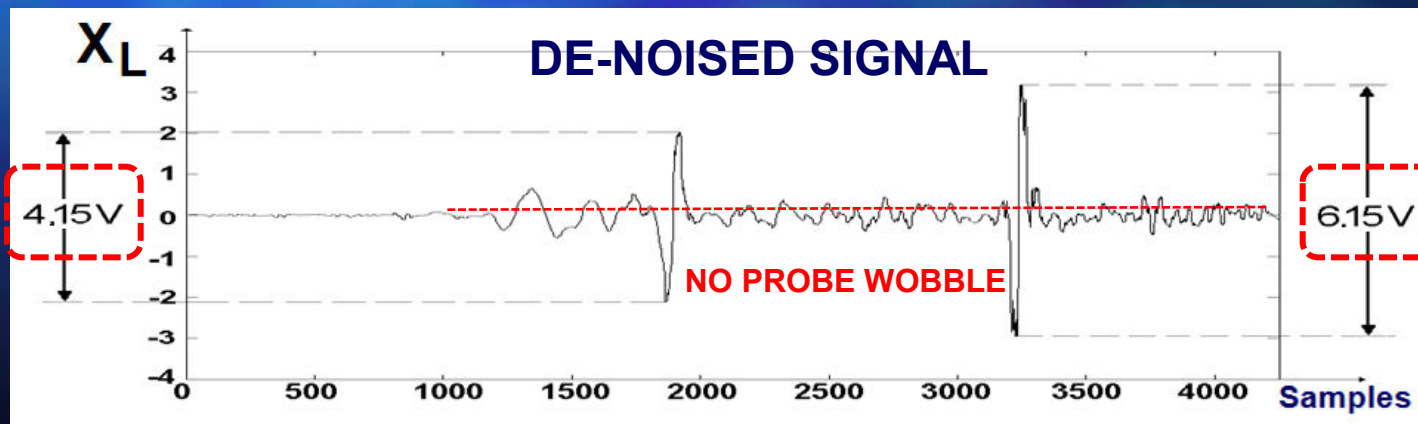
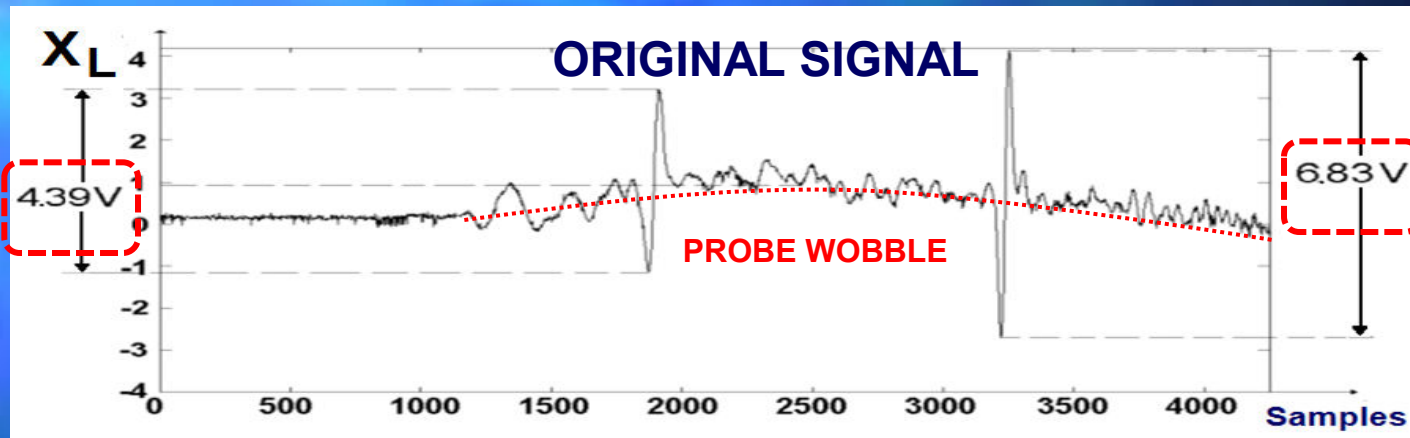
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De-Noising – Symlet wavelet



→ No probe wobble; different X_L ...

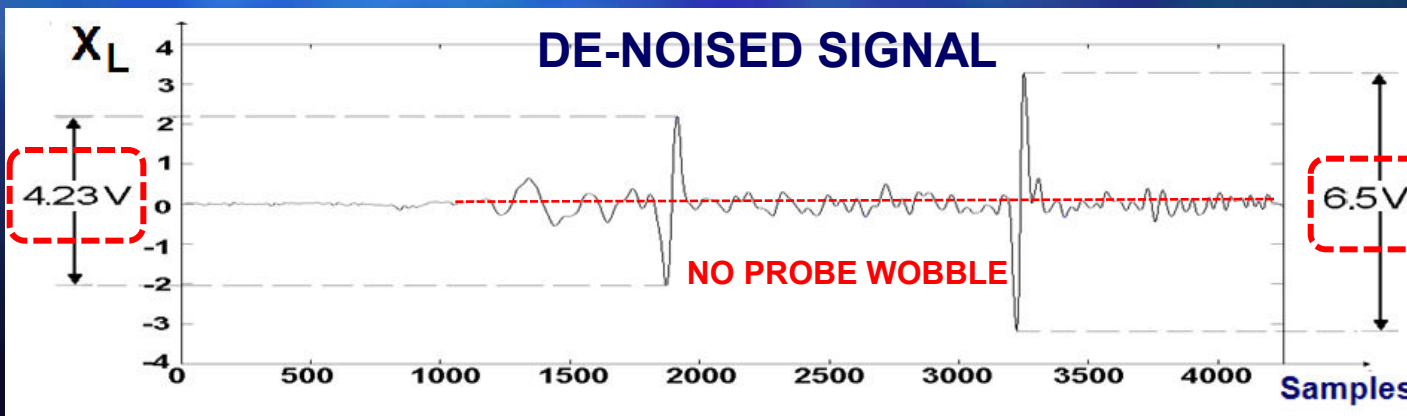
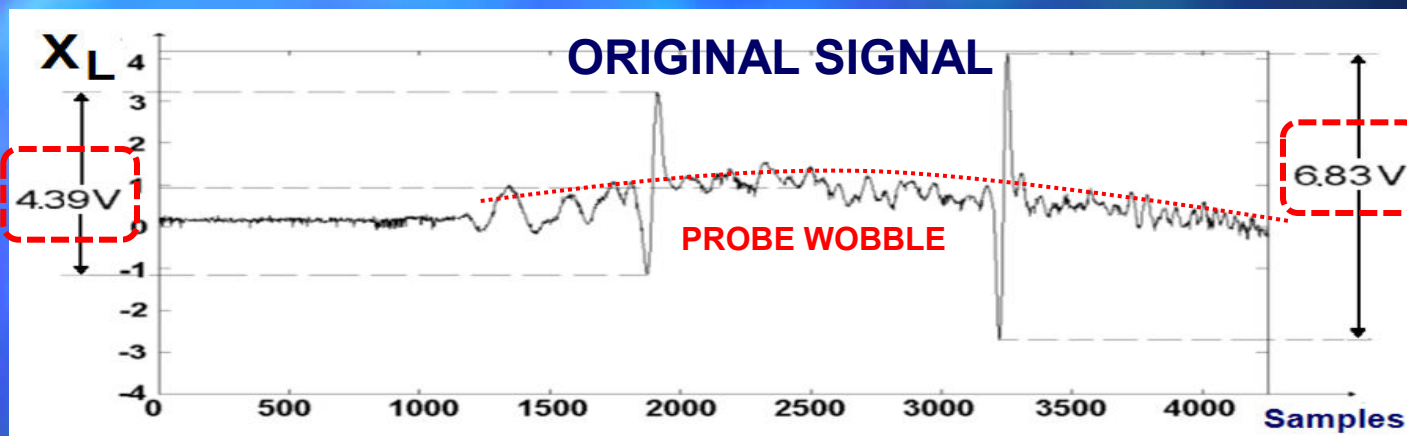
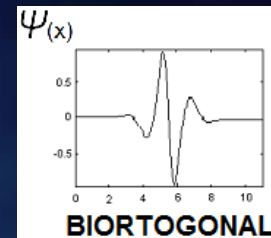


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De-Noising – Biortogonal wavelet

→ No probe wobble; different X_L ...

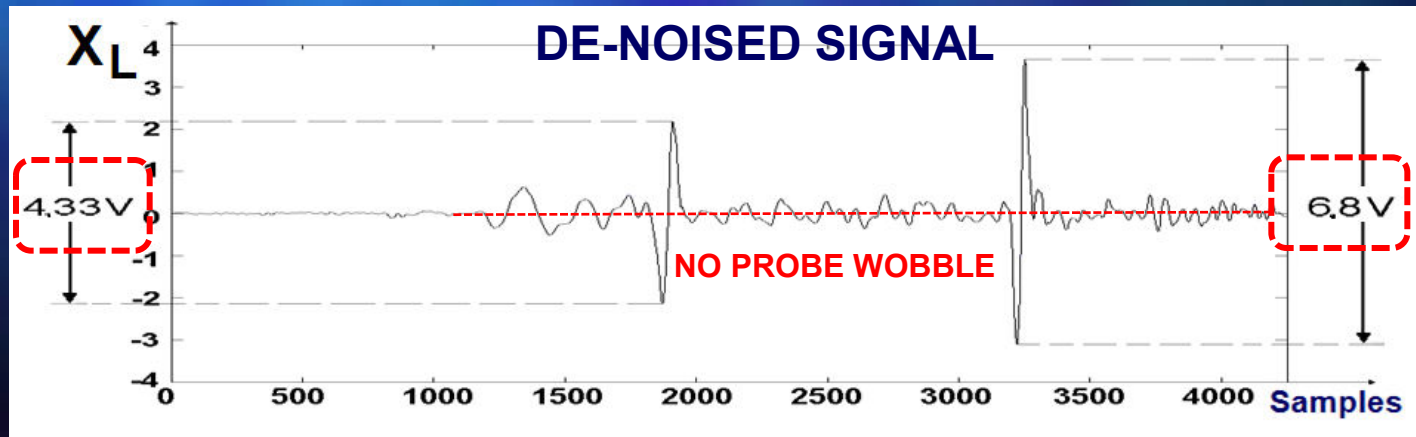
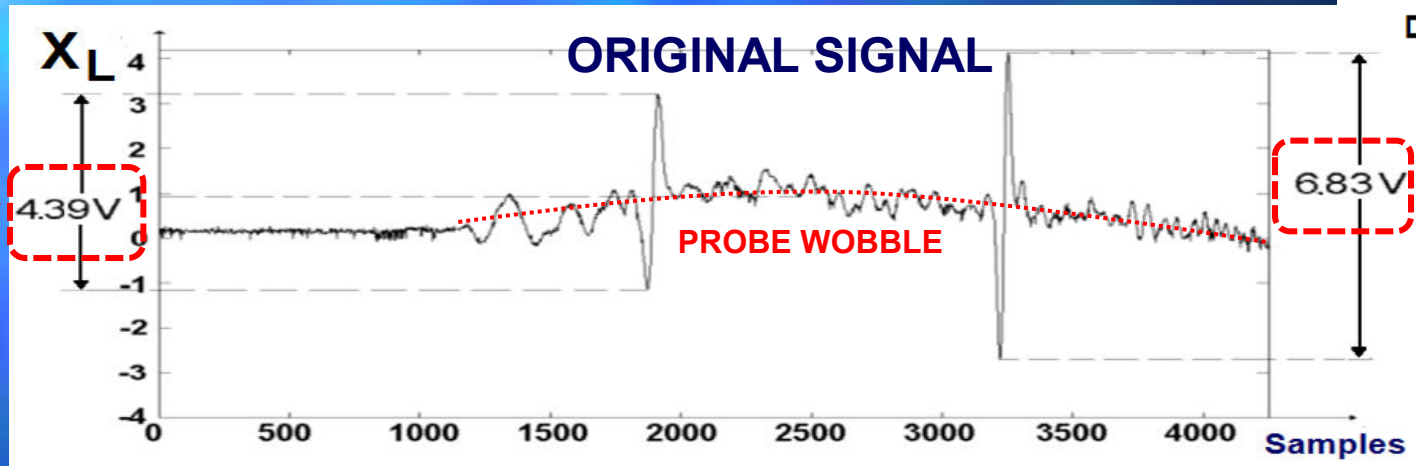
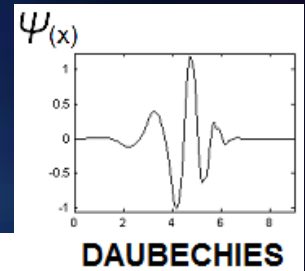


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De-Noising – Daubechies wavelet

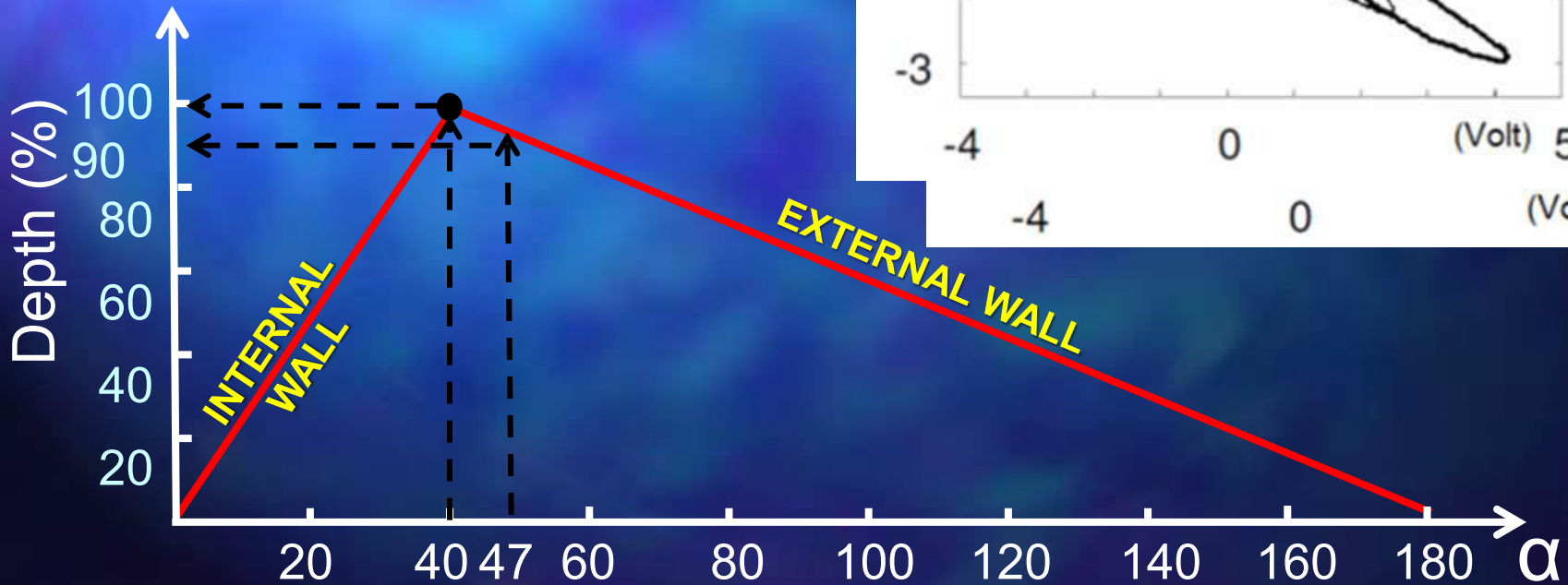
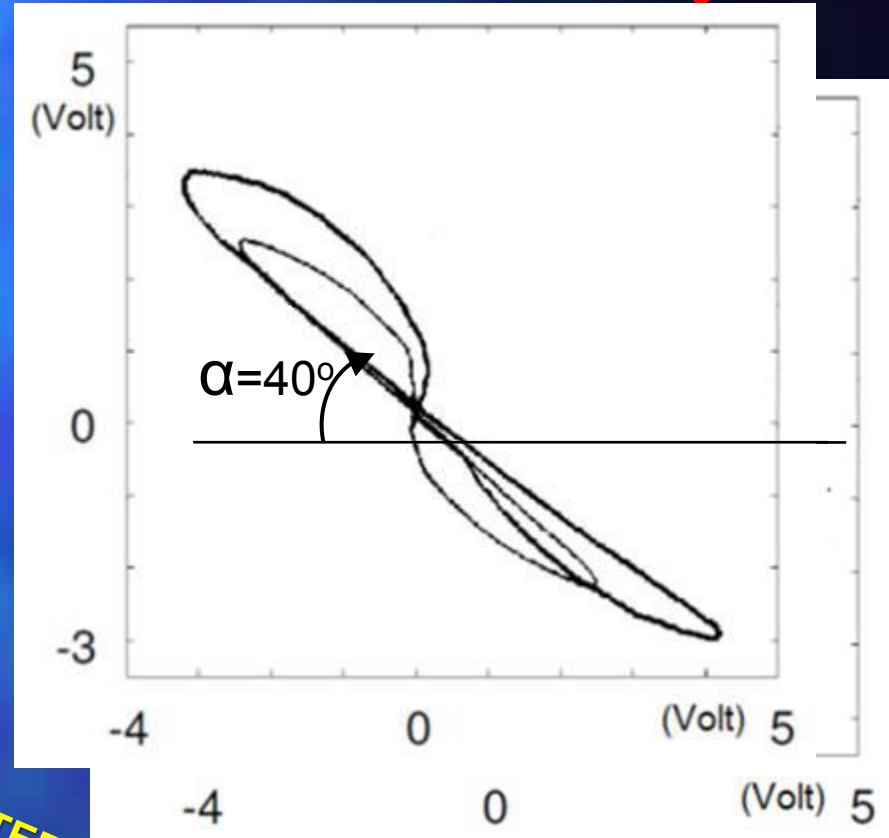
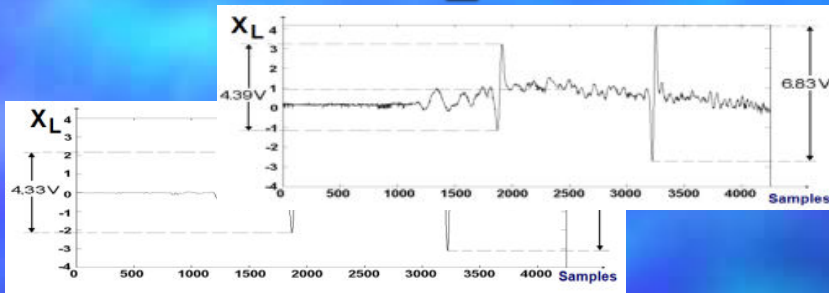
→ No probe wobble, similar X_L !



De-Noising – Daubechies wavelet

→ No probe wobble

→ Similar X_L !

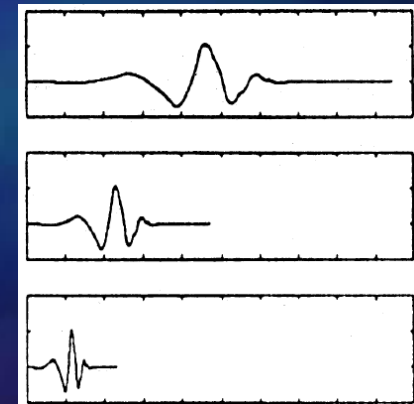
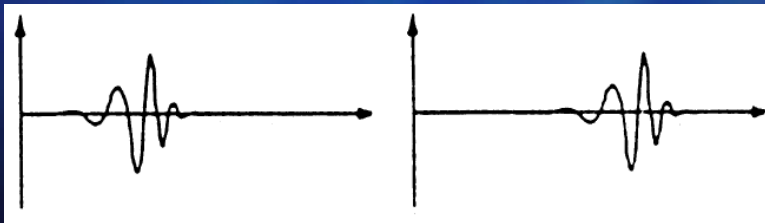
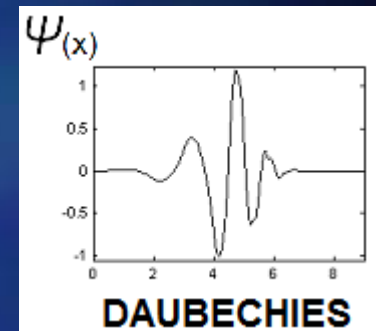


Conclusions

→ The denoising of Eddy Current signals is a process which determines the better

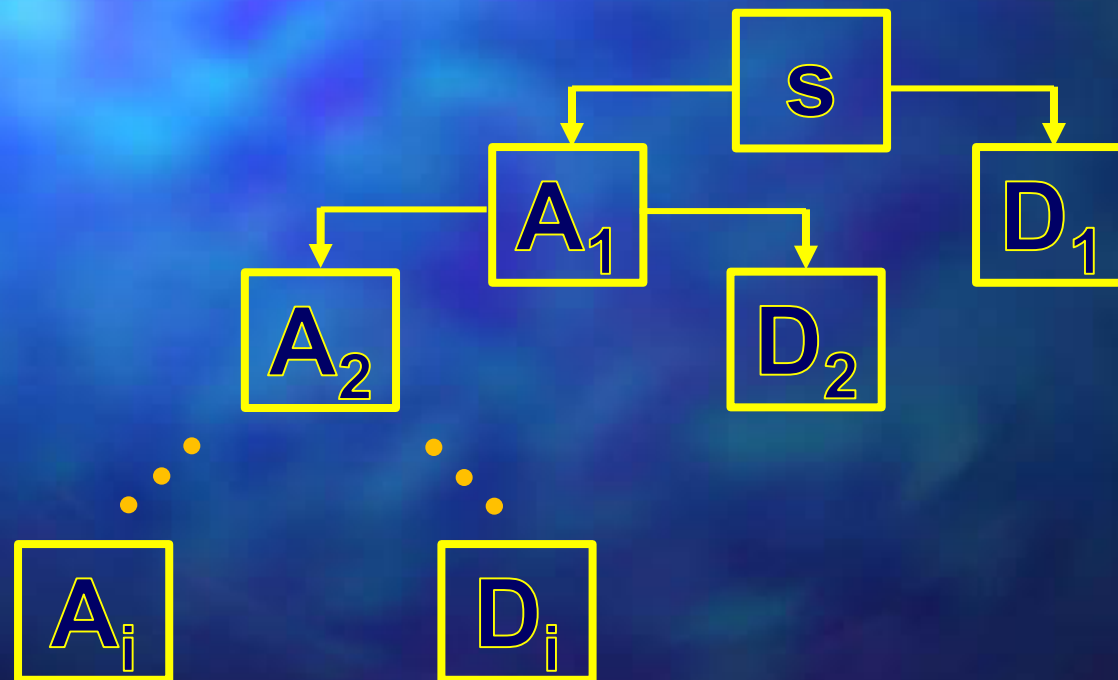
Wavelet function $\psi_{a,b}(x)$:

→ ... and the adequate set of coefficients $C_{a,b}$, with the characteristic parameters “a” scale and “b” position.



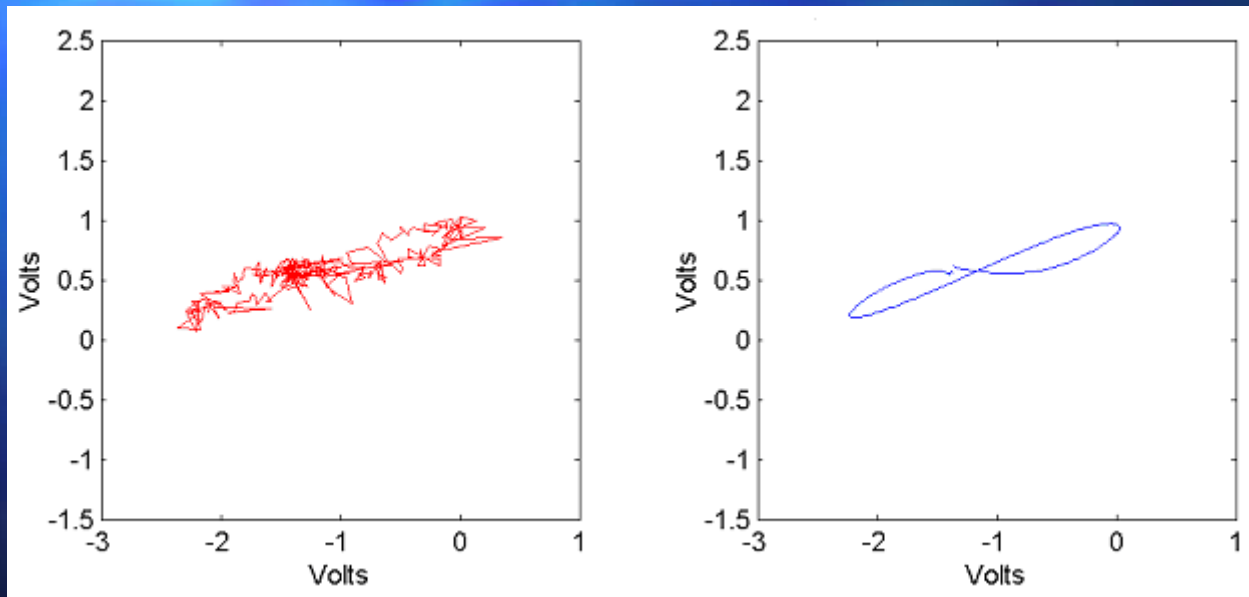
Conclusions

→ Besides that, the noisy signal s must be decomposed in each scale level into an “*approximation*” component A_i and a “*detail*” component D_i .



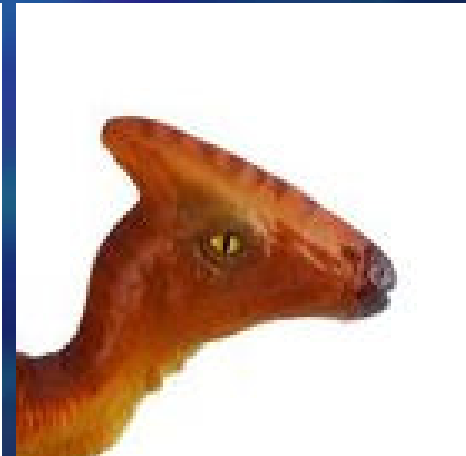
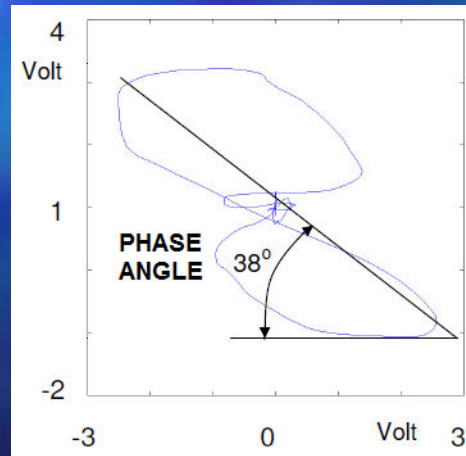
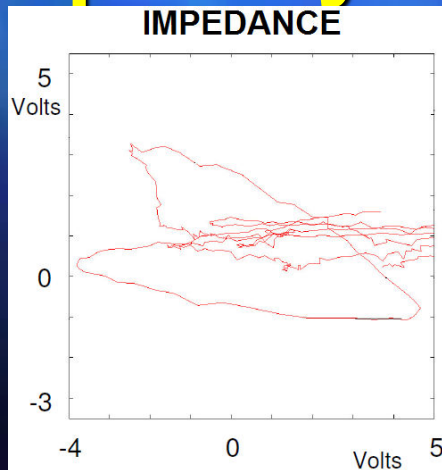
Conclusions

→ The data acquisition instrumentation noise, although it may distort a lot the signal, it is not difficult to be removed as it is a high frequency noise.



Conclusions

- The material noise is originated in the magnetic properties variations caused for example by the manufacturing process of the tube.
- The Eddy Current Lissajous figure is completely distorted by this noise.



Conclusions

- The probe wobble signal de-noising offers reliable results as it allows an efficient removal of noise, maintaining the essential signal information.
- The methodology above described is being applied in inspection automation of fuel tubes by using a Matlab software enhanced by artificial intelligence.